



US Department  
of Transportation  
Federal Aviation  
Administration

**MAJOR REPAIR AND ALTERATION**  
**(Airframe, Powerplant, Propeller, or Appliance)**

Form Approved  
OMB No. 2120-0020  
5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9 1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark N310CM	Serial No. 310P-0178	
	Make CESSNA	Model 310P	Series
2. Owner	Name (As shown on registration certificate) JAMES E MCKINNON	Address (As shown on registration certificate) Address 237 CLEARFIELD RD	
		City KNOXVILLE	State TN
		Zip 37922-6801	Country US

**3. For FAA Use Only**

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
<input type="checkbox"/>	<input checked="" type="checkbox"/>	AIRFRAME	_____	(As described in Item 1 above)	_____
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type _____ Manufacturer _____		

**6. Conformity Statement**

A. Agency's Name and Address		B. Kind of Agency	
Name KENNETH D PHILLIPS		U. S. Certificated Mechanic	Manufacturer
Address 1235 AIRPORT RD		Foreign Certificated Mechanic	C. Certificate No.
City SEVIERVILLE	State TN	Certificated Repair Station	
Zip 37862	Country US	Certificated Maintenance Organization	3400858

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual KENNETH D PHILLIPS <i>Kenneth D Phillips</i> 3 Feb 2020
--	---

**7. Approval for Return to Service**

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> Approved <input type="checkbox"/> Rejected					
BY	FAA Flt. Standards Inspector	Manufacturer	Maintenance Organization	Persons Approved by Canadian Department of Transport	
	FAA Designee	Repair Station	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)	
Certificate or Designation No. 3400858IA		Signature/Date of Authorized Individual KENNETH D PHILLIPS <i>Kenneth D Phillips</i> 3 Feb 2020			

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

N310CM

3 Feb 2020

8. Description of Work Accomplished

Nationality and Registration Mark

Date

1. REMOVED MECHANICAL FUEL FLOW GUAGE, CAPPED FUEL LINE AT WING ROOT.
2. INSTALATION OF JP INSTRUMENTS FUEL FLOW INDICATING SYSTEM IN ACCORDANCE WITH JPI STC SA00861SE - INSTALATION MANUAL FS-450, REPORT NO. 400, REVISION B DATED 12-12-2012.
3. INSTALLED 1 AMP CIRCUIT BREAKER . DRAW ON FS-450 1/4 AMP
4. NO SIGNIFICANTE CHANGE IN WEIGHT AND BALANCE

----- END -----

[ ] Additional Sheets Are Attached

United States of America  
Department of Transportation—Federal Aviation Administration  
**Supplemental Type Certificate**

*Number* SA00861SE

*This certificate, issued to*

J. P. Instruments  
P.O. Box 7033  
Huntington Beach, CA 92646

*certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part \* of the \* Regulations.*

*Original Product—Type Certificate Number:*

\*See attached FAA Approved Model List (AML)

*Make:*

No. SA00861SE for a list of approved airplane

*Model:*

models and applicable airworthiness regulations.

*Description of the Type Design Change:* Installation of J.P. Instruments (JPI) fuel flow indicating system in accordance with JPI Installation Manual FS-450, Report No. 400, Revision -, dated August 16, 2000, or later FAA approved revision.

*Limitations and Conditions:* Approval of this change in type design applies to the aircraft models listed on the AML only. This approval should not be extended to other aircraft of these models on which other previously approved modifications are incorporated unless it is determined that the relationship between this change and any of those other previously approved modifications, including changes in type design, will introduce no adverse effect upon the airworthiness of that aircraft. A copy of this Certificate, FAA Approved Model List (AML) No. SA00861SE, and Airplane Flight Manual Supplement No. 1, dated December 18, 2000, or later FAA approved revision, must be maintained as part of the permanent records for the modified aircraft.

Note: This STC requires the installation of a fuel flow transducer per STC SA00432SE; or aircraft listed on the FAA approved AML SA00432SE and that have been previously modified with a fuel flow indication system that utilizes the Flowscan fuel flow transducer, P/N: 201-A, 201-B, 201-C or 231 are eligible for installation for the FS-450. This certificate does not constitute installation approval of the fuel flow transducer.

If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

*This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.*

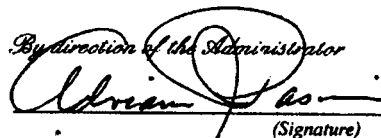
*Date of application:* April 7, 2000

*Date reissued:*

*Date of issuance:* December 18, 2000

*Date amended:*



*By direction of the Administrator*  
  
(Signature)

Acting Manager, Seattle Aircraft  
Certification Office

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.



**J.P. INSTRUMENTS FAA APPROVED MODEL LIST for STC SA00861SE  
FOR  
INSTALLATION OF THE (FS-450) 450000 SERIES FUEL FLOW INSTRUMENT**

Issue Date: December 18, 2000

ITEM	AIRCRAFT MAKE	AIRCRAFT MODEL	TYPE CERTIFICATE NUMBER	CERTIFICATION BASIS FOR ALTERATION	AML REVISION DATE
41.	Cessna	177RG	A20CE	FAR 23	12-18-2000
42.	Cessna	180, 180A, B, C, D, E, F, G, H, J, K	5A6	CAR 3	12-18-2000
43.	Cessna	182, 182A, B, C, D, E, F, G, H, J, K, L, M, N, P, Q, R	3A13	CAR 3	12-18-2000
44.	Cessna	182E, F, G, H, J, K, L, M, N, P, Q (Peterson, STC SA3825SW)	3A13	CAR 3	5-31-2001
45.	Cessna	182T	3A13	CAR 3	5-31-2001
46.	Cessna	T182T	3A13	CAR 3	5-31-2001
47.	Cessna	182 Series Air Planes Corp, STC SA00152W/ Engine IO-520 and IO-550	3A13	CAR3	12-18-2000
48.	Cessna	182S	3A13	FAR 23	12-18-2000
49.	Cessna	R182, T182, TR182	3A13	CAR 3	12-18-2000
50.	Cessna	185, 185A, B, C, D, E, A185E, A185F	3A24	CAR 3	12-18-2000
51.	Cessna	188, 188A, 188B, A188, A188A, A188B, T188C	A9CE	FAR 21	12-18-2000
52.	Cessna	195, A, B	A-790	CAR 3	12-18-2000
53.	Cessna	206, 206, P206A, B, C, D, E, TP206A, B, C, D, E, U206, U206A, B, C, D, E, F, G, TU206A, B, C, D, E, F, G	A4CE	CAR 3	12-18-2000
54.	Cessna	206H	A4CE	CAR 3	5-31-2001
55.	Cessna	T206H	A4CE	FAR 23	12-18-2000
56.	Cessna	U206A, B, C, D, E, F, G. (with Lycoming TIO-540- series engine per STC SA2123NM)	A4CE	CAR 3	12-18-2000
57.	Cessna	207, 207A, T207, T207A	A16CE	FAR 23	12-18-2000
58.	Cessna	210, 210A, B, C, D, E, F 210-5, 210-5A, T210F, 210G, T210G, 210H, T210H, 210J, T210J, T210K, 210K, 210L, T210L, 210M, T210M, 210N, P210N, T210N, 210R, P210R, T210R	3A21	FAR 23	12-18-2000
59.	Cessna	T303	A34CE	FAR 23	1-27-2003
60.	Cessna	310, C, D, E, F, G, H, E310H, 310I, J, K, L, N, P, Q, R 310J-1, E310J, T310P, Q, R, T310Q, 310R, T310R	3A10	CAR 3	12-18-2000
61.	Cessna	320, A, B, C, D, E, F, 320-1, 335, 340, 340A	3A25	CAR 3	12-18-2000
62.	Cessna	336	A2CE	CAR 3	12-18-2000



**J.P. INSTRUMENTS**  
**Supplement No. 1**  
**P.O. Box 7033**  
**Huntington Beach, CA 92646**

**FAA APPROVED**  
**AIRPLANE/ROTORCRAFT FLIGHT MANUAL SUPPLEMENT OR**  
**SUPPLEMENTAL AIRPLANE FLIGHT MANUAL (INCLUDING POH AND**  
**FAA AFM)**  
**(FOR THOSE AIRCRAFT WITHOUT A BASIC AIRPLANE FLIGHT**  
**MANUAL)**

**FS-450 FUEL FLOW INDICATOR**  
**FOR**

**Single and Twin, Reciprocating Engine Powered Aircraft as listed**  
**On Approved Model List of**

**STC SA00861SE**

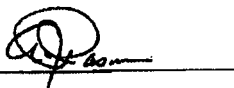
**REG. NO.** N310CM

**SER. NO.** 310P-0178

This Supplement must be attached to the  
**FAA Approved Airplane/Rotorcraft Flight Manual** when the J.P. Instruments FS-450  
is installed in accordance with Supplemental Type Certificate SA 00432SE. For those  
airplanes without a basic Airplane Flight Manual, the Supplemental AFM must be in the  
aircraft when the FS-450 is installed.

The information contained in this **Airplane/Rotorcraft Flight Manual Supplement/**  
**Supplemental Aircraft Flight Manual** supplements or supersedes the basic manual/  
placards only in those areas listed. For limitations, procedures and performance information  
not contained in this supplement, consult the basic Airplane Flight manual, Markings and  
Placards

**FAA APPROVED:**



**Manager, Special Certification Branch, ANM-190S**  
**Federal Aviation Administration**  
**Seattle Aircraft Certification Office**  
**Transport Airplane Directorate**

**Date:** December 18, 2000



***J.P.Instruments Fuel Flow Installation Manual***  
***FS-450***

Report # 400  
Page 1 of 16  
Rev A : Date 02/1/2002

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J.P.INSTRUMENTS  
PO BOX 7033  
HUNTINGTON BEACH CA 92646



# ***J.P.Instruments Fuel Flow Installation Manual***

FS-450

FAA Approved

Report # 400

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Rev B : Date 12/12/2012

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J.P.INSTRUMENTS  
PO BOX 7033  
HUNTINGTON BEACH CA 92704



# Pilot's Guide

## Fuel Scan

### FS-450M TWIN

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Printed in the United States of America

#### **J . P . INSTRUMENTS**

**Information:** P. O. Box 7033  
Huntington Beach, CA 92646

**Factory:** 3185B Airway  
Costa Mesa, CA 92626

(714) 557-5434 Fax (714) 557-9840

[www.jp instruments.com](http://www.jp instruments.com)

Rev NC 11/03





US Department  
of Transportation  
Federal Aviation  
Administration

**MAJOR REPAIR AND ALTERATION**  
**(Airframe, Powerplant, Propeller, or Appliance)**

Form Approved  
OMB No. 2120-0020  
5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9 1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark N310CM	Serial No. 310P-0178	
	Make CESSNA	Model 310P	Series
2. Owner	Name (As shown on registration certificate) JAMES E MCKINNON	Address (As shown on registration certificate) Address 237 CLEARFIELD RD	
		City KNOXVILLE	State TN
		Zip 37922-6801	Country US

**3. For FAA Use Only**

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
<input type="checkbox"/>	<input checked="" type="checkbox"/>	AIRFRAME	_____	(As described in Item 1 above)	_____
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type _____ Manufacturer _____		

**6. Conformity Statement**

A. Agency's Name and Address		B. Kind of Agency	
Name KENNETH D PHILLIPS		U. S. Certificated Mechanic	Manufacturer
Address 1235 AIRPORT RD		Foreign Certificated Mechanic	C. Certificate No.
City SEVIERVILLE	State TN	Certificated Repair Station	
Zip 37862	Country US	Certificated Maintenance Organization	3400858

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual KENNETH D PHILLIPS <i>Kenneth D Phillips</i> 3 Feb 2020
--	---

**7. Approval for Return to Service**

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ Approved ☐ Rejected

BY	FAA Fit. Standards Inspector	Manufacturer	Maintenance Organization	Persons Approved by Canadian Department of Transport
	FAA Designee	Repair Station	X Inspection Authorization	Other (Specify)

Certificate or Designation No. 3400858IA	Signature/Date of Authorized Individual KENNETH D PHILLIPS <i>Kenneth D Phillips</i> 3 Feb 2020
--	---

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

N310CM

3 Feb 2020

8. Description of Work Accomplished

Nationality and Registration Mark

Date

1. REMOVED MECHANICAL FUEL FLOW GUAGE, CAPPED FUEL LINE AT WING ROOT.
2. INSTALATION OF JP INSTRUMENTS FUEL FLOW INDICATING SYSTEM IN ACCORDANCE WITH JPI STC  
SA00861SE - INSTALATION MANUAL FS-450, REPORT NO. 400, REVISION B DATED 12-12-2012.
3. INSTALLED 1 AMP CIRCUIT BREAKER . DRAW ON FS-450 1/4 AMP
4. NO SIGNIFICANTE CHANGE IN WEIGHT AND BALANCE

----- END -----

[ ] Additional Sheets Are Attached

United States of America  
Department of Transportation—Federal Aviation Administration  
**Supplemental Type Certificate**

*Number* SA00861SE

*This certificate, issued to*

J. P. Instruments  
P.O. Box 7033  
Huntington Beach, CA 92646

*certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part \* of the \* Regulations.*

*Original Product—Type Certificate Number:*

\*See attached FAA Approved Model List (AML)

*Make:*

No. SA00861SE for a list of approved airplane

*Model:*

models and applicable airworthiness regulations.

*Description of the Type Design Change:* Installation of J.P. Instruments (JPI) fuel flow indicating system in accordance with JPI Installation Manual FS-450, Report No. 400, Revision -, dated August 16, 2000, or later FAA approved revision.

*Limitations and Conditions:* Approval of this change in type design applies to the aircraft models listed on the AML only. This approval should not be extended to other aircraft of these models on which other previously approved modifications are incorporated unless it is determined that the relationship between this change and any of those other previously approved modifications, including changes in type design, will introduce no adverse effect upon the airworthiness of that aircraft. A copy of this Certificate, FAA Approved Model List (AML) No. SA00861SE, and Airplane Flight Manual Supplement No. 1, dated December 18, 2000, or later FAA approved revision, must be maintained as part of the permanent records for the modified aircraft.

Note: This STC requires the installation of a fuel flow transducer per STC SA00432SE; or aircraft listed on the FAA approved AML SA00432SE and that have been previously modified with a fuel flow indication system that utilizes the Flowscan fuel flow transducer, P/N: 201-A, 201-B, 201-C or 231 are eligible for installation for the FS-450. This certificate does not constitute installation approval of the fuel flow transducer.

If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

*This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.*

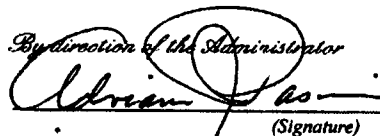
*Date of application:* April 7, 2000

*Date reissued:*

*Date of issuance:* December 18, 2000

*Date amended:*



*By direction of the Administrator*  
  
(Signature)

Acting Manager, Seattle Aircraft  
Certification Office

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.



**J.P. INSTRUMENTS FAA APPROVED MODEL LIST for STC SA00861SE  
FOR  
INSTALLATION OF THE (FS-450) 450000 SERIES FUEL FLOW INSTRUMENT**

Issue Date: December 18, 2000

ITEM	AIRCRAFT MAKE	AIRCRAFT MODEL	TYPE CERTIFICATE NUMBER	CERTIFICATION BASIS FOR ALTERATION	AML REVISION DATE
41.	Cessna	177RG	A20CE	FAR 23	12-18-2000
42.	Cessna	180, 180A, B, C, D, E, F, G, H, J, K	5A6	CAR 3	12-18-2000
43.	Cessna	182, 182A, B, C, D, E, F, G, H, J, K, L, M, N, P, Q, R	3A13	CAR 3	12-18-2000
44.	Cessna	182E, F, G, H, J, K, L, M, N, P, Q (Peterson, STC SA3825SW)	3A13	CAR 3	5-31-2001
45.	Cessna	182T	3A13	CAR 3	5-31-2001
46.	Cessna	T182T	3A13	CAR 3	5-31-2001
47.	Cessna	182 Series Air Planes Corp, STC SA00152WI Engine IO-520 and IO-550	3A13	CAR3	12-18-2000
48.	Cessna	182S	3A13	FAR 23	12-18-2000
49.	Cessna	R182, T182, TR182	3A13	CAR 3	12-18-2000
50.	Cessna	185, 185A, B, C, D, E, A185E, A185F	3A24	CAR 3	12-18-2000
51.	Cessna	188, 188A, 188B, A188, A188A, A188B, T188C	A9CE	FAR 21	12-18-2000
52.	Cessna	195, A, B	A-790	CAR 3	12-18-2000
53.	Cessna	206, 206, P206A, B, C, D, E, TP206A, B, C, D, E, U206, U206A, B, C, D, E, F, G, TU206A, B, C, D, E, F, G	A4CE	CAR 3	12-18-2000
54.	Cessna	206H	A4CE	CAR 3	5-31-2001
55.	Cessna	T206H	A4CE	FAR 23	12-18-2000
56.	Cessna	U206A, B, C, D, E, F, G. (with Lycoming TIO-540- series engine per STC SA2123NM)	A4CE	CAR 3	12-18-2000
57.	Cessna	207, 207A, T207, T207A	A16CE	FAR 23	12-18-2000
58.	Cessna	210, 210A, B, C, D, E, F 210-5, 210-5A, T210F, 210G, T210G, 210H, T210H, 210J, T210J, T210K, 210K, 210L, T210L, 210M, T210M, 210N, P210N, T210N, 210R, P210R, T210R	3A21	FAR 23	12-18-2000
59.	Cessna	T303	A34CE	FAR 23	1-27-2003
60.	Cessna	310, C, D, E, F, G, H, E310H, 310I, J, K, L, N, P, Q, R 310J-1, E310J, T310P, Q, R, T310Q, 310R, T310R	3A10	CAR 3	12-18-2000
61.	Cessna	320, A, B, C, D, E, F, 320-1, 335, 340, 340A	3A25	CAR 3	12-18-2000
62.	Cessna	336	A2CE	CAR 3	12-18-2000



**J.P.INSTRUMENTS**  
**Supplement No. 1**  
**P.O. Box 7033**  
**Huntington Beach, CA 92646**

**FAA APPROVED**  
**AIRPLANE/ROTORCRAFT FLIGHT MANUAL SUPPLEMENT OR**  
**SUPPLEMENTAL AIRPLANE FLIGHT MANUAL (INCLUDING POH AND**  
**FAA AFM)**  
**(FOR THOSE AIRCRAFT WITHOUT A BASIC AIRPLANE FLIGHT**  
**MANUAL)**

**FS-450 FUEL FLOW INDICATOR**  
**FOR**

**Single and Twin, Reciprocating Engine Powered Aircraft as listed**  
**On Approved Model List of**

**STC SA00861SE**

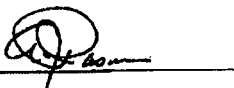
**REG. NO.** N310CM

**SER. NO.** 310P-0178

This Supplement must be attached to the  
**FAA Approved Airplane/Rotorcraft Flight Manual** when the J.P. Instruments FS-450  
is installed in accordance with Supplemental Type Certificate SA 00432SE. For those  
airplanes without a basic Airplane Flight Manual, the Supplemental AFM must be in the  
aircraft when the FS-450 is installed.

The information contained in this **Airplane/Rotorcraft Flight Manual Supplement/**  
**Supplemental Aircraft Flight Manual** supplements or supersedes the basic manual/  
placards only in those areas listed. For limitations, procedures and performance information  
not contained in this supplement, consult the basic Airplane Flight manual, Markings and  
Placards

**FAA APPROVED:**



**Manager, Special Certification Branch, ANM-190S**  
**Federal Aviation Administration**  
**Seattle Aircraft Certification Office**  
**Transport Airplane Directorate**

**Date:** December 18, 2000



***J.P.Instruments Fuel Flow Installation Manual***  
***FS-450***

Report # 400  
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Rev A : Date 02/1/2002

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J.P.INSTRUMENTS  
PO BOX 7033  
HUNTINGTON BEACH CA 92646



# ***J.P.Instruments Fuel Flow Installation Manual***

FS-450

FAA Approved

Report # 400

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Rev B : Date 12/12/2012

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J.P.INSTRUMENTS  
PO BOX 7033  
HUNTINGTON BEACH CA 92704



# Pilot's Guide

## Fuel Scan

### FS-450M TWIN

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**J . P . INSTRUMENTS**

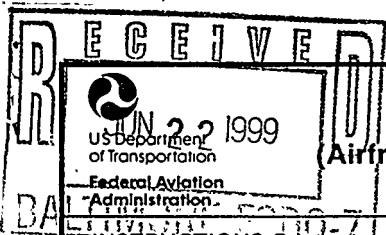
**Information:** P. O. Box 7033  
Huntington Beach, CA 92646

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Costa Mesa, CA 92626  
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[www.jp instruments.com](http://www.jp instruments.com)

Rev NC 11/03





**MAJOR REPAIR AND ALTERATION**  
**(Airframe, Powerplant, Propeller, or Appliance)**

Form Approved  
OMB No. 2120-0020

**For FAA Use Only**  
Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make <b>Cessna Aircraft</b>	Model <b>310P</b>
	Serial No. <b>310P-0178</b>	Nationality and Registration Mark <b>N310CM</b>
2. Owner	Name (As shown on registration certificate) <b>Williams, James A.</b>	Address (As shown on registration certificate) <b>P.O. Box 4541 Newark, DE. 19711</b>

**3. For FAA Use Only**

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				
POWERPLANT	<b>Lycoming</b>	<b>T10-540-A2B</b>	<b>L-4870-61 L-1275-61</b>		<b>X</b>
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

**6. Conformity Statement**

A. Agency's Name and Address <b>Gilbert G. Raith 2326 Seavies Rd. Balto. MD. - 21222-3217</b>	B. Kind of Agency <input checked="" type="checkbox"/> U.S. Certified Mechanic <input type="checkbox"/> Foreign Certified Mechanic <input type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. <b>A+P 1169441</b>
--	---	--

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date <b>15 April 99</b>	Signature of Authorized Individual <b>Gilbert G. Raith</b>
----------------------------	---

**7. Approval for Return To Service**

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ **APPROVED** ☐ **REJECTED**

BY	FAA Fit. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/>	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection <b>20 June 99</b>		Certificate or Designation No. <b>A+P 1169441</b>		Signature of Authorized Individual <b>Gilbert G. Raith</b>	

### NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

#### 8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Removed Prestolite starters Model-MHB4018 and installed Sky-Tec starters Model-149(V)PM in accordance with mfg.'s installation/instructions dated: 19 Jan. 98, and approved per STC 00218 NY.


Enter corrections to equipment list and weight and balance records

Make necessary maintenance record entry in log-book.

Operational grd. check-out before return to active service.

☐ Additional Sheets Are Attached

SEP 21 1992

 <b>MAJOR REPAIR AND ALTERATION</b> <b>(Airframe, Powerplant, Propeller, or Appliance)</b>		Form Approved OMB No. 2120-0020 For FAA Use Only <b>BAL</b> Office Identification <b>FSDO-7</b>	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).			
1. Aircraft	Make CESSNA	Model 310P	
	Serial No. 310P-0178	Nationality and Registration Mark USA N310CM	
2. Owner	Name (As shown on registration certificate) WILLIAMS JAMES A		Address (As shown on registration certificate) PO BOX 4541 NEWARK DE 19711
	3. For FAA Use Only		
4. Unit Identification			
Unit	Make	Model	Serial No.
AIRFRAME	(As described in Item 1 above)		
POWERPLANT			
PROPELLER			
APPLIANCE	Type		
	Manufacturer		
5. Type			
6. Conformity Statement			
A. Agency's Name and Address		B. Kind of Agency	
COLLEGE PARK AVIONICS, INC. 6709 CPL FRANK SCOTT DRIVE COLLEGE PARK, MD 20740		<input type="checkbox"/> U.S. Certificated Mechanic	
		<input checked="" type="checkbox"/> Foreign Certificated Mechanic	
		<input checked="" type="checkbox"/> Certificated Repair Station	
		<input type="checkbox"/> Manufacturer	
		C. Certificate No. LI1R297K	
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.			
Date 09/15/92		Signature of Authorized Individual RANDALL D. COX <i>R. D. Cox</i>	
7. Approval for Return To Service			
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED			
BY	FAA FTL Standards Inspector	Manufacturer	Inspection Authorization
	FAA Designee	<input checked="" type="checkbox"/> Repair Station	Person Approved by Transport Canada Airworthiness Group
Date of Approval or Rejection 09/15/92		Certificate or Designation No. LI1R297K	Signature of Authorized Individual RANDALL D. COX <i>R. D. Cox</i>

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

**8. Description of Work Accomplished**

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

- 1.) Removed the following:  
NARCO MK 16 NAV/COM  
NARCO MP 16 CONVERTER  
NARCO UGR 2 G/S RECEIVER  
BENDIX/KING KN 71 CONVERTER  
BENDIX/KING KI 206 INDICATOR  
2.) Installed the following:  
BENDIX/KING KX 155 NAV/COM  
BENDIX/KING KI 209 INDICATOR

3.) The above equipment installed in accordance with the manufacturer's installation manual, BENDIX/KING p/n 006-0179-05.

4.) The above equipment installed in accordance with AC 43.13-1A, Ch. 11, 13, 15; AC 43.13-2A, Ch. 2, 3.

5.) Equipment checked with no adverse interference to any other aircraft system.

6.) Maximum continuous electrical load does not exceed 80% of the alternators capacity.

7.) Weight and balance and equipment list revised to reflect these changes.

8.) For further details see work order no. OCM-2250, this repair station.

//////**END**//////

☐ Additional Sheets Are Attached

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				NOV 5 1986		Form Approved Budget Bureau No. 04-R060.1	
<b>MAJOR REPAIR AND ALTERATION</b> (Airframe, Powerplant, Propeller, or Appliance)						FOR FAA USE ONLY	
						OFFICE IDENTIFICATION <i>AEA Gado-3</i>	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.							
1. AIRCRAFT	MAKE CESSNA			MODEL 310 P			
	SERIAL NO 310P0178			NATIONALITY AND REGISTRATION MARK N 310 CM			
2. OWNER	NAME (As shown on registration certificate) ALPINE AVIATION			ADDRESS (As shown on registration certificate) Lisbon, New Hampshire			
3. FOR FAA USE ONLY							
4. UNIT IDENTIFICATION							
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE			
				REPAIR	ALTERATION		
AIRFRAME	As described in item 1 above				X		
POWERPLANT							
PROPELLER							
APPLIANCE	TYPE						
	MANUFACTURER						
6. CONFORMITY STATEMENT							
A. AGENCY'S NAME AND ADDRESS				B. KIND OF AGENCY		C. CERTIFICATE NO.	
Stephen K Demarest RD 3, Box 3 Sussex, New Jersey 07461				<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC		2077558	
				<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC			
				<input type="checkbox"/> CERTIFICATED REPAIR STATION			
				<input type="checkbox"/> MANUFACTURER			
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.							
DATE 11/3/86				SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Stephen K. Demarest</i>			
7. APPROVAL FOR RETURN TO SERVICE							
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED							
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/>	INSPECTION AUTHORIZATION		OTHER (Specify)	
	FAA DESIGNEE	REPAIR STATION	<input type="checkbox"/>	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT			
DATE OF APPROVAL OR REJECTION 11/3/86		CERTIFICATE OR DESIGNATION NO. 2077558		SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Stephen K. Demarest</i>			

### NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

- Airframe modified in accordance with STC SA4213 WE, held by Bob Fields Aerocessories, by installation of inflatable door seal in accordance with Drawing 3920 and instructions issued with the STC.

NOTHING FOLLOWS

☐ ADDITIONAL SHEETS ARE ATTACHED

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				AUG 20 1980	Form Approved Budget Bureau No. 04-R060.1
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION RSC GADO T-2-14	
INSTRUCTIONS: Print or type all entries. See FAR 43.9; FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE <b>Cessna</b>	MODEL <b>310P</b>			
	SERIAL NO. <b>310P-0178</b>	NATIONALITY AND REGISTRATION MARK <b>N310CM</b>			
2. OWNER	NAME (As shown on registration certificate) <b>RICHARD GRAY</b>		ADDRESS (As shown on registration certificate) <b>19300 MT. ELLIOTT DETROIT, MI. 48234</b>		
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
				REPAIR	ALTERATION
AIRFRAME	***** (As described in item 1 above) *****				<b>X</b>
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS			B. KIND OF AGENCY		C. CERTIFICATE NO.
<b>Vee Neal, Inc. R.D. #1 Box 397 Latrobe, PA 15650</b>			U.S. CERTIFICATED MECHANIC		<b>114-30</b>
			FOREIGN CERTIFICATED MECHANIC		
			<input checked="" type="checkbox"/> CERTIFICATED REPAIR STATION		
			MANUFACTURER		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE <b>8-15-80</b>			SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Stephen C. Agnello</i>		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION		OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION <b>8-15-80</b>		CERTIFICATE OR DESIGNATION NO. <b>114-30</b>	SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Stephen C. Agnello</i>		

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

**8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)**

Installed this date Ryan WX-7A weather mapping system.

Receiver (model 175C27) and display (model 275) installed in instrument panel as per AC 43.13-2, Fig. 2-2.

Processor (model 377) remote mounted on fwd. Avionics shelf as per AC 43.13-2, Fig. 3.

Loop antenna (model 277) installed on bottom of A/C as per AC 43.13-2, Fig 3.15.

System wired and installed as per installation manual #377A200.

Circuit breakers and wiring is provided for all units installed as recommended by manufacturer.

Compass deviation well within limits of AC 43.13-2, Chapter 2, Part F. Alternator load well within 80% capacity.

Appropriate log book entries were made. Weight and balance revised.

END

☐ ADDITIONAL SHEETS ARE ATTACHED

FEDERAL AVIATION AGENCY MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				Form Approved Budget Bureau No. 04-R060.1 FOR FAA USE ONLY OFFICE IDENTIFICATION	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE <b>Cessna</b>	MODEL <b>310P</b>		NATIONALITY AND REGISTRATION MARK <b>N 310CH</b>	
	SERIAL NO. <b>310P0178</b>				
2. OWNER	NAME (As shown on registration certificate) <b>Biley Turbo Sales</b>		ADDRESS (As shown on registration certificate) <b>2900 Preston Tower Dallas, Texas 75225</b>		
	3. FOR FAA USE ONLY				
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	(As described in item 1 above)			REPAIR	ALTERATION
POWERPLANT					<b>X</b>
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS <b>Avion Aero Corp. P. O. Box 8 Addison, Texas 75001</b>			B. KIND OF AGENCY <input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC <input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC <input type="checkbox"/> CERTIFICATED REPAIR STATION <input type="checkbox"/> MANUFACTURER		C. CERTIFICATE NO. <b>Willard A. Webb A &amp; P 494429</b>
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE <b>November 14, 1969</b>		SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Willard A. Webb</i> <b>Willard A. Webb</b>			
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION		OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION <b>11-14-69</b>		CERTIFICATE OR DESIGNATION NO. <b>494485</b>	SIGNATURE OF AUTHORIZED INDIVIDUAL <i>J. R. Campbell</i>		

# NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed Narco GLC-60 course line computer system. Installed two Narco Mark 16 Nav/Com sets with one VOA-50H navigation indicator, one Narco MHT-24 Marker Receiver, QAR Radio altimeter, IRE indicator, Narco transponder AT6-A, panel unit installed Bendix T-12-C receiver and indicator, Narco GLC display and off set control head. Above equipment located in center radio panel space factory provided. Two Narco UGR-2 Glide Slope Receivers and two Narco HP-16 converters, installed in factory radio rack, left hand side of nose compartment, course line computer, located on bottom radio rack left nose section/

Installed Narco UHA-1 Marker Antenna, ARCA-15 VHF Com/Antenna aft of rear spar on belly, VHA-10 antenna on top of vertical fin. Installed King KA-21A VHF com antenna top of fuselage, Narco, UDA-1000000000 antenna lower left hand nose section, installed RCA IRE antenna bottom of right center section. Installed UDI-2AR units right side in forward radio rack factory provided with UDI-2R control in radio panel. Installed ground avoidance radar altimeter indicator in radio panel, installed unit in tail section aft of baggage compartment. Installed Bendix ADF loop model 2321 bottom of fuselage, Narco AT6A Flightguard Transponder located left nose section. All equipment installed as per manufacturers installation manual. Manufacturers wire harness and MIL-W-5086 wiring used. All work performed in accordance with AC43-13-1, chapters 2, 5, & 11, AC43-13-2, chapters 1, 2, & 3. This aircraft is equipped with two 50 amp alternators and the load is in accordance with AC43-13-1, chapters 11, section 2.

## WEIGHT AND BALANCE COMPUTATION

Cessna 310CH

S/N 310P0178

Item	Weight (lbs)	Arm (Inches)	Moment (lbs. x inches)
Aircraft empty	3520.	35.8	126,320
VHA-1 Marker Antenna	+1.0	+165.	+165.
IRE Antenna	.5	+12.	+6.
King KA-21 Com Ant.	2.0	+48	+96.
ARCA-15 Com Antenna	.8	+75	+60.
VHA-10 Antenna	1.0	+224	+224.
QAR Transceiver Ant.	1.25	+141	+176.25
UDI-2AR Units	14.3	-22	-314.6
Narco VHF-37 Ant. Reg	.8	+234	+187.2
MHT-24 Marker Beacon	1.5	-18.	-27.0
KA-11 isolation amp.	1.0	-28	-28.0
two each URG-2UHF			
Glide Slope Receivers	4.70	-10	-47.
HP16 Converter	2.00	-28	-56.
HP16 Converter	2.00	-17	-34.
Narco flight guard	5.3	-28	-148.4

2 ADDITIONAL SHEETS ARE ATTACHED

Riley Turbo Sales  
2900 Preston Tower  
Dallas, Texas 75225

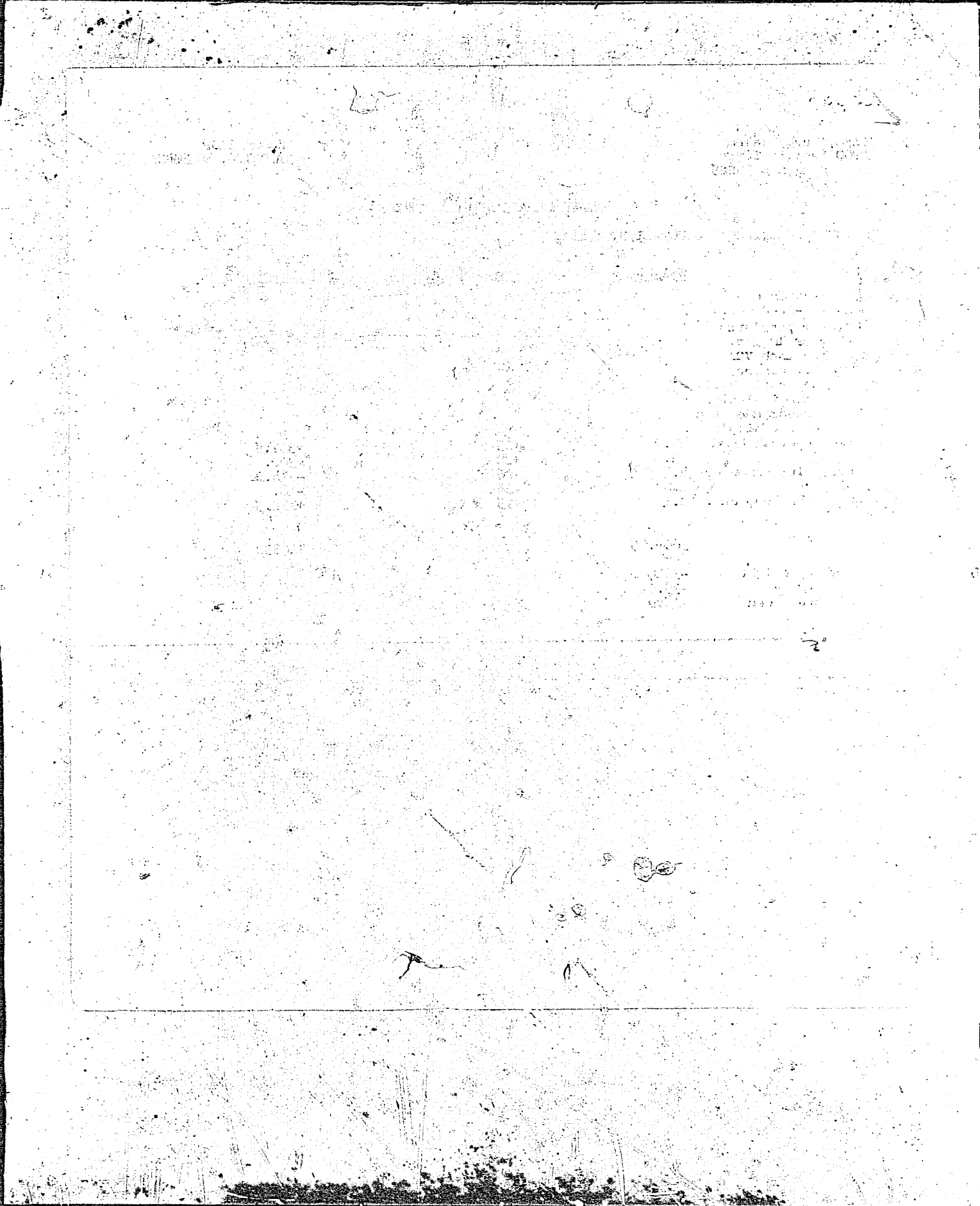
Cessna 310P  
310P0178, N 310CH

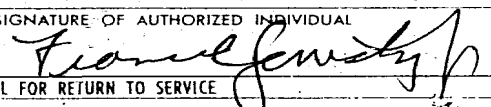
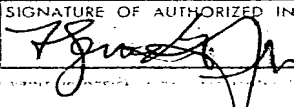
"MAJOR REPAIR AND ALTERATION"

WEIGHT AND BALANCE COMPUTATION, cont'd

Item	Weight	Arm	Moment
Equipment in center			
Bendix T-12-C rec.+			
indicator panel-two			
each Mark 16 nav.			
com, two each VOA			
-11s nav. conver-			
ters, GAR radio alti-			
meter, DME indicator,			
Narco transponder AT6-			
A, Narco cdc display,			
Offset control head	29.7	+15	+367.5
Wire	15.0	+20	+300.
Course line computer	4.4	-26	-114.0
Radio Rack	1.0	-26	-26.
Bendix ADF loop ant.	1.5	+75	+112.5
	3609.75		126332
New emp y weight	-3610		
New emp CG	-35.0		
New useful load	-1590		

END



DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION <b>MAJOR REPAIR AND ALTERATION</b> (Airframe, Powerplant, Propeller, or Appliance)				Form Approved Budget Bureau No. 04-R060.1 FOR FAA USE ONLY OFFICE IDENTIFICATION NE-CAO-19	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT MAKE Cessna SERIAL NO. 310P-0178		MODEL T310P NATIONALITY AND REGISTRATION MARK N310CM			
2. OWNER NAME (As shown on registration certificate) Alpine Aviation		ADDRESS (As shown on registration certificate) 51 Highland Ave. Lisbon, NH 03585			
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					5. TYPE
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	***** (As described in item 1 above) *****				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS		B. KIND OF AGENCY		C. CERTIFICATE NO	
Northeastern Avionics, Inc. Barnes Municipal Airport Westfield, MA 01085		U.S. CERTIFICATED MECHANIC		E19-49	
		FOREIGN CERTIFICATED MECHANIC			
		X CERTIFICATED REPAIR STATION			
		MANUFACTURER			
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE 3-29-83		SIGNATURE OF AUTHORIZED INDIVIDUAL 			
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION		OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION 3-29-83		CERTIFICATE OR DESIGNATION NO. E19-49		SIGNATURE OF AUTHORIZED INDIVIDUAL 	

### NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. King KNS-80 Integrated Navigation System previously installed with King KI-206 by another agency.
2. Conformity inspection performed by this agency this date in accordance with FAA approved Form 337 dated 1/19/79 for aircraft N5881N S/N 14426.
3. This installation conforms to the procedure outlined in Northeastern Avionics, Inc. - RNAV Installation Procedure Manual as approved by NE GADO letter dated August 28, 1979.
4. The RNAV system is approved for VFR, IFR enroute and approach modes.

END

☐ ADDITIONAL SHEETS ARE ATTACHED

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1 FOR FAA USE ONLY OFFICE IDENTIFICATION <b>GL-GADO-20</b>	
<b>MAJOR REPAIR AND ALTERATION</b> (Airframe, Powerplant, Propeller, or Appliance)					
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE <b>CESSNA</b>		MODEL <b>310 P</b>		
	SERIAL NO. <b>0178</b>		NATIONALITY AND REGISTRATION MARK <b>310CH</b>		
2. OWNER	NAME (As shown on registration certificate) <b>GRAY, RAPAPORT, MARTIN</b>		ADDRESS (As shown on registration certificate) <b>19300 MT ELLIOTT DETROIT MICH 48234</b>		
	3. FOR FAA USE ONLY				
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS.			B. KIND OF AGENCY		C. CERTIFICATE NO.
<b>JOHN F. JOHNS 2135 N TIRWAY PONTIAC MICH 48054</b>			<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC		<b>1519936</b>
			<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC		
			<input type="checkbox"/> CERTIFICATED REPAIR STATION		
			<input type="checkbox"/> MANUFACTURER		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE <b>3-10-80</b>			SIGNATURE OF AUTHORIZED INDIVIDUAL <i>John F. Johns</i>		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION		OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION <b>3-10-80</b>		CERTIFICATE OR DESIGNATION NO. <b>1519936</b>		SIGNATURE OF AUTHORIZED INDIVIDUAL <i>John F. Johns</i>	

# NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

## REPAIRED AIRCRAFT NOSE SECTION BY REPLACING THE FOLLOWING PARTS

0813300-1 GEAR DOOR  
0813043-11 GEAR DOOR  
0813300-2 GEAR DOOR  
0813022-3 BULKHEAD  
0813000-171 SUPPORT  
0811100-14 STRINGER  
0813000-187 FORMER  
0813000-35 BRACKET  
0813000-59 bracket  
0812102-2 INNER  
0812121-1 TUBE  
0810125-6 ROD  
0813001-11 SKIN  
0813001-15 SKIN  
0813300-77 ROD  
0813045-1 BULKHEAD

ALL WORK DONE IN ACCORDANCE WITH PROCEDURES RECOMMENDED IN MANUAL,

43-13 SEC 3

	CHIEF	
A D M	Received GADO 20 MAR 12 1980 Ypsilanti, Michigan	A P S
OPS	File	A/W ✓

☐ ADDITIONAL SHEETS ARE ATTACHED

MAR 30 1970

Form Approved  
Budget Bureau No. 04-R0058

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION APPLICATION FOR AIRWORTHINESS CERTIFICATE				INSTRUCTIONS—Print or type. Do not write in shaded areas; these are for FAA use only. Submit original only to an authorized FAA Representative. If additional space is required, use an attachment. For special flight permits complete Sections II and VI or VII as applicable.																																																																															
I. AIRCRAFT DESCRIPTION	1. REGISTRATION MARK	2. AIRCRAFT BUILDER'S NAME (make)	3. AIRCRAFT MODEL DESIGNATION	4. YR. MFG.	FAA CODING																																																																														
	N310CM	Cessna	310P	1969	2074238																																																																														
	5. AIRCRAFT SERIAL NO.	6. ENGINE BUILDER'S NAME (make)	7. ENGINE MODEL DESIGNATION	41533																																																																															
	310P-0178	Lycoming	TIO-540-A2B																																																																																
8. NUMBER OF ENGINES	9. PROPELLER BUILDER'S NAME (make)	10. PROPELLER MODEL DESIGNATION	11. AIRCRAFT IS																																																																																
2	Hartzell	HC-E3YR-2A/8468-6R	NEW <input checked="" type="checkbox"/> USED <input type="checkbox"/> IMPORT <input type="checkbox"/>																																																																																
APPLICATION IS HEREBY MADE FOR: (Check applicable items)																																																																																			
A <input checked="" type="checkbox"/> STANDARD AIRWORTHINESS CERT. (Indicate category) <input checked="" type="checkbox"/> NORMAL <input type="checkbox"/> UTILITY <input type="checkbox"/> ACROBATIC <input type="checkbox"/> TRANSPORT <input type="checkbox"/> GLIDER <input type="checkbox"/> BALLOON																																																																																			
B <input type="checkbox"/> SPECIAL AIRWORTHINESS CERTIFICATE (Check appropriate items)																																																																																			
<table border="1"> <tr> <td>2</td> <td>LIMITED</td> <td>1</td> <td>CLASS I</td> </tr> <tr> <td>5</td> <td>PROVISIONAL (Indicate class)</td> <td>2</td> <td>CLASS II</td> </tr> <tr> <td>3</td> <td>RESTRICTED (Indicate operation(s) to be conducted)</td> <td>3</td> <td>AGRICULTURE &amp; PEST CONTROL</td> </tr> <tr> <td></td> <td></td> <td>4</td> <td>FOREST (Wild life conservation)</td> </tr> <tr> <td></td> <td></td> <td>5</td> <td>PATROLLING</td> </tr> <tr> <td></td> <td></td> <td>6</td> <td>WEATHER CONTROL</td> </tr> <tr> <td></td> <td></td> <td>0</td> <td>OTHER (Specify)</td> </tr> <tr> <td>4</td> <td>EXPERIMENTAL (Indicate operation(s) to be conducted)</td> <td>1</td> <td>RESEARCH AND DEVELOPMENT</td> </tr> <tr> <td></td> <td></td> <td>2</td> <td>AMATEUR BUILT</td> </tr> <tr> <td></td> <td></td> <td>3</td> <td>EXHIBITION</td> </tr> <tr> <td></td> <td></td> <td>4</td> <td>RACING</td> </tr> <tr> <td></td> <td></td> <td>5</td> <td>CREW TRAINING</td> </tr> <tr> <td></td> <td></td> <td>6</td> <td>MKT. SURVEY</td> </tr> <tr> <td></td> <td></td> <td>0</td> <td>TO SHOW COMPLIANCE WITH FAR</td> </tr> <tr> <td>8</td> <td>SPECIAL FLIGHT PERMIT (Indicate operation to be conducted then complete Section VI or VII as applicable on reverse side)</td> <td>1</td> <td>FERRY FLIGHT FOR REPAIRS, ALTERATIONS, MAINTENANCE OR STORAGE</td> </tr> <tr> <td></td> <td></td> <td>2</td> <td>EVACUATE FROM AREA OF IMPENDING DANGER</td> </tr> <tr> <td></td> <td></td> <td>3</td> <td>OPERATION IN EXCESS OF MAX. CERTIFICATED TAKE-OFF WEIGHT</td> </tr> <tr> <td></td> <td></td> <td>4</td> <td>DELIVERING OR EXPORT</td> </tr> <tr> <td></td> <td></td> <td>5</td> <td>PRODUCTION FLIGHT TESTING</td> </tr> </table>								2	LIMITED	1	CLASS I	5	PROVISIONAL (Indicate class)	2	CLASS II	3	RESTRICTED (Indicate operation(s) to be conducted)	3	AGRICULTURE & PEST CONTROL			4	FOREST (Wild life conservation)			5	PATROLLING			6	WEATHER CONTROL			0	OTHER (Specify)	4	EXPERIMENTAL (Indicate operation(s) to be conducted)	1	RESEARCH AND DEVELOPMENT			2	AMATEUR BUILT			3	EXHIBITION			4	RACING			5	CREW TRAINING			6	MKT. SURVEY			0	TO SHOW COMPLIANCE WITH FAR	8	SPECIAL FLIGHT PERMIT (Indicate operation to be conducted then complete Section VI or VII as applicable on reverse side)	1	FERRY FLIGHT FOR REPAIRS, ALTERATIONS, MAINTENANCE OR STORAGE			2	EVACUATE FROM AREA OF IMPENDING DANGER			3	OPERATION IN EXCESS OF MAX. CERTIFICATED TAKE-OFF WEIGHT			4	DELIVERING OR EXPORT			5	PRODUCTION FLIGHT TESTING
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C <input checked="" type="checkbox"/> MULTIPLE AIRWORTHINESS CERTIFICATE (Check appropriate Restricted Operation and Standard or Limited as applicable above)																																																																																			
A. REGISTERED OWNER (As shown on Certificate of Aircraft Registration) IF DEALER, CHECK HERE <input type="checkbox"/>																																																																																			
NAME Riley Turbo Sales Corporation				ADDRESS 2900 Preston Tower, Dallas, Texas 75225																																																																															
B. AIRCRAFT CERTIFICATION BASIS (Check applicable blocks and complete items as indicated)																																																																																			
AIRCRAFT SPECIFICATION OR TYPE CERTIFICATION DATA SHEET (Give No. and Revision No.) 3A10-35				AIRCRAFT LISTING (Give page No(s).)																																																																															
X				X																																																																															
AIRCRAFT LISTING (Give page No(s).)				SUPPLEMENTAL TYPE CERTIFICATE (List number of each STC incorporated)																																																																															
				X SA2082WE																																																																															
C. AIRCRAFT OPERATION AND MAINTENANCE RECORDS																																																																																			
CHECK IF RECORDS IN COMPLIANCE WITH FAR 91.173		TOTAL AIRFRAME HOURS—Enter for used aircraft only		EXPERIMENTAL ONLY—Enter hours flown since last certificate issued or renewed																																																																															
X		13.7		3		1																																																																													
D. CERTIFICATION—I hereby certify that I am the owner (or his agent) of the aircraft described above; that the aircraft is registered with the Federal Aviation Administration in accordance with Section 501 of the Federal Aviation Act of 1958, and applicable Federal Aviation Regulations; and that the aircraft has been inspected and is airworthy and eligible for the airworthiness certificate requested.																																																																																			
DATE OF APPLICATION		NAME AND TITLE (Print or type)		SIGNATURE																																																																															
3/20/70		G. F. Adams Agent		G. F. Adams																																																																															
A. THE AIRCRAFT DESCRIBED ABOVE HAS BEEN INSPECTED AND FOUND AIRWORTHY BY: (Complete this section only if FAR 21.183 (d) applies)																																																																																			
2		FAR PART 121 OR 127 CERTIFICATE HOLDER (Give Certificate No.)		3		X CERTIFICATED MECHANIC (Give Certificate No.)																																																																													
5		AIRCRAFT MANUFACTURER (Give Name of Firm)		6		X CERTIFICATED REPAIR STATION (Give Certificate No.)																																																																													
A129789-40				6																																																																															
DATE		TITLE		SIGNATURE																																																																															
3/20/70		AI Chief Inspector 29789-40		G. F. Adams																																																																															
I have inspected the aircraft described in Item I or VII and find it meets the requirements for:																																																																																			
A. STANDARD AIRWORTHINESS CERTIFICATE				B. SPECIAL AIRWORTHINESS CERTIFICATE																																																																															
4				X C. AMENDMENT OR MODIFICATION OF ITS CURRENT AIRWORTHINESS CERTIFICATE																																																																															
DATE		DISTRICT OFFICE		DESIGNEE'S SIGNATURE AND NO.		FAA INSPECTOR'S SIGNATURE																																																																													
3-20-70		SW-EMDO-42		4		1																																																																													
		2-42				Clarence E. Knight																																																																													

<b>VI. PRODUCTION FLIGHT TESTING</b>	<b>A. MANUFACTURER</b>		
	NAME		ADDRESS
	<b>B. PRODUCTION BASIS (Check applicable item)</b>		
	<input type="checkbox"/> PRODUCTION CERTIFICATE (Give production certificate number) <input type="checkbox"/> TYPE CERTIFICATE ONLY <input type="checkbox"/> APPROVED PRODUCTION INSPECTION SYSTEM		
<b>VII. SPECIAL FLIGHT PERMIT PURPOSES OTHER THAN PRODUCTION FLIGHT TEST</b>	<b>C. GIVE QUANTITY OF CERTIFICATES REQUIRED FOR OPERATING NEEDS:</b>		
	DATE OF APPLICATION	NAME AND TITLE (Print or type)	SIGNATURE
	<b>A. DESCRIPTION OF AIRCRAFT</b>		
	REGISTERED OWNER		ADDRESS
	BUILDER (Make)		MODEL
	SERIAL NUMBER		REGISTRATION MARK
	<b>B. DESCRIPTION OF FLIGHT</b>		
	FROM		TO
	VIA		DEPARTURE DATE
			DURATION
	<b>C. CREW REQUIRED TO OPERATE THE AIRCRAFT AND ITS EQUIPMENT</b>		
	<input type="checkbox"/> PILOT <input type="checkbox"/> CO-PILOT <input type="checkbox"/> NAVIGATOR <input type="checkbox"/> OTHER (Specify)		
	<b>D. THE AIRCRAFT DOES NOT MEET THE APPLICABLE AIRWORTHINESS REQUIREMENTS AS FOLLOWS:</b>		
	<b>E. THE FOLLOWING RESTRICTIONS ARE CONSIDERED NECESSARY FOR SAFE OPERATION (Use attachment if necessary)</b>		
<b>F. CERTIFICATION</b> —I hereby certify that I am the registered owner (or his agent) of the aircraft described above; that the aircraft is registered with the Federal Aviation Administration in accordance with Section 501 of the Federal Aviation Act of 1938, and applicable Federal Aviation Regulations; and that the aircraft has been inspected and is airworthy for the flight described.			
DATE	NAME AND TITLE (Print or type)	SIGNATURE	
<b>VIII. AIRWORTHINESS DOCUMENTATION (FAA use only)</b>			
<input checked="" type="checkbox"/>	A. Operating Limitations and Markings in Compliance with FAR 91.31 as Applicable	<input checked="" type="checkbox"/> G. Statement of Conformity, FAA Form 317 (Attach when required)	
<input type="checkbox"/>	B. Current Operating Limitations Attached	<input type="checkbox"/> H. Foreign Airworthiness Certification for Import Aircraft (Attach when required)	
<input type="checkbox"/>	C. Data, Drawings, Photographs, etc. (Attach when required)	<input type="checkbox"/> I. Previous Airworthiness Certificate Issued in Accordance with FAR 21.191 CAR (Original attached)	
<input checked="" type="checkbox"/>	D. Current Weight and Balance Information Available in Aircraft	<input checked="" type="checkbox"/> J. Current Airworthiness Certificate Issued in Accordance with FAR 21.183 (Copy attached)	
<input checked="" type="checkbox"/>	E. Major Repair and Alteration, FAA 337 (Attach when required)		
<input checked="" type="checkbox"/>	F. This Inspection Recorded in Aircraft Records		

DEPARTMENT OF TRANSPORTATION  
FEDERAL AVIATION ADMINISTRATION  
SOUTHWEST REGION  
FORT WORTH, TEXAS

283-2846

SW EMDO  
2-0-42EXPERIMENTAL OPERATING LIMITATIONSIssue Date Oct. 9, 1969 Expiration Date Jan. 10, 1970MAKE Cessna MODEL 310P SERIAL NO. 310P-0178 REG. NO. N310CM

1. This aircraft is certificated in the Experimental classification for the purpose of: To Show Compliance with F.A.R.
2. The aircraft may not be operated for other than the purpose for which the certificate was issued.
3. The aircraft is to be identified "experimental" in accordance with FAR 45.23.
4. Persons or property shall not be carried for compensation or hire.
5. Occupancy of the aircraft is restricted to persons essential to the flight test involved.
6. All flights shall be conducted under Visual Flight Rules (VFR).  
Day or Night
7. All flights shall be conducted in accordance with the applicable FAA air traffic and general operation rules of FAR Part 91.
8. The control tower shall be notified of the experimental nature of the aircraft when operating into or out of airports with operating control towers.
9. All flights shall be conducted within the following boundaries and/or corridors: N/A
10. This aircraft does not comply with the International Airworthiness Standards of ICAO, Annex 8, Part II.

*Charles E. Harris*  
Charles E. Harris  
FAA Representative

SW-EMDO-42  
Designation

MICRO

APR 21 1970

UNITED STATES OF AMERICA  
DEPARTMENT OF TRANSPORTATION  
FEDERAL AVIATION ADMINISTRATION  
STATEMENT OF CONFORMITYForm Approved Budget  
Bureau No. 04-R0146FTW EMDQ  
2-0-42

SECTION I - AIRCRAFT	
1. MAKE <b>Cessna</b>	2. MODEL <b>310P</b>
3. SERIAL NO. <b>310P-0178</b>	4. REGISTRATION NO. <b>N310CM</b>
SECTION II - ENGINE	
1. MAKE	2. MODEL
3. SERIAL NO.	
SECTION III - PROPELLER	
1. MAKE	2. HUB MODEL
2. BLADE MODEL	4. HUB SERIAL NO.
5. BLADE SERIAL NOS.	
SECTION IV - CERTIFICATION	
I hereby certify that:	
<input checked="" type="checkbox"/> A. I have complied with Section 21.33(a).	
<input checked="" type="checkbox"/> B. The aircraft described above, produced under type certificate only (FAR 21 Subpart F), conforms to its type certificate, is in a condition for safe operation, and was flight checked on _____ (Date)	
<input type="checkbox"/> C. The engine or propeller described above, presented herewith for type certification, conforms to the type design therefor.	
<input type="checkbox"/> D. The engine or propeller described above produced under type certificate only (FAR 21 Subpart F), conforms to its type certificate and is in a condition for safe operation. The engine or, if applicable, the variable pitch propeller was subjected by the manufacturer to a final operational check on _____ (Date)	
Deviations: <b>Install Lycoming TIO-540-A2B engines and Hartzell HC-E3YR-2A/C8468-6R Under Turbo Propulsion Corp. STC SA 2082 WE "H" &amp; "J" summary.</b>	
SIGNATURE OF CERTIFIER <b>G. F. Adams</b> <i>G. F. Adams</i>	TITLE <b>AI 29789-40 Chief Inspector</b>
ORGANIZATION <b>Turbo Propulsion Corp.</b>	DATE <b>3/20/70</b>

# **MERO** INSTRUCTIONS

This form should be submitted to a representative of the Administrator under the following circumstances:

1. By the applicant for a type certificate or a supplemental type certificate at the time he presents an aircraft or parts thereof to the FAA for tests.
2. By the applicant for a type certificate or a supplemental type certificate for each engine or propeller submitted for type certification.
3. By the type certificate holder or licensee manufacturing products under a type certificate only, upon the initial transfer by him of the ownership of each product or upon application for the original issue of an aircraft airworthiness certificate, or an Airworthiness Approval Tag (FAA Form 186).

This form should be completed as follows:

**Section I. Aircraft.** Complete the pertinent part of only this section when certification covers an aircraft or part thereof.

**Section II. Engine.** Complete this section when certification covers an engine.

**Section III. Propeller.** Complete this section when certification covers a propeller.

## **Section IV. Certification.**

**Item A.** Check this block when an aircraft or part thereof is presented for flight or ground tests during type certification or supplemental type certification.

**Item B.** Check this block when the holder or licensee of a type certificate only, initially transfers the ownership of an aircraft manufactured under that type certificate, or applies for the original issuance of an airworthiness certificate.

**Item C.** Check this block when an engine or propeller is presented for type certification.

**Item D.** Check this block when an engine or propeller is presented for airworthiness approval and insert the date the product completed a final operational check.

The certification must be signed by an authorized person who holds a responsible position in the manufacturing organization.

FAA AC F89-1738

Punched by Operator #20

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
<b>MAJOR REPAIR AND ALTERATION</b> (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1. (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE <b>Cessna</b>		MODEL <b>310P</b>		
	SERIAL NO. <b>310P-0178</b>		NATIONALITY AND REGISTRATION MARK <b>N310CM</b>		
2. OWNER	NAME (As shown on registration certificate) <b>R.M. Gray/Ray Rapaport &amp; Michael Martin</b>		ADDRESS (As shown on registration certificate) <b>19300 Mt. Elliott Detroit, Mich. 48234</b>		
3. FOR FAA USE ONLY					
The data/alteration identified herein complies with the applicable airworthiness requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in FAR 43 Section 43.7.					
DATE: <b>2/5/79</b> SIGNATURE OF FAA INSPECTOR: <i>LC. Ferguson AGL-Gado-20</i>					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
POWERPLANT					X
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS			B. KIND OF AGENCY		C. CERTIFICATE NO.
QCE, INC. 6544 Highland Rd. Pontiac, Mich. 48042			<input type="checkbox"/> U.S. CERTIFICATED MECHANIC		C63-38
			<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC		
			<input checked="" type="checkbox"/> CERTIFICATED REPAIR STATION		
			<input type="checkbox"/> MANUFACTURER		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE <b>February 5, 1979</b>			SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Earl L. Wilson</i>		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION		OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION <b>2-5-79</b>		CERTIFICATE OR DESIGNATION NO. <b>C63-38</b>		SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Earl L. Wilson</i>	

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

- (1) Installed King KNS-80 Digital Area Navigation System in accordance with King installation manual 006-0154-00. Dated 7-22-76. Other manuals used, include King KI-206 and manual 006-0137-00.
- (2) Bench & ground tests were conducted in accordance with manufacturers recommendations & AC 90-45A, para.3.
- (3) The system was Flight tested by QCE personnel over the course, as on file at G.A.D.O. # 20.
- (4) All work accomplished in accordance with AC 43.13-1 & 2.
- (5) Weight and Balance computed & entered in A/C permanent file.

☐ ADDITIONAL SHEETS ARE ATTACHED

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION GL-GADU	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE	Cessna		MODEL	310P
	SERIAL NO.	310P-0178		NATIONALITY AND REGISTRATION MARK	N310CM
2. OWNER	NAME (As shown on registration certificate) R.M. Gray, Ray Rapaport, and Michail Martin			ADDRESS (As shown on registration certificate) 19300 Mt. Elliott Detroit Michigan 48234	
	3. FOR FAA USE ONLY				
4. UNIT IDENTIFICATION					5. TYPE
UNIT	MAKE	MODEL		SERIAL NO.	REPAIR ALTERATION
AIRFRAME	***** (As described in item 1 above) *****				
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS			B. KIND OF AGENCY		C. CERTIFICATE NO.
Michail Martin 8100 Narhan Oscar Dr. Clarkson, Michigan 48016			<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC		A/P 1606218
			<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC		
			<input type="checkbox"/> CERTIFICATED REPAIR STATION		
			<input type="checkbox"/> MANUFACTURER		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE 3-5-78			SIGNATURE OF AUTHORIZED INDIVIDUAL Michail Martin		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/>	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	
DATE OF APPROVAL OR REJECTION 3-7-78		CERTIFICATE OR DESIGNATION NO. A/P 1539234		SIGNATURE OF AUTHORIZED INDIVIDUAL Donald F Moore Donald F Moore Inspection Authorization	

# **NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed wing locker fuel system L.H. and R.H. in accordance with Cessna approved accessory Kit AK 320-I67J, L.H. installation AK 320-I68K, R.H. installation. AK 402-I49A, Partial plumbing R.H. AK 402-I47A, Partial plumbing L.H. And installation Manuals.

Complete system operational check normal and pressure leak tested.

Pilot check list, owners manual supplement, and fuel selector placards, installed in proper order.

Aircraft log book entry and Wt. & Balance report and equipment list revised.

All reworked structure prepaired and painted with No. 6I60 Matterhorn White Alumigrip Enamel.

END.

☐ ADDITIONAL SHEETS ARE ATTACHED

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION <b>MAJOR REPAIR AND ALTERATION</b> (Airframe, Powerplant, Propeller, or Appliance)		24 NOV 1976 Form Approved Budget Bureau No. 04-R060.1 FOR FAA USE ONLY OFFICE IDENTIFICATION 1203
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.		
1. AIRCRAFT	MAKE <b>Cessna</b>	MODEL <b>310P</b>
	SERIAL NO. <b>310P-0178</b>	NATIONALITY AND REGISTRATION MARK <b>N310CM</b>
2. OWNER	NAME (As shown on registration certificate) <b>Clymer, Lester V.</b>	
	ADDRESS (As shown on registration certificate) <b>950 California Rd. Quakertown, Pa. 18951</b>	
3. FOR FAA USE ONLY		
4. UNIT IDENTIFICATION		
UNIT	MAKE	MODEL
AIRFRAME	(As described in item 1 above)	
POWERPLANT		
PROPELLER		
APPLIANCE	TYPE	
	MANUFACTURER	
5. TYPE		
	REPAIR	ALTERATION
		<b>X</b>
6. CONFORMITY STATEMENT		
A. AGENCY'S NAME AND ADDRESS		B. KIND OF AGENCY
<b>Reading Aviation Service, Inc.</b> <b>P. O. Box 1201</b> <b>Reading, Pa. 19603</b>		U.S. CERTIFICATED MECHANIC
		FOREIGN CERTIFICATED MECHANIC
		CERTIFICATED REPAIR STATION
		MANUFACTURER
		C. CERTIFICATE NO.
		<b>3535</b>
		<b>Airframe 1 &amp; 3</b>
		<b>Radio 1, 2, &amp; 3</b>
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.		
DATE <b>7 September 1976</b>		SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Charles E. Combs</i> <b>Charles E. Combs Rep. 362324145</b>
7. APPROVAL FOR RETURN TO SERVICE		
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED		
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER
	FAA DESIGNEE <b>X</b>	REPAIR STATION
		INSPECTION AUTHORIZATION
		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT
DATE OF APPROVAL OR REJECTION <b>7 September 1976</b>		CERTIFICATE OR DESIGNATION NO. <b>CRS 3535</b> SIGNATURE OF AUTHORIZED INDIVIDUAL <i>James H. Franklin</i> <b>James H. Franklin - Inspector</b>

# **NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

ITEM	WEIGHT	ARM	LOCATION
Installed:			
King KT-76 Transponder & Rack	3.5	11.0	Center Instr. Panel
Trans-Cal D120-P2-T			
Encoding Altimeter	1.2	13.0	Center Instr. Panel
King Transponder Installation Kit	.5	12.0	Fed. Fuselage

Blind encoder installed in accordance with Trans-Cal drawing E-100124 installation instructions and AC43:13-2.

Wiring connected to King KT-76 per installation manual 006-0067-01 drawing 696-3003-00.

Ground checked in accordance with Table 2 of AC43-6.

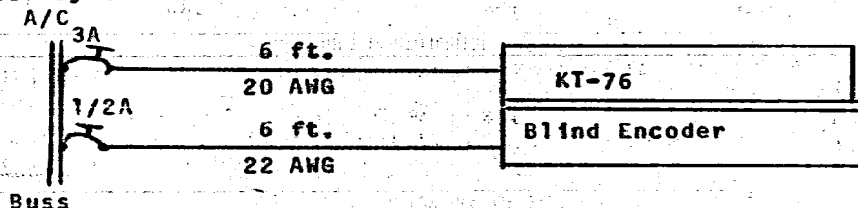
Equipment list revised.

All pertinent data referenced herein is on file at CRS 3535.

Work has been accomplished in accordance with AC43:13-1 Chapter 11 AC43:13-2 Chapters 1 & 2.

Continuous electrical load does not exceed 80% of the total output of 100 amperes produced by two 50 ampere generators.

See weight and balance section of Aircraft Flight Manual for changes reflected by this alteration.



This installation is approved based on a similar installation of Trans-Cal D120-P2-T Blind Encoder and King KT-76 approved by GADO #3 on FAA 337 dated 25 March 1975 for Piper PA 32-300 N1287T.

END

☐ ADDITIONAL SHEETS ARE ATTACHED

TEST SHEET  
BLIND ENCODING ALTIMETER CHECK  
AC 43-6, PAR 7, TABLE 2

TEST ALTITUDE	D2	D4	A1	A2	A4	B1	B2	B4	C1	C2	C4	UP READING	OBSERVED ERROR	DOWN READING	OBSERVED ERROR
-1,000 FT.	0	0	0	0	0	0	0	0	0	1	0	-1070	-20	-1000	50
0 FT.	0	0	0	0	0	0	1	1	0	T	0	70	-20	-10	60
500 FT.	0	0	0	0	0	0	T	0	0	T	0	550	0	530	20
1,000 FT.	0	0	0	0	0	1	T	0	0	T	0	950	0	990	60
1,100 FT.	0	0	0	0	0	T	T	0	1	T	0	1010	-40	1100	50
1,300 FT.	0	0	0	0	0	T	T	1	T	0	0	1220	-30	1310	40
1,500 FT.	0	0	0	0	0	T	T	T	0	1	0	1420	-30	1520	30
1,800 FT.	0	0	0	0	0	T	0	T	0	0	1	1850	0	1810	40
2,000 FT.	0	0	0	0	0	T	0	T	0	1	0	1950	0	1990	60
2,600 FT.	0	0	0	0	0	T	0	0	0	T	1	2550	0	2620	30
2,700 FT.	0	0	0	0	0	T	0	0	0	0	T	2650	0	2720	30
3,000 FT.	0	0	0	0	1	T	0	0	0	1	0	2910	-40	3000	50
4,000 FT.	0	0	0	0	T	T	1	1	0	T	0	3930	-20	3930	70
6,000 FT.	0	0	0	1	T	0	0	0	0	0	1	5920	-30	5990	50
6,800 FT.	0	0	0	T	T	0	1	1	0	1	0	6730	-20	6800	50
8,000 FT.	0	0	0	T	T	0	1	1	0	1	0	7900	-50	7970	30
10,000 FT.	0	0	0	T	T	1	0	T	0	T	0	9890	-60	9950	100
12,000 FT.	0	0	0	T	0	T	1	T	0	T	0	11860	-90	11930	120
14,000 FT.	0	0	0	T	0	0	0	T	0	T	0	13880	-70	13930	120
14,800 FT.	0	0	1	T	0	0	0	0	0	0	1	14700	-50	14780	80
16,000 FT.	0	0	T	T	0	0	1	1	0	1	0	14890	-60	15970	80
18,000 FT.	0	0	T	T	0	1	0	T	0	T	0	17890	-60	17930	120
20,000 FT.	0	0	T	T	1	T	1	T	0	T	0	19870	-80	19970	80
22,000 FT.	0	0	T	T	T	0	0	T	0	T	0	-	-	-	-
25,000 FT.	0	0	T	0	T	1	1	0	0	T	0	-	-	-	-
30,000 FT.	0	0	T	0	0	0	0	1	0	T	0	-	-	-	-
30,800 FT.	0	1	1	0	0	0	0	0	0	0	1	-	-	-	-
35,000 FT.	0	T	T	0	1	0	0	0	0	1	0	-	-	-	-
40,000 FT.	0	T	T	1	0	1	1	1	0	T	0	-	-	-	-
45,000 FT.	0	T	T	T	0	0	T	0	0	T	0	-	-	-	-
50,000 FT.	0	T	0	T	0	1	0	1	0	T	0	-	-	-	-

ENCODING ALTIMETER TYPE AND MODEL Trans-Cal D-120-P2-T

ALTIMETER S/N 26390

MODEL TRANSPONDER KT-76

INTERFACE YES NO XX MODEL        (if Applicable)

A/C MAKE AND MODEL Cessna 310 A/C # H310CM

CHECK PERFORMED BY J.M. Hechler CRS # 3535

DATE PERFORMED 11 November 1976

TEST EQUIPMENT USED Michels 730

FFRFAIC#600

CLASS OF TRANSPONDER Class I

CAMERA NO. 4/11 DATE: 12-29-83

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				11th Approved Sage Bureau No. 04-R060.1	
<b>MAJOR REPAIR AND ALTERATION</b> (Airframe, Powerplant, Propeller, or Appliance)				DEC 1972	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.				OFFICE IDENTIFICATION PNE 1209	
1. AIRCRAFT	MAKE CESSNA	MODEL 310P	NATIONALITY AND REGISTRATION MARK N310CM		
	SERIAL NO. 310P-0178				
2. OWNER	NAME (As shown on registration certificate) CLYMER MACHINE CO., INC.		ADDRESS (As shown on registration certificate) FINLAND ROAD TRUMBAYERSVILLE, PA. 18970		
	3. FOR FAA USE ONLY				
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
POWERPLANT					XX
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS GEORGE HAMMERSCHMIDT FOR J.W.S. ELECTRONICS, INC. ROBERT J. MILLER AIRPARK TOMS RIVER, N. J. 08753			B. KIND OF AGENCY		C. CERTIFICATE NO. A&E1205476
			<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC		
			<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC		
			<input type="checkbox"/> CERTIFICATED REPAIR STATION		
			<input type="checkbox"/> MANUFACTURER		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse of attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE DECEMBER 4, 1972			SIGNATURE OF AUTHORIZED INDIVIDUAL <i>George Hammerschmidt</i>		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/> INSPECTION AUTHORIZATION CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	OTHER (Specify)	
	FAA DESIGNEE	REPAIR STATION			
DATE OF APPROVAL OR REJECTION 12-4-72		CERTIFICATE OR DESIGNATION NO. IA1205476	SIGNATURE OF AUTHORIZED INDIVIDUAL <i>George Hammerschmidt</i>		

### NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Removed GAR Radar Altimeter and related antenna.

Installed Bonzer TRN71 Radar Altimeter and related antenna and indicator in accordance with Bonzer Installation Manual and AC43-13-2 Chapters 1, 2 and 3.

Wired in accordance with AC43-13-1, Chapter 11.

New Weight and Balance computation entered in aircraft log and equipment list revised.

Aircraft was test flown for operational aspects as set forth in FAR 91.167 and FAR 135.143b. All components were operating properly.

ADDITIONAL SHEETS ARE ATTACHED

FEDERAL AVIATION AGENCY				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION 21-02	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE <b>Cessna</b>		MODEL <b>310P</b>		
	SERIAL NO. <b>310P - 0178</b>		NATIONALITY AND REGISTRATION MARK <b>310CM</b>		
2. OWNER	NAME (As shown on registration certificate) <b>Riley Turbo Sales Corporation</b>		ADDRESS (As shown on registration certificate) <b>2900 Preston Texas 75295</b>		
	3. FOR FAA USE ONLY				
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
POWERPLANT				REPAIR	ALTERATION
PROPELLER				REPAIR	ALTERATION
APPLIANCE	TYPE			REPAIR	ALTERATION
	MANUFACTURER			REPAIR	ALTERATION
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS			B. KIND OF AGENCY		C. CERTIFICATE NO.
<b>Robert P. Steinbach 1187 Belew Irving Texas</b>			<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC		<b>AP 1735827</b>
			<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC		
			<input type="checkbox"/> CERTIFICATED REPAIR STATION		
			<input type="checkbox"/> MANUFACTURER		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE <b>March 21, 1970</b>			SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Robert P. Steinbach</i>		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/>	INSPECTION AUTHORIZATION	
	FAA DESIGNEE	REPAIR STATION	<input type="checkbox"/>	OTHER (Specify)	
DATE OF APPROVAL OR REJECTION <b>March 21, 1970</b>		CERTIFICATE OR DESIGNATION NO. <b>AI 1735827</b>		SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Robert P. Steinbach</i>	

# NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed Alcor Ekonomix EGT system In Accordance With Alcor  
Report 65-112 Model 36 S/N 414439 STC SA522SW also revised  
equipment list No change in weight an balance

END

☐ ADDITIONAL SHEETS ARE ATTACHED

FEDERAL AVIATION AGENCY MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				<small>Item Approved Budget Bureau No. 04-R060.1</small> <b>FOR FAA USE ONLY</b> OFFICE IDENTIFICATION <small>Fin-6000</small> <span style="border: 1px solid black; padding: 2px;">260</span>	
<small>INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.</small>					
1. AIRCRAFT	MAKE <b>Cessna</b>	MODEL <b>310P</b>			
	SERIAL NO. <b>310P-0178</b>	NATIONALITY AND REGISTRATION MARK <b>N 310 CM</b>			
2. OWNER	NAME (As shown on registration certificate) <b>Riley Turbo Sales Corporation</b>		ADDRESS (As shown on registration certificate) <b>2580 Preston Tower Dallas Texas 75225</b>		
	3. FOR FAA USE ONLY				
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
				REPAIR	ALTERATION
AIRFRAME	***** (As described in item 1 above) *****				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS <b>G. F. Adams Turbo propulsion Corp. P.O. Box 2286 Wichita Falls Texas 76307</b>			<input checked="" type="checkbox"/> B. KIND OF AGENCY <input type="checkbox"/> U.S. CERTIFICATED MECHANIC <input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC <input type="checkbox"/> CERTIFICATED REPAIR STATION <input type="checkbox"/> MANUFACTURER		C. CERTIFICATE NO. <b>AT 29789-40</b>
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE <b>March 20, 1970</b>			SIGNATURE OF AUTHORIZED INDIVIDUAL <b>G. F. Adams</b> <i>G. F. Adams</i>		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, this unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/>	INSPECTION AUTHORIZATION OTHER (Specify)	
	FAA DESIGNEE	REPAIR STATION	<input type="checkbox"/>		
DATE OF APPROVAL OR REJECTION <b>3/20/70</b>			SIGNATURE OF AUTHORIZED INDIVIDUAL <b>G. F. Adams</b> <i>G. F. Adams</i>		

## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

**B. DESCRIPTION OF WORK ACCOMPLISHED** (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

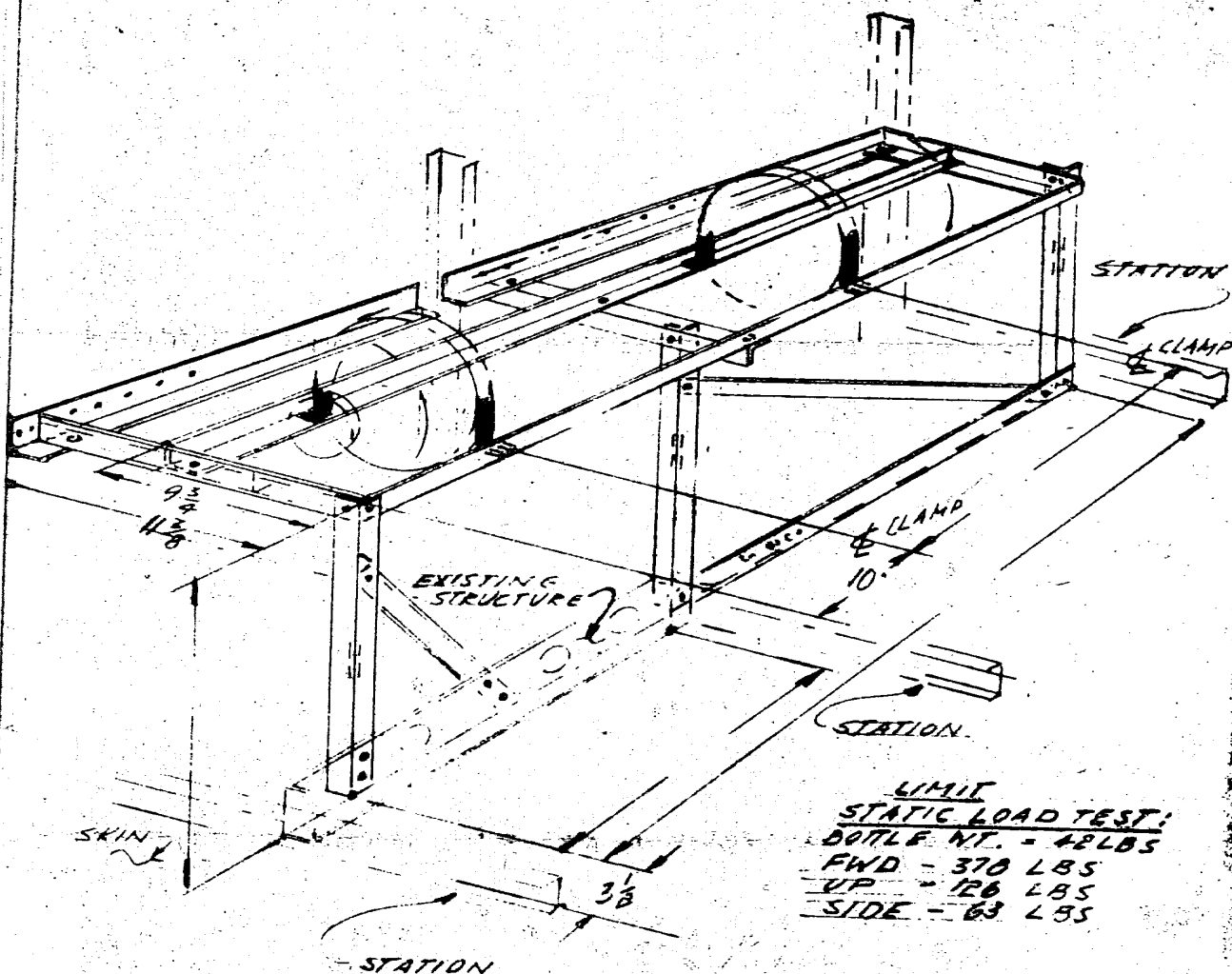
Install oxygen system. Sky Ox ICC- 3AA 114.9 cu. ft. 1800 psi bottle installed in rack behind baggage compartment. Rack installed between station 182.00 to 177.50. See attached sketch. Original Sky Ox valve used. Bottle and valve tested April 1967. Zep regulator A2000-WA mounted front inboard side of bottle rack. 3/16 " hi pressure copper tubing with silver soldered end fittings from bottle to gauge to filler valve to regulator. Zep ZV681C Filler valve. Scott p/n 8808-1 Pressure gauge red lined at 1750 psi. Gauge mounted in forward face of bulkhead 132.00 to be visible to pilot. Filler valve and shut off valve accessible from cabin through door in bulkhead 132.00. 5/16 " seamless aluminum tubing with AN fittings used from regulator to Zep outlets. 2 Zep Z651C outlets installed in factory mount aft of top instrument light, For pilot and copilot use. 2 Zep Z651 passenger outlets mounted in factory holes in individual reading light and air vent bracket above windows. All parts used listed in 310P parts book. All lines routed under headlining in top of aircraft properly secured and routed away from oil, fuel, electrical and moving control systems. System was leaked checked and no leaks found. System was functionally tested and all outlets performed properly. Installation in accordance with Chapter 6 paragraph 71 through 76.

Installed two whelen model SA strobe lights in aft fuselage. One on top of fuselage at + 136. Installed with .040 doubler T3 aluminum, similar to figure 4-2 of chapter 4 of 43-13-2. One installed at + 160 on bottom of fuselage. .040 T3 doubler similar to figure 4.1 of chapter 4 of 43-13-2. Wiring was MIL-5086 shielded # 16 wire from a 5 amp circuit breaker to switch to light. Switch properly marked. Lights functionally tested and effects no other equipment or vision. Work accomplished in accordance with chapter 4 of 43-13-2.

Install Goodrich propeller de-ice kit for Cessna 310P with Lycoming engines and Hartzell propellers. Kit # 77-360 and parts list dated 2/17/70. Goodrich system 77-080 and report # 60-726A and maintenance publication # 63-200. All parts furnished by Goodrich and installed in accordance with Goodrich drawings and instructions. Timer 3E1150-3 mounted on left lower front side of station 0. System functionally tested satisfactory. Electrical load with all systems turned on does not exceed 80% of the alternator capacity. All wiring harnesses, switches and circuit breakers furnished by Goodrich. Aircraft was reweighed after all equipment was installed. Weight and balance data revised and equipment list revised. All work accomplished in accordance with 43-13-1 and 2.

The End

ADDITIONAL SHEETS ARE ATTACHED



OXYGEN BOTTLE RACK IN CESSNA 310 SERIES

MATERIAL: ALL ANGLES - AND 10133-1002: BOLTED:  
CONNECTIONS USING AN 3-4A BOLTS, AN 365-1032 NUTS,  
AN 960-3 WASHERS. BOTTOM & SIDE ANGLES  
RIVETED TO SKIN WITH AN 470 AD4 RIVETS, 1/2 SPRING.  
OXYGEN BOTTLE: SKY OX ICC 3AA 114.9 CU. FT.  
& 1800 PSI. BOTTLE CLAMPS: ZEP AERO G1000-1, CREED.

RILEY AERONAUTICS  
RKS 650827-1

FOR RECORD  
CAMERA NO. 4/11 DATE: 12-29-83

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
<b>MAJOR REPAIR AND ALTERATION</b> (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
OFFICE IDENTIFICATION FTW-CADD 2-1-04					
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE <b>Cessna</b>		MODEL <b>310P</b>		
	SERIAL NO. <b>310P-0178</b>		NATIONALITY AND REGISTRATION MARK <b>N 310CM</b>		
2. OWNER	NAME (As shown on registration certificate) <b>Riley Turbo Sales Corporation</b>		ADDRESS (As shown on registration certificate) <b>2900 Preston Tower Dallas Texas 75225</b>		
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	***** (As described in item 1 above) *****				I
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
5. TYPE					
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS			B. KIND OF AGENCY		C. CERTIFICATE NO.
<b>G. F. Adams Turbo Propulsion Corp. Wichita Falls, Texas 76307 P.O. Box 2286</b>			<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC		<b>AI 29789-40</b>
			<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC		
			<input type="checkbox"/> CERTIFICATED REPAIR STATION		
			<input type="checkbox"/> MANUFACTURER		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE <b>March 19, 1970</b>		SIGNATURE OF AUTHORIZED INDIVIDUAL <b>G. F. Adams</b>			
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION		OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION <b>3/20/70</b>		CERTIFICATE OR DESIGNATION NO. <b>AI 29789-40</b>		SIGNATURE OF AUTHORIZED INDIVIDUAL <b>G. F. Adams</b>	

# **NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Modifications carried out in accordance with Turbo Propulsion Corporation  
STC SA 2082 WB and drawing summary "H" and "J", of TP31000 drawing list  
Install two new Lycoming TIO-540-A2B engines and two new Hartzell HC-E3YR-2A /  
C8468-6R propellers and Piqua F-6-11A governors. Installed in conformity  
with the above STC and drawing list.

THE END

☐ ADDITIONAL SHEETS ARE ATTACHED

U.S. GOVERNMENT PRINTING OFFICE: 1967 OF-272-085

UNITED STATES OF AMERICA — FEDERAL AVIATION AGENCY <small>FTW EMD0</small>			
STANDARD AIRWORTHINESS CERTIFICATE 2-0-42			
1. NATIONALITY AND REGISTRATION MARKS	2. MANUFACTURER AND MODEL	3. AIRCRAFT SERIAL NUMBER	4. CATEGORY
N310CM	Cessna 310P	210P-0178	Normal
5. AUTHORITY AND BASIS FOR ISSUANCE This airworthiness certificate is issued pursuant to the Federal Aviation Act of 1958 and certifies that, as of the date of issuance, the aircraft to which issued has been inspected and found to conform to the type certificate therefor, to be in condition for safe operation, and has been shown to meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex 8 to the Convention on International Civil Aviation, except as noted herein. Exceptions: <b>None</b>			
6. TERMS AND CONDITIONS Unless sooner surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator, this airworthiness certificate is effective as long as the maintenance, preventive maintenance, and alterations are performed in accordance with Parts 23, 25, and 91 of the Federal Aviation Regulations, as appropriate, and the aircraft is registered in the United States.			
DATE OF ISSUANCE <b>3/20/77</b> (Amended)	FAA REPRESENTATIVE <b>Clarence E. Knight</b>	CERTIFICATE NUMBER <b>2-0-42</b>	
Any alteration, reproduction, or misuse of this certificate may be punishable by a fine not exceeding \$1,000, or imprisonment not exceeding 3 years, or both. THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.			
FAA Form 1362 (7-65)		* GPO : 1965 O - 798-353 0052-040-8000 (8100)	

CAMERA NO. 4N DATE: 12-29-83

MICRO

UNITED STATES OF AMERICA FEDERAL AVIATION AGENCY		
CERTIFICATE OF AIRWORTHINESS		
1. NATIONALITY AND REGISTRATION MARKS N310CM	2. AIRCRAFT AIRWORTHINESS CLASSIFICATION EXPERIMENTAL-TO SHOW COMPLIANCE WITH FAR (SEE REVERSE SIDE)	
3. This Certificate of Airworthiness is issued pursuant to the Federal Aviation Act of 1958. The aircraft identified hereon is considered airworthy when maintained and operated in accordance with the FAR Regulations and applicable aircraft Operation Limitations.		
4. This Certificate will remain in effect in full force and effect as long as the aircraft is maintained in accordance with the FAR Regulations and applicable aircraft Operation Limitations. <del>It is the responsibility of the owner to maintain the aircraft in accordance with the FAR Regulations and applicable aircraft Operation Limitations.</del> Expires Jan. 10, 1970		
5. DATE OF ISSUANCE 10/9/69	6. SIGNATURE OF REPRESENTATIVE <i>Charles E. Harris</i> Charles E. Harris	7. DESIGNATION SW-EMDO-42
8. Any alteration or misuse of this Certificate is punishable by a fine of not exceeding \$1,000 or imprisonment not exceeding 3 years, or both.		

1. Operating Limitations Dated 10/9/69 are a part of this Certificate.
2. This aircraft does not comply with the International Airworthiness Standards of ICAO, Annex 8, Part 11. Special permission must be obtained for flights over foreign countries.
3. Flights for purposes other than to show compliance with FAR prohibited.

END.

APR 2 1970

FTW EMDO  
2-0-46

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION FTW-GADO 2-1-04	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE	Cessna		MODEL	310P
	SERIAL NO.	310P-0178		NATIONALITY AND REGISTRATION MARK	N 310 CN
2. OWNER	NAME (As shown on registration certificate)			ADDRESS (As shown on registration certificate)	
	Hiley Turbo Sales Corporation			2800 Preston Tower Dallas Texas 75225	
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
POWERPLANT					X
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY NAME AND ADDRESS			B. KIND OF AGENCY		
G. F. Adams Turbo Propulsion Corp. P.O. Box 2286 Wichita Falls Texas 76307			AI 28789-40		
			C. CERTIFICATE NO.		
			X U.S. CERTIFICATED MECHANIC		
			FOREIGN CERTIFICATED MECHANIC		
			CERTIFICATED REPAIR STATION		
			MANUFACTURER		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE: March 20, 1970			SIGNATURE OF AUTHORIZED INDIVIDUAL		
			H. F. Adams		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION	OTHER (Specify)	
	FAA DESIGNEE	REPAIR STATION			
			CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE: 3/20/78		CERTIFICATE NO. AI 28789-40		SIGNATURE OF AUTHORIZED INDIVIDUAL	
REJECTION				H. F. Adams	

# NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

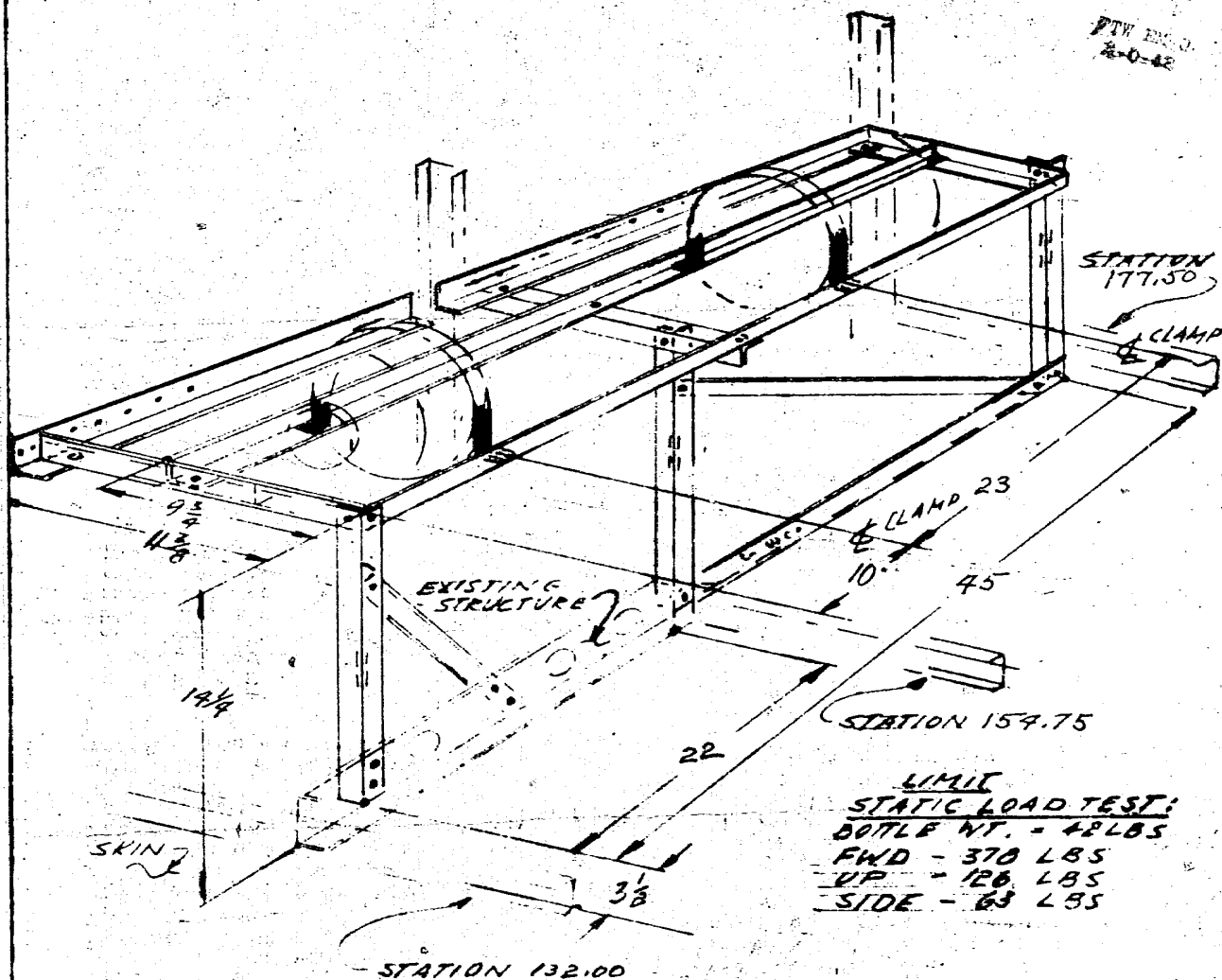
Install oxygen system. Sky Ox ICC- 3AA 114.9 cu. ft. 1800 psi bottle installed in rack behind baggage compartment. Rack installed between station 132.00 to 177.50. See attached sketch. Original Sky Ox valve used. Bottle and valve tested April 1967. Zep regulator A2000-WA mounted front inboard side of bottle rack. 3/16 " h1 pressure copper tubing with silver soldered end fittings from bottle to gauge to filler valve to regulator. Zep ZV681C Filler valve. Scott p/n 8808-1 Pressure gauge red lined at 1750 psi. Gauge mounted in forward face of bulkhead 132.00 to be visible to pilot. Filler valve and shut off valve accessible from cabin through door in bulkhead 132.00. 5/16 " seamless aluminum tubing with AN fittings used from regulator to Zep outlets. 2 Zep Z651C outlets installed in factory mount aft of top instrument light. For pilot and copilot use. 2 Zep Z651 passenger outlets mounted in factory holes in individual reading light and air vent bracket above windows. All parts used listed in 310P parts book. All lines routed under headlining in top of aircraft properly secured and routed away from oil, fuel, electrical and moving control systems. System was leak checked and no leaks found. System was functionally tested and all outlets performed properly. Installation in accordance with Chapter 6 paragraph 71 through 76.

Installed two whelen model SA strobe lights in aft fuselage. One on top of fuselage at + 136. Installed with .040 doubler T3 aluminum, similar to figure 4-2 of chapter 4 of 43-13-2. One installed at + 160 on bottom of fuselage. .040 T3 doubler similar to figure 4.1 of chapter 4 of 43-13-2. Wiring was MIL-5086 shielded # 16 wire from a 5 amp circuit breaker to switch to light. Switch properly marked. Lights functionally tested and effects no other equipment or vision. Work accomplished in accordance with chapter 4 of 43-13-2.

Install Goodrich propeller de-ice kit for Cessna 310P with Lycoming engines and Hartzell propellers. Kit # 77-360 and parts list dated 2/17/70. Goodrich system 77-080 and report # 60-726A and maintenance publication # 63-200. All parts furnished by Goodrich and installed in accordance with Goodrich drawings and instructions. Timer 3E1150-3 mounted on left lower front side of station 0. System functionally tested satisfactory. Electrical load with all systems turned on does not exceed 80% of the alternator capacity. All wiring harnesses switches and circuit breakers furnished by Goodrich. Aircraft was reweighed after all equipment was installed. Weight and balance data revised and equipment list revised. All work accomplished in accordance with 43-13-1 and 2.

The End

☐ ADDITIONAL SHEETS ARE ATTACHED



OXYGEN BOTTLE RACK IN CESSNA 310 SERIES

MATERIAL: ALL ANGLES - AND 10133-1002. BOLTED CONNECTIONS USING AN 3-4A BOLTS, AN 365-1032 NUTS, AN 960-3 WASHERS. BOTTOM & SIDE ANGLES RIVETED TO SKIN WITH AN 470-AD4 RIVETS, 1/2 SPACING. OXYGEN BOTTLE: SKY OX ICC 3AA 114.9 CU. FT. @ 1800 PSI. BOTTLE CLAMPS: ZEP AERO 61000-1, CREED.

RILEY AERONAUTICS  
RKS 65087-1

FAA AIRCRAFT  
CAMERA NO. 411 DATE: 12-29-83

PERCUTED

DEC 7 12 49 PM '80  
FEDERAL BUREAU OF INVESTIGATION  
MANUFACTURING SECTION

APR 25 1970

FTW EMDG

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.				OFFICE IDENTIFICATION FTW-GA00 2-1-04	
1. AIRCRAFT	MAKE	Cessna	MODEL	310P	
	SERIAL NO.	310P-0178	NATIONAL AND REGISTRATION MARK	3108m	
2. OWNER	NAME (As shown on registration certificate)		ADDRESS (As shown on registration certificate)		
	Riley Turbo Sales Corporation		Dallas Texas 75225		
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS			B. KIND OF AGENCY		
Turbo Propulsion Corp. Wichita Falls, Texas 76307 P.O. Box 2286			<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC <input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC <input type="checkbox"/> CERTIFICATED REPAIR STATION <input type="checkbox"/> MANUFACTURER		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE March 19, 1970			SIGNATURE OF AUTHORIZED INDIVIDUAL O. F. Adams		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION	OTHER (Specify)	
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION March 20, 1970		CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL O. F. Adams		

# NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Modifications carried out in accordance with Turbo Propulsion Corporation  
STC SA 2682 WB and drawing summary "H" and "J", of TP31600 drawing list  
Install two new Lycoming TIO-540-A2B engines and two new Hartzell HC-E3YR-2A /  
C8468-6R propellers and Piqua P-6-11A governors. Installed in conformity  
with the above STC and drawing list.

THE END

PROPOSED

LOADING/UNLOADING

VALIDITY

END

NAME

ADDRESS

DATE

TIME

PLACE

REMARKS

☐ ADDITIONAL SHEETS ARE ATTACHED

W.O. # 22051

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
OFFICE IDENTIFICATION					
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE	Cessna		MODEL	310-P
	SERIAL NO.	310P-0178		NATIONALITY AND REGISTRATION MARK	N 310 CM
2. OWNER	NAME (As shown on registration certificate)			ADDRESS (As shown on registration certificate)	
	Clymer Machine Co., Inc.			Finland Road Trumbauersville, Penna. 18970	
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1. above)*****			REPAIR	ALTERATION
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS			B. KIND OF AGENCY		C. CERTIFICATE NO.
ElectroSonics Division 4885 Sawyer Road Columbus International Airport Columbus, Ohio 43219			U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC CERTIFICATED REPAIR STATION MANUFACTURER		Repair Station 1009 Radio Class 1, 2, and 3 Limited Instruments Limited
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE April 16, 1971			SIGNATURE OF AUTHORIZED INDIVIDUAL <i>William H. Hensinger</i>		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION	OTHER (Specify)	
	FAA DESIGNER	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION 4-16-71		CERTIFICATE OR DESIGNATION NO. 1009	SIGNATURE OF AUTHORIZED INDIVIDUAL <i>William H. Hensinger</i>		

# NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. Removed Mitchell directional gyro and VOA-50M. Installed KPI-550, KN-71, SA-10 in accordance with AC 43.13-2 Chapter 2.
2. Installed KMT-110 flux detector in rear fuselage in accordance with AC 43.13-2 Chapter 3.
3. All circuit breakers, switches, wiring and method of installation are in accordance with AC 43.13-1 Chapter 11 Sections 2 & 3.
4. Electrical load checked in accordance with AC 43.13-2 Chapter 2 Paragraph 27 and found to be within specified limits.
5. Moment 134042.7 = New Empty Weight CG 34.8 = New Useful Load 1386.8  
Weight 3813.2

END

EXAMINER			
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FEDERAL AVIATION AGENCY APPLICATION FOR AIRWORTHINESS CERTIFICATE (FAR 21)				INSTRUCTIONS: Print or type. Do not write in shaded areas, these are for FAA use only. Submit original only to an authorized FAA Representative. If additional space is required, use reverse side.	
I. AIRCRAFT DESCRIPTION	1. REGISTRATION MARK N310CM	2. AIRCRAFT MAKE Cessna	3. AIRCRAFT MODEL DESIGNATION 310P	FAA CODING 2074238	
	4. AIRCRAFT SERIAL NO. 310P-0178	5. ENGINE MAKE Lycoming	6. ENGINE MODEL DESIGNATION TIO-540-A2B	41533	
	7. NO. OF ENGINES 2	8. PROPELLER MAKE Hartzell	9. PROPELLER MODEL DESIGNATION HC-E3YR-2A	10. YR. ARC. MANUFACTURED 1969	
II. CERTIFICATION REQUESTED	A. APPLICATION IS HEREBY MADE FOR:				
	<input checked="" type="checkbox"/> ORIGINAL CERTIFICATE—AIRCRAFT IS NEW <input type="checkbox"/> AMENDMENT OR MODIFICATION OF CURRENT CERTIFICATE USED <input checked="" type="checkbox"/> IMPORT <input type="checkbox"/> OTHER (Specify)				
	B. AIRWORTHINESS CLASSIFICATION (For multiple certification, check appropriate items)				
	<input type="checkbox"/> STANDARD (Indicate category) <input type="checkbox"/> NORMAL <input type="checkbox"/> UTILITY <input type="checkbox"/> ACROBATIC <input type="checkbox"/> TRANSPORT <input type="checkbox"/> LIMITED				
	<input type="checkbox"/> RESTRICTED (Indicate operation(s) to be conducted) <input type="checkbox"/> AGRICULTURE & PEST CONTROL <input type="checkbox"/> AERIAL SURVEYING <input type="checkbox"/> AERIAL ADVERTISING <input type="checkbox"/> FOREST (Wild life conservation) <input type="checkbox"/> PATROLLING <input type="checkbox"/> WEATHER CONTROL <input type="checkbox"/> OTHER (Specify)				
	<input checked="" type="checkbox"/> EXPERIMENTAL (Indicate operation(s) to be conducted) <input checked="" type="checkbox"/> RESEARCH AND DEVELOPMENT <input type="checkbox"/> AMATEUR BUILT <input type="checkbox"/> EXHIBITION <input type="checkbox"/> RACING <input checked="" type="checkbox"/> TO SHOW COMPLIANCE WITH FAR <input type="checkbox"/> PROVISIONAL (Indicate class) <input type="checkbox"/> CLASS I <input type="checkbox"/> CLASS II				
	FAA CODING				
III. OWNER'S CERTIFICATION	A. REGISTERED OWNER (As shown on Certificate of Aircraft Registration)				
	NAME Biley Turbo Sales Corp.		ADDRESS (Number, street, city, State and ZIP code) 2900 Preston Towers Dallas Texas 75225		
	B. AIRCRAFT CERTIFICATION BASIS				
	<input checked="" type="checkbox"/> AIRCRAFT SPECIFICATION OR TYPE CERTIFICATION DATA SHEET (Give No. and Rev. No.) 3A10-33		AIRWORTHINESS DIRECTIVE SUMMARY (Give year and last card No.) 69-20		
	<input type="checkbox"/> AIRCRAFT LISTING (Give page No(s.)) None		SUPPLEMENTAL TYPE CERTIFICATE (Give No(s.)) None		
IV. INSPECTION AGENCY VERIFICATION	C. AIRCRAFT OPERATION AND MAINTENANCE RECORDS				
	<input checked="" type="checkbox"/> AIRCRAFT NEW—NO PREVIOUS HISTORY <input checked="" type="checkbox"/> RECORDS MAINTAINED IN COMPLIANCE WITH FAR 91.173		TOTAL AIRFRAME HOURS RECORDED 1:35		FAA CODING
	I hereby certify that I am the registered owner (or his agent) of the aircraft described above; that the aircraft is registered with the Federal Aviation Agency in accordance with section 501 of the Federal Aviation Act of 1958, and applicable Federal Aviation Regulations; and that the aircraft has been inspected and is airworthy and eligible for the classification(s) requested.				
	DATE OF APPLICATION 10-9-69		NAME AND TITLE (Print or type) G. F. Adams Inspector		SIGNATURE <i>G. F. Adams</i>
	IN ACCORDANCE WITH FAR 21.183, THE AIRCRAFT DESCRIBED ABOVE HAS BEEN INSPECTED AND FOUND AIRWORTHY BY:				
V. AIRWORTHINESS DOCUMENTATION	<input checked="" type="checkbox"/> CERTIFICATED AIR CARRIER (Give Certificate No.) <input checked="" type="checkbox"/> AIRCRAFT MANUFACTURER (Give name of firm)		<input checked="" type="checkbox"/> CERTIFICATED MECHANIC (IA) (Give Certificate No.) <input type="checkbox"/> CERTIFICATED DOMESTIC REPAIR STATION (Give Certificate No.)		FAA CODING
	DATE		TITLE		SIGNATURE
	<input checked="" type="checkbox"/> A. CURRENT OPERATING LIMITATIONS AVAILABLE IN AIRCRAFT <input checked="" type="checkbox"/> B. CURRENT OPERATING LIMITATIONS ATTACHED <input type="checkbox"/> C. CURRENT APPROVED FLIGHT MANUAL AVAILABLE IN AIRCRAFT <input type="checkbox"/> D. DATA, DRAWINGS, PHOTOGRAPHS, ETC. (Attach when required) <input checked="" type="checkbox"/> E. CURRENT WEIGHT & BALANCE INFORMATION AVAILABLE IN AIRCRAFT <input type="checkbox"/> F. MAJOR REPAIR AND ALTERATION, FAA FORM 337 (Attach when req.)				
	<input checked="" type="checkbox"/> G. THIS INSPECTION RECORDED IN AIRCRAFT RECORDS <input type="checkbox"/> H. STATEMENT OF CONFORMITY, FAA FORM 317 (Attach when required) <input checked="" type="checkbox"/> I. PREVIOUS AIRWORTHINESS CERTIFICATE ISSUED IN ACCORDANCE WITH FAR 21.183A CAR (Orig. attached) <input checked="" type="checkbox"/> J. AIRWORTHINESS CERTIFICATE ISSUED IN ACCORDANCE WITH FAR 21.191 (Copy attached)				
	<input checked="" type="checkbox"/> A. I have inspected the aircraft described above, find it conforms to its type certificate, and is in condition for safe operation. <input type="checkbox"/> B. I have inspected the aircraft described above, find it is in condition for safe experimental operation. <input type="checkbox"/> C. Current Airworthiness Certificate amended.				
VI. FAA REPRESENTATIVE CERTIFICATION	DATE 10/9/69		DESIGNATION NO. Mfg. Insp.		OFFICE NO. 2-42 SW-EMDO-42
					SIGNATURE <i>Charles E. Harris</i> Charles E. Harris

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RECEIVED

Oct 15 09 53 269  
ENGINEERING AND  
MANUFACTURING BRANCH

Punched by Operator #41

DEPARTMENT OF TRANSPORTATION  
FEDERAL AVIATION ADMINISTRATION  
SOUTHWEST REGION  
FORT WORTH, TEXASTTF EDOO  
2-0-42EXPERIMENTAL OPERATING LIMITATIONSIssue Date Oct. 9, 1969 Expiration Date Jan. 10, 1970MAKE Cessna MODEL 310P SERIAL NO. 310P-0178 REG. NO. N310CM

1. This aircraft is certificated in the Experimental classification for the purpose of: To Show Compliance with F.A.R.
2. The aircraft may not be operated for other than the purpose for which the certificate was issued.
3. The aircraft is to be identified "experimental" in accordance with FAR 45.23.
4. Persons or property shall not be carried for compensation or hire.
5. Occupancy of the aircraft is restricted to persons essential to the flight test involved.
6. All flights shall be conducted under Visual Flight Rules (VFR).  
Day or Night
7. All flights shall be conducted in accordance with the applicable FAA air traffic and general operation rules of FAR Part 91.
8. The control tower shall be notified of the experimental nature of the aircraft when operating into or out of airports with operating control towers.
9. All flights shall be conducted within the following boundaries and/or corridors: N/A
10. This aircraft does not comply with the International Airworthiness Standards of ICAO, Annex 8, Part II.

Charles E. Harris  
FAA RepresentativeSW-EMDO-42  
Designation

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Punched by Operator #41

Verified by Operator #4

UNITED STATES OF AMERICA — FEDERAL AVIATION AGENCY FTW EMDO  
STANDARD AIRWORTHINESS CERTIFICATE 2-0-42

1. NATIONALITY AND REGISTRATION MARKS <b>N310CM</b>	2. MANUFACTURER AND MODEL <b>Cessna 310P</b>	3. AIRCRAFT SERIAL NUMBER <b>310P0178</b>	4. CATEGORY <b>Normal</b>
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5. AUTHORITY AND BASIS FOR ISSUANCE  
This airworthiness certificate is issued pursuant to the Federal Aviation Act of 1958 and certifies that, as of the date of issuance, the aircraft to which issued has been inspected and found to conform to the type certificate therefor, to be in condition for safe operation, and has been shown to meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex 8 to the Convention on International Civil Aviation, except as noted herein.

Exceptions:

NONE

**SUPERSEDED**

6. TERMS AND CONDITIONS  
Unless sooner surrendered, suspended, revoked, or annulled, this certificate is effective as long as the requirements, provisions, maintenance, and alterations are performed in accordance with Parts 21, 43, and 91 of the Federal Aviation Regulations, as appropriate, and the aircraft is registered in the United States.

DATE OF ISSUANCE <b>7-11-69</b>	FAA REPRESENTATIVE <i>Raymond M. Rowden</i> <b>Raymond M. Rowden</b>	DESIGNATION NUMBER <b>CE-3</b>
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Any alteration, reproduction, or misuse of this certificate may be punishable by a fine not exceeding \$1,000, or imprisonment not exceeding 3 years, or both. THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.

FAA Form 1362 (7-65)

\* GPO : 1966 O - 798-353

0052-040-8000 (8100)

ICRO

UNITED STATES OF AMERICA FEDERAL AVIATION AGENCY		FTW EMDG 2-0-42
<b>CERTIFICATE OF AIRWORTHINESS</b>		
1. NATIONALITY AND REGISTRATION MARKS 310CM	2. AIRCRAFT AIRWORTHINESS CLASSIFICATION EXPERIMENTAL-TO SHOW COMPLIANCE WITH FAR (SEE REVERSE SIDE)	
3. This Certificate of Airworthiness is issued pursuant to the Federal Aviation Act of 1958. The aircraft identified hereon is considered airworthy when maintained and operated in accordance with the Federal Aviation Regulations and applicable aircraft Operation Limitations.		
4. This Certificate will remain in effect as long as the aircraft is maintained in accordance with Part 43 of the Federal Aviation Regulations unless suspended, annulled, or revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Agency. Expires Jan. 10, 1979		
5. DATE OF ISSUANCE 10/9/69	6. SIGNATURE OF REPRESENTATIVE <i>Charles L. Harris</i> Charles L. Harris	7. DESIGNATION NO. S4-EMDG-42
8. Any alteration or misuse of this Certificate is punishable by a fine of not exceeding \$1,000 or imprisonment not exceeding 3 years, or both.		

1. Operational limitations dated 10/9/67 are a part of this certificate.
2. This aircraft does not comply with the International Airworthiness Standards of ICAO, Annex 8, Part 11. Special permission must be obtained for flights over foreign countries.
3. Flights for purposes other than to show compliance with FAR prohibited.

END.

MICRO

Form Approved  
Budget Bureau No. 04-R058.2

FEDERAL AVIATION AGENCY APPLICATION FOR AIRWORTHINESS CERTIFICATE (FAR 21)				INSTRUCTIONS: Print or type. These are for FAA use only. Authorized FAA Representative, use reverse side.		Do not write in shaded areas. Submit original only to an office if additional space is required.		
I. AIRCRAFT DESCRIPTION	1. REGISTRATION MARK	2. AIRCRAFT MAKE	3. AIRCRAFT MODEL DESIGNATION	FAA CODING				
	N310CM	Cessna	310P	2074238				
	4. AIRCRAFT SERIAL NO.	5. ENGINE MAKE	6. ENGINE MODEL DESIGNATION	17027				
	310POL78	Continental	IO-470-VO	1969				
	7. NO. OF ENGINES	8. PROPELLER MAKE	9. PROPELLER MODEL DESIGNATION	15. AIRC. MANUFACTURED				
	Two	McCauley	DAF34C71/82JF-3					
II. CERTIFICATION REQUESTED	A. APPLICATION IS HEREBY MADE FOR:							
	3. <input checked="" type="checkbox"/> ORIGINAL CERTIFICATE—AIRCRAFT IS <input checked="" type="checkbox"/> NEW <input type="checkbox"/> USED <input type="checkbox"/> IMPORT							
	4. <input type="checkbox"/> AMENDMENT OR MODIFICATION OF CURRENT CERTIFICATE							
	5. <input type="checkbox"/> OTHER (Specify)							
	B. AIRWORTHINESS CLASSIFICATION (For multiple certification, check appropriate items)							
	1. <input checked="" type="checkbox"/> STANDARD (Indicate category) <input checked="" type="checkbox"/> NORMAL <input type="checkbox"/> UTILITY <input type="checkbox"/> ACROBATIC <input type="checkbox"/> TRANSPORT							
	2. <input type="checkbox"/> LIMITED							
	3. <input type="checkbox"/> RESTRICTED (Indicate operation(s) to be conducted)							
	4. <input type="checkbox"/> EXPERIMENTAL (Indicate operation(s) to be conducted)							
	5. <input type="checkbox"/> PROVISIONAL (Indicate class)							
III. OWNER'S CERTIFICATION	A. REGISTERED OWNER (As shown on Certificate of Aircraft Registration)							
	NAME			ADDRESS (Number, street, city, State and ZIP code)				
	Cessna Aircraft Company			West K-42 Highway, P.O. Box 1977 Wichita, Kansas 67201				
	B. AIRCRAFT CERTIFICATION BASIS							
	X. AIRCRAFT SPECIFICATION OR TYPE CERTIFICATION DATA SHEET (Give No. and Rev. No.)			X. AIRWORTHINESS DIRECTIVE SUMMARY (Give year and list card No.)				
	3A10 Rev. 32			1968 69-13				
	AIRCRAFT LISTING (Give page No(s.))			SUPPLEMENTAL TYPE CERTIFICATE (Give No(s.))				
	C. AIRCRAFT OPERATION AND MAINTENANCE RECORDS							
	X. AIRCRAFT NEW—NO PREVIOUS HISTORY			TOTAL AIRFRAME HOURS RECORDED		FAA CODING		
	RECORDS MAINTAINED IN COMPLIANCE WITH FAR 91.173			1:35 hrs.				
IV. INSPECTION AGENCY VERIFICATION	I hereby certify that I am the registered owner (or his agent) of the aircraft described above; that the aircraft is registered with the Federal Aviation Agency in accordance with section 501 of the Federal Aviation Act of 1958, and applicable Federal Aviation Regulations; and that the aircraft has been inspected and is airworthy and eligible for the classification(s) requested.							
	DATE OF APPLICATION		NAME AND TITLE (Print or type)		SIGNATURE			
	7-11-69		W. H. Prewitt, Quality Control Manager		<i>W. H. Prewitt</i>			
	IN ACCORDANCE WITH FAR 21.183, THE AIRCRAFT DESCRIBED ABOVE HAS BEEN INSPECTED AND FOUND AIRWORTHY BY:							
	2. CERTIFICATED AIR CARRIER (Give Certificate No.)		3. CERTIFICATED MECHANIC (IA) (Give Certificate No.)		6. CERTIFICATED DOMESTIC REPAIR STATION (Give Certificate No.)			
	5. AIRCRAFT MANUFACTURER (Give name of firm)		FAA CODING					
			4					
	DATE		TITLE		SIGNATURE			
V. AIRWORTHINESS DOCUMENTATION	A. CURRENT OPERATING LIMITATIONS AVAILABLE IN AIRCRAFT			X. G. THIS INSPECTION RECORDED IN AIRCRAFT RECORDS				
	B. CURRENT OPERATING LIMITATIONS ATTACHED			H. STATEMENT OF CONFORMITY, FAA FORM 317 (Attach when required)				
	C. CURRENT APPROVED FLIGHT MANUAL AVAILABLE IN AIRCRAFT			I. PREVIOUS AIRWORTHINESS CERTIFICATE ISSUED IN ACCORDANCE WITH				
	D. DATA, DRAWINGS, PHOTOGRAPHS, ETC. (Attach when required)			FAR _____ CAR _____ (Orig. attached)				
	X. E. CURRENT WEIGHT & BALANCE INFORMATION AVAILABLE IN AIRCRAFT			J. AIRWORTHINESS CERTIFICATE ISSUED IN ACCORDANCE WITH				
	F. MAJOR REPAIR AND ALTERATION, FAA FORM 337 (Attach when req.)			X. FAR 21.183a (Copy attached)				
	A. I have inspected the aircraft described above, find it conforms to its type certificate, and is in condition for safe operation.							
	B. I have inspected the aircraft described above, find it is in condition for safe experimental operation.							
	C. Current Airworthiness Certificate amended.							
	VI. FAA REPRESENTATIVE CERTIFICATION	DATE		DESIGNATION		OFFICE NO.		SIGNATURE
7-11-69		Delegation		ICT EMDO		<i>Raymond M. Rowden</i>		
		Option PC 312		3-0-43		Raymond M. Rowden		

FAA AIRCRAFT REGISTRATION  
CAMERA NO. 4N DATE: 12-29-83  
[Faint, mostly illegible text from the registration form, including fields for aircraft identification, owner information, and registration details.]

OKLAHOMA CITY, OKLA  
JUL 15 12 20 PM '89

CONVEYANCE  
REGISTRY  
FAA AIRCRAFT REGISTRATION

Punched by Operator 148

UNITED STATES OF AMERICA - FEDERAL AVIATION AGENCY  
STANDARD AIRWORTHINESS CERTIFICATE

1. NATIONALITY AND REGISTRATION MARKS <b>N310CM</b>	2. MANUFACTURER AND MODEL <b>Cessna 310P</b>	3. AIRCRAFT SERIAL NUMBER <b>310P0178</b>	4. CATEGORY <b>Normal</b>
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5. AUTHORITY AND BASIS FOR ISSUANCE  
This airworthiness certificate is issued pursuant to the Federal Aviation Act of 1958 and certifies that, as of the date of issuance, the aircraft to which issued has been inspected and found to conform to the type certificate therefor, to be in condition for safe operation, and has been shown to meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex B to the Convention on International Civil Aviation, except as noted herein.  
Exceptions: **NONE**

6. TERMS AND CONDITIONS  
Unless sooner surrendered, suspended, revoked, or a transition date is otherwise established by the Administrator, this airworthiness certificate is effective as long as the aircraft is maintained in accordance with the applicable provisions of Parts 23, 25, 27, 29, 31, and 33 of the Federal Aviation Regulations, or appropriate, and the aircraft is registered in the United States.

DATE OF ISSUANCE <b>7-11-69</b>	FAA REPRESENTATIVE <b>ESYON</b>	DESIGNATION NUMBER <b>CB-3</b>
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Any alteration, reproduction, or misuse of this certificate may be punishable by a fine not exceeding \$1,000, or imprisonment not exceeding 3 years, or both. THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.

FAA Form 1362 (7-65) \* GPO : 1966 O - 798-355 0052-040-8000 (8100)

FAA AIRCRAFT REGISTRY  
CAMERA NO. 41 DATE: 12-29-83

