

ENGINE LIFE LIMITED PARTS STATUS / INSPECTION STATUS

PRATT & WHITNEY PW210A - ENGINE # 1, P/N 30L1860-01, S/N PCE-BP0030 - AW169, S/N 69025

T.S.N.:	177,9
C.S.N.:	331
T.S.O.:	N/A
C.S.O.:	N/A
TSHSI / MPI:	N/A
CSHSI / MPI	N/A

MFG DATE	22-feb-16
OH DATE	N/A
OH ETT	N/A
OH ETCy	N/A
H.S.I. DATE	N/A
H.S.I. ETT	N/A

DESCRIPTION	DATE LAST C/W	HOURS LAST C/W	CYCLCS LAST C/W	INTERVAL	NEXT DUE	REMAINING
ENGINE OVERHAUL				4000	4000	3822,1
HOT SECTION INSPECTION				O/C		
AIR-COOLED OIL COOLER (ACOC)				4500	4500	4322,1
FUEL CONTROL UNIT (FCU)				4500	4500	4322,1
FLOW DIVIDER				4500	4500	4322,1
FUEL-OIL HEAT EXCHANGER (FOHE)				4500	4500	4322,1
INLET-GUIDE VANE (IGV) ACTUATOR				4500	4500	4322,1
12 HR INSP.	22-jun-22	177,9	331	12	189,9	12
50 HR INSPECTION - POWER ASSURANCE CHECK	22-jun-22	177,9	331	50	227,9	50
12 MONTH INSPECTION	22-jun-22	177,9	331	12 MO	30/jun/23	373
400 HR / 12 MONTH INSP.	22-jun-22	177,9	331	400	577,9	400
	22-jun-22	177,9	331	12 MO	30/jun/23	373
600 HR INSPECTION				600	600	422,1
800 HR INSPECTION				800	800	622,1
800 HR / 24 MONTH INSP.	29-dic-20	162,6	299	800	962,6	784,7
				24 MO	31/dic/22	192
OIL FILTER REPLACEMENT				900	900	722,1
FUEL FILTER REPLACEMENT				900	900	722,1
1600 HR / 48 MONTH INSP.	29-dic-20	162,6	299	1600	1762,6	1584,7
				48 MO	01/dic/24	893
1600 HR INSPECTION				1600	1600	1422,1

ENGINE LIFE LIMITED PARTS STATUS / INSPECTION STATUS

PRATT & WHITNEY PW210A - ENGINE # 1, P/N 30L1860-01, S/N PCE-BP0030 - AW169, S/N 69025

T.S.N.:	177,9			MFG DATE	22-feb-16	
C.S.N.:	331			OH DATE	N/A	
T.S.O.:	N/A			OH ETT	N/A	
C.S.O.:	N/A			OH ETCy	N/A	
TSHSI / MPI:	N/A			H.S.I. DATE	N/A	
CSHSI / MPI	N/A			H.S.I. ETT	N/A	
DISCARD THE FIRST STAGE POWER TURBINE BLADES				12000 CY	12000 CY	11669
DISCARD THE SECOND STAGE POWER TURBINE BLADES				12000 CY	12000 CY	11669

Composante	P/N	S/N	C.S.N. / OH @ install.:	T.S.N./ OH @ install.:	Date of installation	Cycles since installation	Hours since installation	Part Life Limit (Cycles)	Next Due	Remaining Cycles	Remaining Hours
COMPRESSOR TURBINE COVER - REAR	38L1810-01	A003LCHC	0	0	25/may/15	331	177,9	10500	10500	10169	
CENTRIFUGAL IMPELLER	30L2036-01	A003KYL4	0	0	25/may/15	331	177,9	12000	12000	11669	
COMPRESSOR TURBINE DISC	30L2221-01	A003LARX	0	0	25/may/15	331	177,9	8330	8330	7999	
HIGH PRESSURE COMPRESSOR ROTOR	30L2531-01	FRAEPTDD105	0	0	25/may/15	331	177,9	12000	12000	11669	
POWER TURBINE DISC - SECOND STAGE	30M0417-01	MHAY000A679	0	0	25/may/15	331	177,9	12000	12000	11669	
POWER TURBINE DISC - FIRST STAGE	30M0723-01	MHAY000A651	0	0	25/may/15	331	177,9	12000	12000	11669	

ENGINE LIFE LIMITED PARTS STATUS / INSPECTION STATUS

PRATT & WHITNEY PW210A - ENGINE # 2, P/N 30L1860-01, S/N PCE-BP0015 - AW169, S/N 69025

T.S.N.:	173,3
C.S.N.:	309
T.S.O.:	N/A
C.S.O.:	N/A
TSHSI / MPI:	N/A
CSHSI / MPI	N/A

MFG DATE	22-feb-16
OH DATE	N/A
OH ETT	N/A
OH ETCy	N/A
H.S.I. DATE	N/A
H.S.I. ETT	N/A

DESCRIPTION	DATE LAST C/W	HOURS LAST C/W	CYCLCS LAST C/W	INTERVAL	NEXT DUE	REMAINING
ENGINE OVERHAUL				4000	4000	3826,7
HOT SECTION INSPECTION				O/C		
AIR-COOLED OIL COOLER (ACOC)				4500	4500	4326,7
FUEL CONTROL UNIT (FCU)				4500	4500	4326,7
FLOW DIVIDER				4500	4500	4326,7
FUEL-OIL HEAT EXCHANGER (FOHE)				4500	4500	4326,7
INLET-GUIDE VANE (IGV) ACTUATOR				4500	4500	4326,7
12 HR INSP.	22-jun-22	173,3	309	12	185,3	12
50 HR INSPECTION - POWER ASSURANCE CHECK	22-jun-22	173,3	309	50	223,3	50
12 MONTH INSPECTION	22-jun-22	173,3	309	12 MO	30/jun/23	373
400 HR / 12 MONTH INSP.	22-jun-22	173,3	309	400	573,3	400
	22-jun-22	173,3	309	12 MO	30/jun/23	373
600 HR INSPECTION				600	600	426,7
800 HR INSPECTION				800	800	626,7
800 HR / 24 MONTH INSP.	29-dic-20	158,5	279	800	958,5	785,2
				24 MO	31/dic/22	192
OIL FILTER REPLACEMENT				900	900	726,7
FUEL FILTER REPLACEMENT				900	900	726,7
1600 HR / 48 MONTH INSP.	29-dic-20	158,5	279	1600	1758,5	1585,2
				48 MO	31/dic/24	923
1600 HR INSPECTION				1600	1600	1426,7

ENGINE LIFE LIMITED PARTS STATUS / INSPECTION STATUS

PRATT & WHITNEY PW210A - ENGINE # 2, P/N 30L1860-01, S/N PCE-BP0015 - AW169, S/N 69025

T.S.N.:	173,3			MFG DATE	22-feb-16	
C.S.N.:	309			OH DATE	N/A	
T.S.O.:	N/A			OH ETT	N/A	
C.S.O.:	N/A			OH ETCy	N/A	
TSHSI / MPI:	N/A			H.S.I. DATE	N/A	
CSHSI / MPI	N/A			H.S.I. ETT	N/A	
DISCARD THE FIRST STAGE POWER TURBINE BLADES				12000 CY	12000 CY	11691
DISCARD THE SECOND STAGE POWER TURBINE BLADES				12000 CY	12000 CY	11691

Composante	P/N	S/N	C.S.N. / OH @ install.:	T.S.N. / OH @ install.:	Date of installation	Cycles since installation	Hours since installation	Part Life Limit (Cycles)	Next Due	Remaining Cycles	Remaining Hours
COMPRESSOR TURBINE COVER - REAR	38L1810-01	A003L7MX	0	0	22/abr/15	309	173,3	10500	10500	10191	
CENTRIFUGAL IMPELLER	30L2036-01	A003KW4K	0	0	22/abr/15	309	173,3	12000	12000	11691	
COMPRESSOR TURBINE DISC	30L2221-01	A003LAT1	0	0	22/abr/15	309	173,3	8330	8330	8021	
HIGH PRESSURE COMPRESSOR ROTOR	30L2531-01	FRAEPPHK33	0	0	22/abr/15	309	173,3	12000	12000	11691	
POWER TURBINE DISC - SECOND STAGE	30M0417-01	MHAY000A577	0	0	22/abr/15	309	173,3	12000	12000	11691	
POWER TURBINE DISC - FIRST STAGE	30M0723-01	MHAY000A544	0	0	22/abr/15	309	173,3	12000	12000	11691	

AIRWORTHINESS DIRECTIVES STATUS

AW169, S/N 69025, Reg # LV-GYK

AIRCRAFT TOTAL TIME	173,3
AIRCRAFT TOTAL CYCLES	424
STATUS DATE	22-jun-22

AD N°	DESCRIPTION	APPL.	DATE LAST C/W	HOURS LAST C/W	CYCLES LAST C/W	REC.	INTERV.	NEXT DUE	REMAIN. TIME	STATUS / METHOD OF COMPL. / NOTES	PERFORMED BY
EASA AD 2017-0110	LANDING GEAR - RETRACTION ACTUATORS - INSTALLATION / REPLACEMENT (S/N 69025 IS GROUP 2) / PART V OF SB 023 DOES NOT APPLY AS BT 169-018 HAS BEEN COMPLIED WITH AT MANUFACTURE (UPGRADED LANDING GEAR)	Y						SUPERSEDED BY EASA AD 2017-0164		MODIFICATION, PARAGRAPH (2), COMPLY PART VI (INSTALLATION CHECKS) & PART IV OR V (MODIFICATION) OF SBA 169-023	
		Y	28-jul-17	28,6		Y	100 HR / 3 MO			INSPECTION, PARAGRAPH (4), COMPLY PART VII OF SBA 169- 023	MODENA TAR
EASA AD 2017-0112	RFM - LIMITATIONS SECTION - AMENDMENT	Y	28-jul-17	28,6		N				SUPERSEDED BY EASA AD 2017- 0156	MODENA TAR
EASA AD 2017-0137	COLLECTIVE FIXED FRICTION FORCE - CHECK	Y	30-ago-17	45,4	80	N		SUPERSEDED ON 12- SEPT-2018 BY EASA AD 2018-0203		30 FH AFTER AUGUST 15, 2017 / 30 DAYS AFTER AUG-15-2017 / MODENA	MODENA TAR
EASA AD 2017-0155	EMERGENCY EXIT WINDOW SEAL, NON- METALLIC CHANNEL AND HANDLE - REPLACEMENT	N	2-oct-17	47,9	91	N				NOT APPLICABLE PER A/C S/N / MODENA	MODENA TAR
EASA AD 2017-0156	AUTOMATIC FLIGHT CONTROL - SOFTWARE UPDATE / RFM - LIMITATIONS SECTION - AMENDMENT (SUPERSEDES AD EASA 2017- 0112) / SB169-064	Y	7-dic-17	56,2	124	N				INSTALL SOFTWARE UPGRADE WITHIN 100 HR / 3 MONTH SINCE SEPTEMBER 07, 2017	MODENA TAR
EASA AD 2017-0164	LANDING GEAR - RETRACTION ACTUATORS - INSTALLATION / REPLACEMENT (S/N 69025 IS GROUP 2)	Y	13-abr-18	58,9	133	N				MODIFICATION, PARAGRAPH (2), COMPLY PART VI (INSTALLATION CHECKS) & PART V (MODIFICATION) OF SBA 169-023	MODENA TAR
		Y	22-jun-22	173,3	424	Y	100 HR / 3 MO	273,3	 100	INSPECTION, PARAGRAPH (4), COMPLY PART VII OF SBA 169- 023	MODENA TAR
								20/sep/22	 90		
EASA AD 2017-0255	WOW BONDING IMPROVEMENT (SB 169-047)	Y	13-abr-18	58,9	133	N				PART I ACCOMPLISHED / PART II NOT APPLICABLE	MODENA TAR

AD N°	DESCRIPTION	APPL.	DATE LAST C/W	HOURS LAST C/W	CYCLES LAST C/W	REC.	INTERV.	NEXT DUE	REMAIN. TIME	STATUS / METHOD OF COMPL. / NOTES	PERFORMED BY
EASA AD 2018-0197	EMERGENCY EXIT WINDOW - MODIFICATION	Y	28-sep-18	93,7	221	N				SB 169-094	MODENA TAR
EASA AD 2018-0203	PILOT COLLECTIVE STICK ASSEMBLY - FRICTION CHECK	Y	31-oct-18	94	223	N				SBA 169-082	MODENA TAR
EASA AD 2018-0208	EMERGENCY FLOTATION SYSTEM - AFT FLOAT ASSEMBLY INFLATION SYSTEM	N	5-nov-18	95	230	N				NOT APPLICABLE PER FLOTATION SYSTEM NOT INSTALLED	SAS S.A.
EASA EAD 2018-0241-E	TAIL ROTOR FLIGHT CONTROL SYSTEM - INSPECTION	Y	8-nov-18	95,5	232	N	PREVIOUSLY COMPLIED WITH SBA 169-120 @ MODENA TAR ON NOV-06-2018. SUPERSEDED BY EASA EAD 2018-0250-E				SAS S.A.
EASA EAD 2018-0250-E	TAIL ROTOR FLIGHT CONTROL SYSTEM - INSPECTION	Y	21-nov-18	96,5	234	N	SUPERSEDES EASA EAD 2018-0241-E. SUPERSEDED BY EASA AD 2018-0252-E				MODENA TAR
EASA EAD 2018-0252-E	TAIL ROTOR FLIGHT CONTROL SYSTEM - INSPECTION	Y	22-nov-18	96,7	235	N	SUPERSEDES EASA AD 2018-0250-E. CORRESPONDS TO SBEA 169-125				MODENA TAR
EASA EAD 2018-0261-E	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART I (CORRESPONDS TO SBEA 169-126)	Y	21-ene-19	110,1	259	Y	SUPERSEDED BY EASA AD 2019-0223				MODENA TAR
EASA EAD 2018-0261-E	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART II (CORRESPOND TO SBEA 169-126)	Y	3-dic-18	107,6	248	Y	SUPERSEDED BY EASA AD 2019-0223				MODENA TAR
FAA 2018-16-01	B/E AEROSPACE FISCHER PILOT SEATS	N	31-oct-18	94	223	N				VERIFIED NOT APPLICABLE	MODENA TAR
EASA 2019-0019	CONFORMITY CHECK OF TRA INPUT LEVER ASSY INSTALLATION. CORRESPONDS TO SBA 169-129	Y	5-feb-19	115,1	267	N					MODENA TAR
EASA 2019-0022	TAIL ROTOR - DUPLEX BEARING - REPLACEMENT. CORRESPONDS TO SBA 169-134	N	5-feb-19	115,1	267	N				N/A * BRG S/N (INSTALLED BRG IS S/N 15118)	MODENA TAR
EASA EAD 2019-0023	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 1 (HELICOPTER BELONGS TO GROUP I). CORRESPONDS TO SBEA 169-135 PART I - SLIPPAGE MARKING	Y	15-abr-19	120,9	285	Y	10 FH	SUPERSEDES EASA EAD 2018-0261-E / SUPERSEDED BY EASA EAD 2019-0121			MODENA TAR
EASA EAD 2019-0023	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 1 (HELICOPTER BELONGS TO GROUP I). CORRESPONDS TO SBEA 169-135 PART II - SLIPPAGE MARKING	Y	5-feb-19	115,1	267	Y	50 FH	SUPERSEDES EASA EAD 2018-0261-E / SUPERSEDED BY EASA EAD 2019-0121			MODENA TAR
EASA EAD 2019-0023	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 2. CORRESPONDS TO SBEA 169-135 PART III. REPETITIVE THERMAL STRIP INSTALLATION	Y	15-abr-19	120,9	285	Y	20 FH	SUPERSEDES EASA EAD 2018-0261-E / SUPERSEDED BY EASA EAD 2019-0121			MODENA TAR

AD N°	DESCRIPTION	APPL.	DATE LAST C/W	HOURS LAST C/W	CYCLES LAST C/W	REC.	INTERV.	NEXT DUE	REMAIN. TIME	STATUS / METHOD OF COMPL. / NOTES	PERFORMED BY
EASA EAD 2019-0023	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 3. CORRESPONDS TO SBEA 169-135 PART IV. REPETITIVE THERMAL STRIP CHECK	Y	3-dic-18	107,6	248	Y	10 FH			SUPERSEDES EASA EAD 2018-0261-E / SUPERSEDED BY EASA EAD 2019-0121	
EASA EAD 2019-0121 R1	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 1 (HELICOPTER BELONGS TO GROUP I). CORRESPONDS TO SBEA 169-135 PART I - SLIPPAGE MARKING	Y	5-jul-19	123,6	294	Y	10 FH			SUPERSEDES EASA EAD 2019-0023 / SUPERSEDED BY EASA AD 2019-0193	MODENA TAR
EASA EAD 2019-0121 R1	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 1 (HELICOPTER BELONGS TO GROUP I). CORRESPONDS TO SBEA 169-135 PART II - ROUGHNESS AND BREAKAWAY FORCE OF THE T/R DUPLEX BEARING	Y	5-feb-19	115,1	267	Y	50 FH			SUPERSEDES EASA EAD 2019-0023 / SUPERSEDED BY EASA AD 2019-0193	
EASA EAD 2019-0121 R1	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 2. CORRESPONDS TO SBEA 169-135 PART III. REPETITIVE THERMAL STRIP INSTALLATION	Y	15-abr-19	120,9	285	Y	20 FH			SUPERSEDES EASA EAD 2019-0023 / SUPERSEDED BY EASA AD 2019-0193	
EASA EAD 2019-0121 R1	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 3. CORRESPONDS TO SBEA 169-135 PART IV. REPETITIVE THERMAL STRIP CHECK	Y	5-jul-19	123,6	294	Y	10 FH			SUPERSEDES EASA EAD 2019-0023 / SUPERSEDED BY EASA AD 2019-0193	MODENA TAR
EASA EAD 2019-0121 R1	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 4. CORRESPONDS TO SBEA 169-135 PART V. REPETITIVE INSPECTION OF T/R DUPLEX BEARING (PARTICLES)	Y	5-jul-19	123,6	294	Y	10 FH			SUPERSEDES EASA EAD 2019-0023 / SUPERSEDED BY EASA AD 2019-0193	MODENA TAR
EASA EAD 2019-0121 R1	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 4. CORRESPONDS TO SBEA 169-135 PART VI. REPETITIVE INSPECTION OF T/R DUPLEX BEARING (ADDITIONAL ROUGHNESS)	Y	5-jul-19	123,6	294	Y	200 FH			SUPERSEDES EASA EAD 2019-0023 / SUPERSEDED BY EASA AD 2019-0193	MODENA TAR
EASA AD 2019- 0193	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 1 (HELICOPTER BELONGS TO GROUP I). CORRESPONDS TO SBEA 169-135 PART I - SLIPPAGE MARKING	Y	12-sep-19	129,9	307	Y	10 FH			SUPERSEDES EASA EAD 2019-0121R1 / SUPERSEDED BY EASA AD 2020-0048	MODENA TAR
EASA AD 2019- 0193	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 1 (HELICOPTER BELONGS TO GROUP I). CORRESPONDS TO SBEA 169-135 PART II - ROUGHNESS AND BREAKAWAY FORCE OF THE T/R DUPLEX BEARING	Y	5-feb-19	115,1	267	Y	50 FH			SUPERSEDES EASA EAD 2019-0121R1 / SUPERSEDED BY EASA AD 2020-0049	MODENA TAR

AD N°	DESCRIPTION	APPL.	DATE LAST C/W	HOURS LAST C/W	CYCLES LAST C/W	REC.	INTERV.	NEXT DUE	REMAIN. TIME	STATUS / METHOD OF COMPL. / NOTES	PERFORMED BY
EASA AD 2019-0193	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 2. CORRESPONDS TO SBEA 169-135 PART III. REPETITIVE THERMAL STRIP INSTALLATION	Y	15-abr-19	120,9	285	Y	20 FH	SUPERSEDES EASA EAD 2019-0121R1 / SUPERSEDED BY EASA AD 2020-0050			MODENA TAR
EASA AD 2019-0193	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 3. CORRESPONDS TO SBEA 169-135 PART IV. REPETITIVE THERMAL STRIP CHECK	Y	12-sep-19	129,9	307	Y	10 FH	SUPERSEDES EASA EAD 2019-0121R1 / SUPERSEDED BY EASA AD 2020-0051			MODENA TAR
EASA AD 2019-0193	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 4. CORRESPONDS TO SBEA 169-135 PART V. REPETITIVE INSPECTION OF T/R DUPLEX BEARING (PARTICLES)	Y	12-sep-19	129,9	307	Y	10 FH	SUPERSEDES EASA EAD 2019-0121R1 / SUPERSEDED BY EASA AD 2020-0052			MODENA TAR
EASA AD 2019-0193	TAIL ROTOR SERVO ACTUATOR - INSPECTION / PART 4. CORRESPONDS TO SBEA 169-135 PART VI. REPETITIVE INSPECTION OF T/R DUPLEX BEARING (ADDITIONAL ROUGHNESS)	Y	5-jul-19	123,6	294	Y	200 FH	SUPERSEDES EASA EAD 2019-0121R1 / SUPERSEDED BY EASA AD 2020-0053			MODENA TAR
EASA AD 2019-0252	ROTORS FLIGHT CONTROL - YAW PEDAL ASSEMBLIES - MODIFICATION / REWORK (SB A169-147)	Y	15-nov-19	130,2	308	N					MODENA TAR
EASA AD 2020-0013	Fire Protection – Hand-held Fire Extinguishers – Replacement	N	29-dic-20	158,5	373	N				N/A BY P/N INSTALLED	MODENA TAR
EASA AD 2020-0048	ATA 64 – Tail Rotor – Tail Rotor Servo Actuator / Duplex Bearing – Inspection / Check / Modification	Y					SUPERSEDES EASA EAD 2019-0193 / SUPERSEDED BY EASA AD 2020-0197				
EASA AD 2020-0197	TR Servo Actuator / Duplex Bearing – Insp. / Check Modiciation.	N	N/A BY P/N TRA 6F6730V00332 INSTALLED								
	Part 1 (HELICOPTER BELONGS TO GROUP 3). CORRESPONDS TO ASB 169-178 Part I - CASTELLATED NUT SLIPPAGE MARKING		SUPERSEDES EASA EAD 2019-0121 R1								
EASA AD 2020-0197	TR Servo Actuator / Duplex Bearing – Insp. / Check Modiciation. Part 3 (HELICOPTER BELONGS TO GROUP 3). CORRESPONDS TO ASB 169-178 Part II - THERMAL STRIP INSTALLATION	Y	29-dic-20	158,5	373	Y	100 FH	258,5	85,2	SUPERSEDES EASA EAD 2019-0121 R1	

AD N°	DESCRIPTION	APPL.	DATE LAST C/W	HOURS LAST C/W	CYCLES LAST C/W	REC.	INTERV.	NEXT DUE	REMAIN. TIME	STATUS / METHOD OF COMPL. / NOTES	PERFORMED BY
EASA AD 2020-0197	TR Servo Actuator / Duplex Bearing – Insp. / Check Modiciation. Part 4 (HELICOPTER BELONGS TO GROUP 3). CORRESPONDS TO ASB 169-178 Part III - THERMAL STRIP CHECKS	Y	29-dic-20	158,5	373	Y	50 FH	208,5	 35,2	SUPERSEDES EASA EAD 2019- 0121 R1	
EASA AD 2020-0197	TR Servo Actuator / Duplex Bearing – Insp. / Check Modiciation. Part 5 (HELICOPTER BELONGS TO GROUP 3). CORRESPONDS TO ASB 169-178 Part II - PARTICLES	Y	29-dic-20	158,5	373	Y	50 FH	208,5	 35,2	SUPERSEDES EASA EAD 2019- 0121 R1	
EASA AD 2020-0197	TR Servo Actuator / Duplex Bearing – Insp. / Check Modiciation. Part 5 (HELICOPTER BELONGS TO GROUP 3). CORRESPONDS TO ASB 169-178 Part IV - ADDITIONAL ROUGHNESS	Y	29-dic-20	158,5	373	Y	100 FH	258,5	 85,2	SUPERSEDES EASA EAD 2019- 0121 R1	
EASA AD 2020-0197	TR Servo Actuator / Duplex Bearing – Insp. / Check Modiciation. Part 14 (HELICOPTER BELONGS TO GROUP 3). CORRESPONDS TO ASB 169-153 - TRA INSTALLATION	Y	29-dic-20	158,5	373	N				SUPERSEDES EASA EAD 2019- 0121 R1	

AD N°	DESCRIPTION	APPL.	DATE LAST C/W	HOURS LAST C/W	CYCLES LAST C/W	REC.	INTERV.	NEXT DUE	REMAIN. TIME	STATUS / METHOD OF COMPL. / NOTES	PERFORMED BY
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PRATT & WHITNEY PW210A - ENGINE # 1, P/N 30L1860-01, S/N PCE-BP0030 - AW169, S/N 69025

177,9	ENGINE #1 TOTAL TIME		331		ENGINE #1 TOTAL CYCLES					STATUS DATE	22-jun-22
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CF-2020-13	Engine – Manual Calculation of Cycle Count for the Impeller and High Pressure Compressor Rotor - PART I	N									
CF-2020-13	Engine – Manual Calculation of Cycle Count for the Impeller and High Pressure Compressor Rotor - PART II	Y						12000 CS	11701		

PRATT & WHITNEY PW210A - ENGINE # 2, P/N 30L1860-01, S/N PCE-BP0015 - AW169, S/N 69025

173,3	ENGINE #2 TOTAL TIME		309		ENGINE #2 TOTAL CYCLES					Date:	22-jun-22
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CF-2020-13	Engine – Manual Calculation of Cycle Count for the Impeller and High Pressure Compressor Rotor - PART I	N									
CF-2020-13	Engine – Manual Calculation of Cycle Count for the Impeller and High Pressure Compressor Rotor - PART II	Y						12000 CS	11721		