AIRCRAFT TECHNICAL RECORD

1958 Cessna 182B

N2376G

REGISTRATION

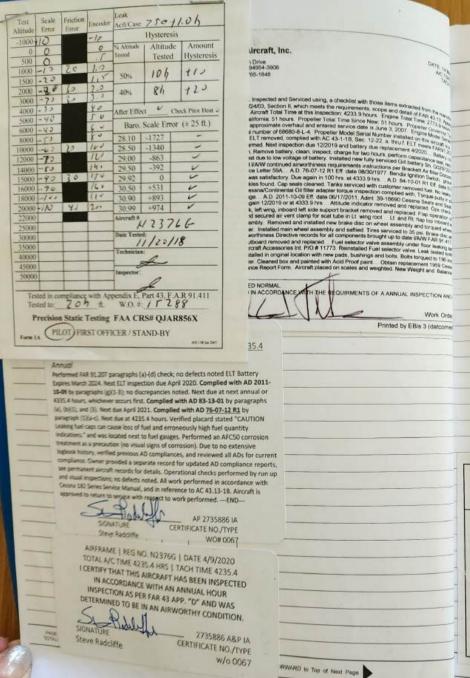
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AIRCRAFT DATA

1.	Nationality & Registration	N2376G
2.	Manufacturer's Designation (Make & Model)	CESSNA 182B
3.	Manufacturer's Serial Number	51676
4.	Type Approval or Specification Number	
5.	Date of Manufacture	1959
6.	Aircraft Inspection Cycle	



	Conf Aver LLC 300, delivered in the 18th Automatic CA speed, Indian Testing		1
	N2376G Cessna, 1828, 5/1: 51676 May 19, 2020		
	Removed LH stabilistic jack screw, disasembled, cleaned, lubricated and reinstalled. Op check good. Stabilistic place some impection parel on tall copie of aircraft using 0712083-1 doubler and With respect to the work performed, this aircrame is approved for return to service. Details of work performed after on the with Cord Aero LLC under work order; WO, 3255.	NAME/ GNATURE	LICENCE AMEOI AMOVACA
wo: WO_39	Cameralay	Annual Airtrame - page 1	of t
110000000	Cord Aero LLC - 395 delimonte center box 281 - Monterey, CA 93540 - 5594752710		
N2376G Tach: 42		May 17, 202	1
2. Dranar and a control of the contr	pection IAW FAR 91.207 (d). All AD's checked through this date, sined used engine oil and removed oil filter. Cut open and inspected oil filter sts found. Installed new oil filter PN AA48108-2 and secured. Serviced engin N50. Ran engine and performed leak check, no leaks noted. formed compression check of engine. Cyl 1) 68/80 Cyl 2) 44/80 Cyl 3) 76/8 80 Cyl 6) 64/80. I washed aircraft. Moved safety wire from alternator securing bolt and loosened bolt. Tightened the safety wired bolt then safety wired. Op check good. Moved old fuel strainer cable PN 0411090-29 and installed new PN 04110 ock and found op check good. Moved with AD 96-12-22 by (a) 1-3 and not evidence of oil leaking. Corrective action required at this time for AD 2020-21-22. Inspection due ich ever occurs later. Then every 500hrs or 5 years. wich ever occurs first. Implied with AD 83-13-01 IAW Para (a). (b)(1).(2) is not applicable by fuel of this AD. Inplied with AD 2011-10-09 by para (g) 1-3. Italled new beacon on belly. Lamp still inop. Rotating beacon on tail still open at this Alframe was inspected in accordance with FAR 43 Appendix D is specifion and was determined to be in an airworthy condition. This aircraft is as of this date. Details of work performed are on file with Cord Aero L. Staub A&P 3612277IA	e with 12 qts Phil 0 Cyl 4) 70/80 Cyl ed alternator belt 90-29. Performed by 12/7/21 or 433 cap part number, erational.	and dop dop do
A21-0082	FORCE AWATON		Adrame page tof
	Force Aviation 5555 Apollo Drivo Dallar, Texas 75237 2143377000		

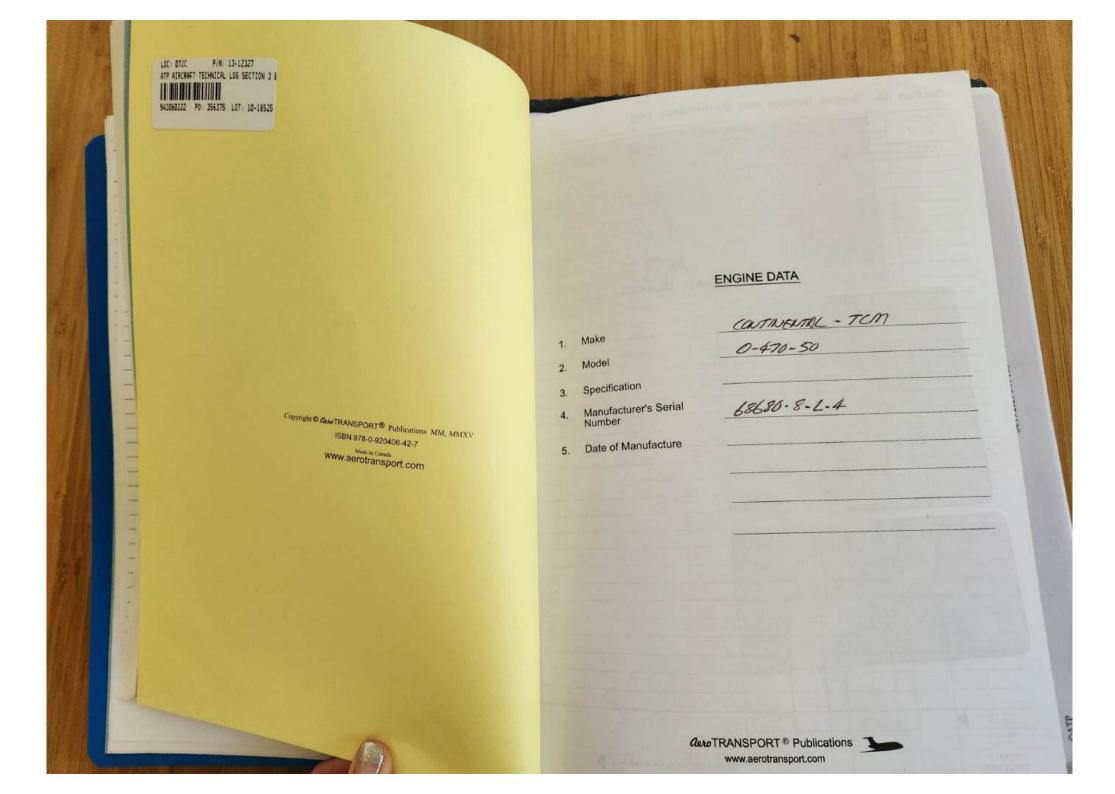
Installed: GMC 507 P/N 010-01946-01 S/N 5H1009031, Pitch Servo P/N 011-02927-11 S/N 5G9013486, Roll Servo P/N 011-02927-11 S/N 5G9013523, OAT Probe P/N 011-00978-00 S/N 47966030, and GAD 13 P/N 011-04938-00 S/N 5YG004782. Installed Garmin GFC 500 Autopitot system and integrated with existing GS system. Installation is approved under Garmin STC SA01866WI and was installed in reference to Garmin GFC 500 furbility in annual number 190-02291-00 Rev 18 and Installation Addendum 190-02291-22 Rev 7. Garmin FAA approved GFC 500 Airplane Flight Manual Supplement 190-02291-38 Rev 5 dated 09/28/2021 provided to customer and is available on board for pilot operation of the system. Garmin FAA approved GFC 500 MM Instructions for Continued Airworthinuss document 190-02291-01 Rev 9 dated 3 Mar 2021 provided to customer. The wiring harness and routing/installation practices were performed using acceptable perhods of AC 43.13-18 CH 11 and AC 43.13-2b CH 2.12. Installation was performed using acceptable practices and is found to meet requirements of 14 CFR part 43 found in AC 43.13-2b. No change to electrical load analysis. All affected systems were tested accordingly to show compliance with 14 CFR 23.1309. A complete EM/ground functional test has been performed to ensure proper operation thereof and to confirm that other aircraft systems are free form injecterence as a result of the installation. The aircraft equipment list and weight and balance were updated in aircraft POH.

AIRTIME LANDINGS/CYCLES				
DARLYTOTAL TOTAL TOTAL	DETAILS OF TARK III			2
HRS MIN HOURS MIN	DETAILS OF TASK (INSPECTION, REPAIRS & REPLACE			
	* REPLAN			LICENCE
FORCE AV	Arriane pige 2 of 2		NAME/ SIGNATURE	AME or AMO/ACA
				AMOTACA
Force Aviation - 5555 Apollo Orive - Delias		Clarel Association - Series destroyed review from 2011 - Nationary, CA 2004011 2011 ASS 10005		
pect to the work performed, this airframe is approved for retu- nation under work order: A21-0082.	The service. Details of work performed are on file with	93.4 February 22, 2022 ned used engine oil and removed oil filter PN AAAB109.0		
Alle for		ned used engine oil and removed oil Blar PN AA48108-2, call open and inspected oil Bler. PN A448108-2, call open and inspected oil Bler. No resided engine with 10 dis Philips 20Wild. No leaks noted. ones of the PN A448108-2 and		
winton under work order: A21-0082. Immons A&P, IA 3394413 Alay Jen		er ero, ed Adoption page 1 of 1		
		Cood Aars LLC - 385 Adverser center too 391 - Nordang CA 20040 - 351 - 355 - 1055		
		N2376G Cessna, 1828, S.N.: 51676 February 22, 2022		
		Drained used engine oil and removed oil filter PN AAA6106-2, out open and inspected oil filter. No major debris at this time. Serviced engine with 10 dis Phillips 20W50. No leaks noted.		
		Performed compression check on cylinder #2: Cyli #2) 52/90. Compiled with AD 96-12-22 by (a) 1-3 and not evidence of oil leaking.		
		With respect to the work performed, this airframe is approved for return to service. Details of work per- formed are on file with Cord Aero LLC under work order: WO_483.		
		Cordano Simo ASP 3612277/A CC		
		Tach: 4311		
		Performed Annual Inspection in accordance with FAR 43		
		app. (d). serviced wheel bearings inspected brake linings	4	
		lubed airframe, no defects noted. Cleaned aft battery comartment, applied corrosion X. Inspected ELT in		
		accordance with FAR 91,207 (d), replaced engine air filter		
		element BA 8103. Complied with AD 2011-10-09 sept track		
		inspection. Complied with AD 2020-21-22 horizontal stabilizer for cracks, no defects noted.		
		I certify that this airframe was inspected in accordance with Annual Inspection and was found to be in airworthy		
		condition.		
		Patrick Childs AP3334435IA		
		_		
		04/02/2024 Cessna 182 B S/N51676 N 2376G Tach 4311 7 T.T. 4311.7		
		Lubed control cables hinges pulleys rod ends hearings		
		C/VV AD2011-10-09 seat track inspection AD 2020-21-22		
		Tail cone and horizontal stabilizer I certify that this aircraft has been inspected in accordance		
		will an annual inspection and was determined to be in an		
		MI WORTHY COndition this date 04/02/2024		
		Johnny L. Clark A&P253803915IA Pohny I Clare		
		- ming 2 Clare		
GGE PALS				
	CARRY FORWARD to Top of Next Page			

AIRCRAFT TECHNICAL RECORD

Section 3. ENGINE LOG





CARRY FORWARD to T

SAL'S AIRCRAFT CYLINDERS, INC. **15963 PARVIN RD.** PROSPER, TX 75078 972-346-3339 Invoice #: 10450 Date: 6/27/22 W/O#: S-8108 Customer: Childs Aviation Patrick Childs 423-645-8680 300 Airport Rd. Jasper, TN 37347 Type: Continental TSIO-520 ITEM NUMBER DESCRIPTION QUANTITY PRICE TOTAL Stud assembly (S-8074-2) 1 \$500.00 \$500.00 SA24802 Valve keeper 4 \$3.10 \$12.40 SA643873 Exhaust valve 1 \$275.00 \$275.00 SA5000-SC5 Ring set 1 \$142.83 \$142.83 SA520-T1 Gasket set 1 \$38.00 \$38.00 SA642917-1 Induction hose \$25.00 3 \$75.00 SA628260 Exhaust gasket \$11.46 2 \$22.92 SA534857 Rocker cover gasket 2 \$9.73 \$19.46 TOTAL PARTS \$1,085.61 Face intake valve \$15.00 \$15.00 Clean piston Build to cylinder assembly \$35.00 \$35.00 TOTAL LABOR \$70.00 Shipping \$34.39 TOTAL DUE \$1,190.00

SAL'S AIRCRAFT CYLINDERS, INC. 15963 PARVIN RD PROSPER, TX 75078 972-346-3339 CERT # 3317562

TO THE PARTY	763	0500	655167
1/00/08	P/N: 1		10/
Date 6/27/28	M.O./N:		-
Model:			The second of
SN-S-BOLL	Valve assy		And have been
CH assy 🗹		sign	
Venity Zyglo Acceptance	date	Kus	
Installation of Parts	6/2/22	-	
Installation of Parts	Siz	Date & Sign	Inspection &
XIX		Sign	Date
	1000	10	1
Visual .	121	1/2 /20 1	111
Intike guide	.663	10 ET 55 VI	1922 82 MIS
Intake seat	- SERV	1622 35 W	1027 22 MIN
Exhaust guide	-635	627 22/1	027 22 Del
Exhaust scal	- SALV	427 22 By	027 22 Back
Spark plug insert front or upper		LE OI	1 190
Spark plug insert rear or lower		DORU	1
Rocker shaft bushing int outer	133 500	-	6/
Rocker shaft bushing int inner		1300	127
Rocker shaft bushing exh outer -		126 13	13-
Rocker shaft bushing exh inner	- Herman	1300	3
Intake coupling	** ***********	1	Soll
			143
Exhaust port		1	
*			

MAINTENANCE RELEASE

The aircraft engine part identified on the reverse side was repaired and impected in approved for return to service.

The aircraft engine part identified on the reverse side was repaired and impected in approved for return to service.

Salula Albuent X



Finish Machine Intalce guide ream Inspection & Date & Exhaust guide ream Size Intake seat grind Exhaust sent grind Rocker shaft bush int outer Rocker shaft bush int inner Rocker shaft bush exh outer Rocker shaft bush exh inner -Clean -Sand blast Paint-Final acceptance-INOTES: cyl Assy Some STO STEEL BOSE-In ale Man SA643873 Bah Valve Rev. AD LOT Z 19273 J SERU Value Sanny V alve Keeper SERV. 5464 fin Non SA5000-505 Ra ni totogra na beri. In kan stas most with to accompany the property of the Francis the second the rest in second

SERVICEABLE PART TAG L'S AIRCRAFT CYLINDERS, INC. 963 PARVIN RD - PROSPER, TX 75078 MODEL MAKE QUANTITY SERIAL NUMBER PART NUMBER CONDITION SELVICE ABLE 972-346-3339 PART NAME CUSTOMER MAINTENANCE RELEASE I certify that the repair identified on this tag was performed and inspected in conformance to Part 43, US Federal Air Regulations, and is approved for return to service

5-80742



SA5000-SC5 RING SET

FAA-PMA APPROVED FOR INSTALLATION

IN STEEL CYLINDER BORES

ENGINE MODEL	PISTONS
10-520-A.B.C.D.E.F.J.K.L.M	SA631475
- acc. (D'O'D'T') lot 1-1	SA648013
TSIO-520-B,C,D,E,G,H,J,K,L,N	632585
M.P.R.T.BB, DB, EB, IB, KB, LB,	642761
NB,UB,U,VB,WB	SA646303
	SA648044
	648033
GTSIO-520-C,D,F,G,H,J,K,L,M	640945,642761
	641883,641350
1925	SA646303,648033
IO-550 A,B,C,	648046,654726
	654857

RINGS-TOP TO BOTTOM

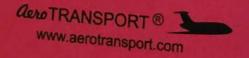
Туре
Compression
Compression
Oil Control
Scraper

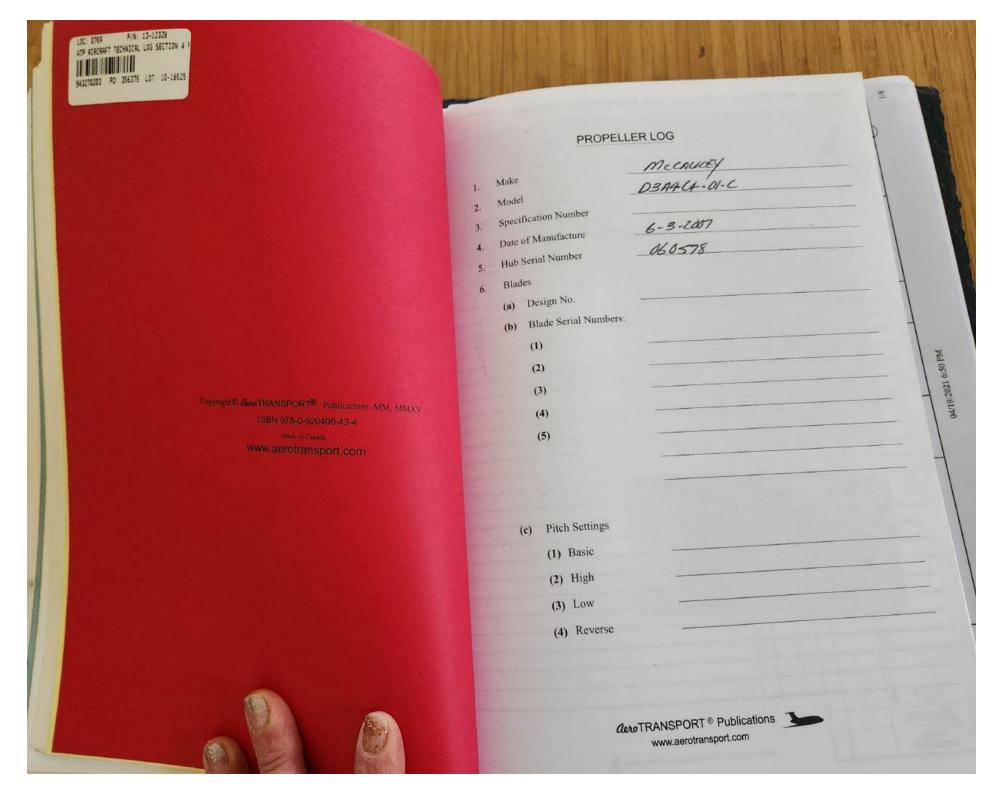
CAUTION:

THESE RINGS ARE NOT TO BE INSTALLED IN CYLINDERS WITH CHROME BORES

AIRCRAFT TECHNICAL RECORD

Section 4. PROPELLER LOG





FAA Airworthiness Directives Compliance Record

Company:

Cord Aero LLC

Category: Airframe

Manufacturer: Cessna Aircraft Company

Model: 182B

Position:

P/N:

S/N: 51676

Aircraft Registration No: N2376G

ATP Revision: 04/15/2021

Issue Number Effective Date Amendment #		Complied		Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2020-21-22 12/07/2020	[Recurring] To Detect and Correct Corrosion and Cracks in the Tailcone and Horizontal Stabilizer Attachment Structure. The Unsafe Condition, if Not Addressed, Could Result in Failure of the Horizontal Stabilizer to,contd.	Hrs: 4272.4	No corrective action required at this time. Inspection due by 12/7/21 or 4335.4, which ever occurs later. Then every 500hrs or 5 years. wich ever occurs first.	Yes	D: — Hrs: — C: —	AP3612277IA/Authorized Inspector Cordaro Staub
2011-10-09 06/17/2011	seat roller housing from departing the seat	04/14/2021 Hrs: 4272.4 C: -	complied with by para (g) 1-3	Yes	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Stau6
008-26-10 C 1/05/2009	altimeter, airspeed, and vertical speed, contd.	04/14/2021 Hrs: 4272.4 C:	Not applicable by part not installed	No	D: Hrs: C:	AP36122771A/Authorized Inspector Cordaro Staub
008-10-02 7/12/2008	To prevent erroneous indications from the altimeter, airspeed, and vertical speed, contd.	04/14/2021 Hrs: 4272.4 C:	Not applicable by part not installed	No	D: - Hrs: - C: -	AP36122771A/Authorized Inspecto
4-19-01 C 01/2004	shoulder harness, which could result in	04/14/2021 Hrs: 4272.4 C:	Not applicable by part number	No	D: Hrs: C:	AP3612277IA/Authorized Inspec

de Number effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
98-16-04 09/21/1998	TO PREVENT WING FAILURE DURING FLIGHT CAUSED BY THE ABSENCE OF AN ANGLE STIFFENER, WHICH COULD CAUSE LOSS OF, CONTD.	THE RESIDENCE OF THE PARTY OF T	Not applicable STC not installed	No	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Staub
97-01-13 02/03/1997	TO PREVENT FUEL, OIL, OR HYDRAULIC SYSTEMS FAILURE CAUSED BY A COLLAPSED HOSE	04/14/2021 Hrs: 4272.4 C:	Not applicable by parts number	No	D: - Hrs: - C: -	AP3612277IA/Authorized Inspector Cordaro Staub
96-12-22 07/31/1996	[Recurring] TO PREVENT LOSS OF ENGINE OIL CAUSED BY LOOSE OR SEPARATED OIL FILTER ADAPTERS, WHICH COULD RESULT IN ENGINE, CONTD.	04/14/2021 Hrs: 4272.4 C:	Complied with by (a) 1-3 and not evidence of oil leaking	Yes	D: - WHEN OIL Hrs: - FILTER S REMOVED	AP3612277IA/Authorized Inspector Cordaro Staub
84-10-01 R1 07/05/1988	[Recurring] TO PREVENT POWER LOSS OR ENGINE STOPPAGE DUE TO WATE CONTAMINATION OF FUEL SYSTEM	04/06/2020 Hrs: 4235.4 C:	Complied with by AD instructions (1) (I -iii), (2) (I-iii), and (3) (1) by borescope inspection. Visually verified quick drain valve installed and operable.	No	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Staub
79-10-14 R1 5/30/1988	에 있는데 있는데 있는데 있다면 없는데 10명에 되었다면서 보고 있는데 있다면 없는데 없다면 없는데 없다면	04/14/2021 Hrs: 4272.4 C:	Not applicable by SN listed in Cessna SL SE77-6. Fuel caps installed PN C15600: -0101		D: Hrs: C:	AP36122771A/Authorized Inspector
3-17-0 6 01/1983	DESTRUCTUVE AILERON FLUTTER	04/14/2021 Hrs: 4272.4 C:	Not applicable STC not installed	No	D: Hrs: C:	AP3612277IA/Authorized Inspecto Cordaro Staub
-13-01 01/1983	THE POTENTIAL EFFECTS OF	04/14/2021 Hrs: 4272.4 C:	Complied with Para (a), (b)(1),(2) is not applicable by fuel capart number, and (3) of this AD	Yes	D: 04/14/2022 Hrs; C:	AP3612277IA/Authorized Inspect
25-07 3/1979	ELECTRICAL OR ELECTRONIC	04/14/2021 Hrs: 4272.4 C:	Not applicable by serial number	No	D; Hrs; C:	AP36122771A/Authorized Inspe Cordaro Staub

ae Number Effective Date Amendment #	Description	Complied		Recur		Facility Cert No. / Type Authorized By Signed
78-01-14 01/23/1978	TO PRECLUDE THE POSSIBILITY OF AN IN-FLIGHT FIRE DUE TO A LOOSE ELT ANTENNA COAXIAL CABLE CONNECTOR, CONTD.	04/14/2021 Hrs: 4272.4 C: —	Not applicable by serial number	No		AP3612277IA/Authorized Inspector Cordaro Staub
77-23-11 11/28/1977	TO PRECLUDE THE POSSIBILITY OF AN IN-FLIGHT FIRE DUE TO A LOOSE ELT ANTENNA COAXIAL CABLE CONNECTOR, CONTD.	04/14/2021 Hrs: 4272.4 C:	Not applicable by aircraft serial number	No	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Staub
77-04-05 03/01/1977	TO PREVENT INGESTION OF THE INDUCTION AIR BOX SEAL INTO THE CARBURETOR	04/14/2021 Hrs: 4272.4 C:	PCW Complied with by paragraph (a) and (b). Inspection done by Radcliff Enterprices AP2735886IA Steve Radcliffe	No	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Staub
75-16-01 17/30/1975	TO PRECLUDE INADVERTENT FUEL EXHAUSTION DUE TO INCORRECT FUEL PLACARDED CAPACITIES	04/14/2021 Hrs: 4272.4 C:	not applicable by model of aircraft	No	D: - Hrs: - C: -	AP3612277IA/Authorized Inspector Cordaro Staub
75-05-02 2/21/1975		04/14/2021 Hrs: 4272.4 C:	Not applicable, STC not installed	No	D: - Hrs: - C: -	AP3612277IA/Authorized Inspecto Cordaro Staub
73-17-01 8/16/1973		04/14/2021 Hrs: 4272.4 C:	Not applicable by part number	No	D: Hrs: C:	AP3612277IA/Authorized Inspector
71-22-02 R(1) /09/1971		04/14/2021 Hrs: 4272.4 C:	Not applicable, nose gear P/N 0543018- 497 installed on 182 from 206	No	D: - Hrs: - C: -	AP36122771A/Authorized Inspecto
ategory: Engine		Position			ATP Revision	: 04/15/2021
anufacturer: Contine odel: O-470-L	ental Motors	P/N: O- S/N: 68	-470-50 3680-8-L-4			

Issue Number Effective Date Amendment #	Description		Method of			Facility Cert No. / Type Authorized By
	Description	Complied	Compliance	Recur	Next Due	Signed 07)

ae Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2016-16-12 09/15/2016	To prevent failure of the cylinder assemblies, which could lead to failure of the engine, in-flight,contd.	04/14/2021 Hrs: 4272.4 C:	Not applicable by part number installed	No	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Staub
2010-11-04 06/16/2010	To prevent excessive hydraulic lifter wear, which can result in loss of engine power & loss of control of the airplane	04/14/2021 Hrs: 4272.4 C:	Not applicable by overhaul date	No	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Staub
2009-19-07 C2 10/07/2009	[Recurring] To prevent loss of engine power due to cracks in the cylinder head, possible engine failure, and fire in, contd.	04/14/2021 Hrs: 4272.4 C:	Not applicable by part number installed	No	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Staub
2007-04-19 R1 05/07/2007	To prevent cylinder separation that can lead to engine failure, possible engine compartment fire, and,contd.	04/14/2021 Hrs: 4272.4 C:	Not applicable by part number installed	No	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Staub
00-00-08 03/26/2002		04/14/2021 Hrs: 4272.4 C:	Not applicable by installation of continental /P pond engine in 2007	No	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Staub
00-00-01 1/22/2001	Assemblies listed in AD 96-12-22	04/06/2020 Hrs: 4272.4 C:	Complied with by (a)(1-3) of AD 96-12 -22 and no evidence of oil leaking	No	D: Hrs: C:	AP3612277IA/Authorized Inspecto Cordaro Staub
/12/2000	journal fracture, which could result in total	04/14/2021 Hrs: 4272.4 C: –	Not applicable by date of overhaul	No	D: Hrs: C:	AP3612277IA/Authorized Inspect
9-19-01 30/1999	crankshaft cheek cracks, which could result	04/14/2021 Hrs: 4272.4 D:	not applicable by date of overhaul	No	D: Hrs: C:	AP3612277IA/Authorized Inspect
-17-11 C 9/1998	DUE TO CRACKING, WHICH COULD	04/14/2021 Hrs: 4272.4 C: —	Not applicable by date of overhaul	No	D; Hrs: C:	AP3612277IA/Authorized Inspec

ue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
98-01-08 E 12/23/1997	TO PREVENT EXHAUST ROLLER ROCKER ARM FAILURE, WHICH CAN RESULT IN BENT PUSH RODS, RUPTURED ROCKER ARM,CONTD.	04/14/2021 Hrs: 4272.4 C:		No	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Staub
97-21-02 10/27/1997	TO PREVENT EXTREME SIDE LOADING OF THE PISTON, AND CONSEQUENT FAILURE OF THE PISTON AND ENGINE	04/14/2021 Hrs: 4272.4 C:	Not applicable by part not installed	No	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Staub
96-12-22 07/31/1996	[Recurring] TO PREVENT LOSS OF ENGINE OIL CAUSED BY LOOSE OR SEPARATED OIL FILTER ADAPTERS, WHICH COULD RESULT IN ENGINE, CONTD.	04/14/2021 Hrs: 4272.4 C:	Complied with by (a)(1-3) of AD 96-12- 22 and no evidence of oil leaking		D: - WHEN OIL Hrs: - WHEN OIL C: - FILTER K REMOVED	AP3612277IA/Authorized Inspector Cordaro Staub
95-21-15 11/28/1995	TO PREVENT DETONATION DUE TO LOW OCTANE, WHICH CAN RESULT IT SEVERE ENGINE DAMAGE AND SUBSEQUENT FAILURE	04/14/2021 Hrs: 4272.4 C:	Not applicable by N number	No	D: Hrs: C: -	AP3612277IA/Authorized Inspector Cordaro Staub
93-08-17 18/23/1993	TO PREVENT AN ENGINE FAILURE	04/14/2021 Hrs: 4272.4 C:	Not applicable by SN	No	D; Hrs; C:	AP3612277IA/Authorized Inspecto Cordaro Staub
93-10-02 8/12/1993	TO PREVENT AN ENGINE FAILURE DUE TO A MISSING CYLINDER VALVE RETAINER KEY	04/14/2021 Hrs: 4272.4 C:	Not applicable by SN	No	D: Hrs: C:	AP3612277IA/Authorized Inspecto
91-19-03 9/29/1991	TO PREVENT OPERATION WITH COLLAPSED OIL FILTER ELEMENTS, WHICH CAN RESULT IN LOSS OF OIL PRESSURE, CONTD.	04/14/2021 Hrs: 4272.4 C:	not applicable by dat code	eNo	D: Hrs: C:	AP36122771A/Authorized Inspect
8-03-06 /15/1988	TO PREVENT POSSIBLE LOSS OF ENGINE OIL AND SUBSEQUENT ENGINE FAILURE	04/14/2021 Hrs: 4272.4 C:	not applicable by PN	No	D: Hrs: C:	AP3612277IA/Authorized Inspec
5-08-02 15/1985	TO PREVENT POSSIBLE VALVE STEM WEAR, OIL CONTAMINATION, VALVE SEIZURE, AND TOTAL LOSS OF ENGINE POWER		not applicable by PN	No	D: Hrs: C:	AP3612277IA/Authorized Inspec

Description TO PREVENT THE POSSESS	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
BILL FRESSI DE INITIO ATTROS	04/14/2021 Hrs: 4272.4 C:	not applicable by SN	No	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Staub
CARBURETOR BOWL DRAIN PLUG	04/14/2021 Hrs: 4272.4 C:	not applicable by SN	No	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Staub
EXHAUST VALVES	04/14/2021 Hrs: 4272.4 C:	Not applicable by part number installed	No	D: - Hrs: - C: -	AP3612277IA/Authorized Inspecto Cordaro Staub
TO PREVENT SERIOUS ENGINE DAMAGE IN FLIGHT	04/14/2021 Hrs: 4272.4 C:	not applicable by PN	No	D: Hrs: - C:	AP3612277IA/Authorized Inspect
	TO PREVENT THE POSSIBLE LOSS OF OIL PRESSURE INDICATION CARBURETOR BOWL DRAIN PLUG EXHAUST VALVES TO PREVENT SERIOUS ENGINE	TO PREVENT THE POSSIBLE LOSS OF O4/14/2021 Hrs: 4272.4 C: CARBURETOR BOWL DRAIN PLUG O4/14/2021 Hrs: 4272.4 C: EXHAUST VALVES O4/14/2021 Hrs: 4272.4 C: TO PREVENT SERIOUS ENGINE DAMAGE IN FLIGHT O4/14/2021 Hrs: 4272.4	Complied TO PREVENT THE POSSIBLE LOSS OF 04/14/2021 Hrs: 4272.4 C: CARBURETOR BOWL DRAIN PLUG EXHAUST VALVES O4/14/2021 Hrs: 4272.4 C: EXHAUST VALVES O4/14/2021 Not applicable by SN Hrs: 4272.4 C: TO PREVENT SERIOUS ENGINE DAMAGE IN FLIGHT O4/14/2021 Hrs: 4272.4 C: O4/14/2021 Not applicable by part number installed C: TO PREVENT SERIOUS ENGINE DAMAGE IN FLIGHT O4/14/2021 Hrs: 4272.4 C:	Complied Compliance Recur TO PREVENT THE POSSIBLE LOSS OF O1/14/2021 Hrs: 4272.4 C: CARBURETOR BOWL DRAIN PLUG 04/14/2021 Hrs: 4272.4 C: EXHAUST VALVES 04/14/2021 Not applicable by SN No EXHAUST VALVES 04/14/2021 Not applicable by SN No TO PREVENT SERIOUS ENGINE 04/14/2021 not applicable by PN No DAMAGE IN FLIGHT 04/14/2021 not applicable by PN No	TO PREVENT THE POSSIBLE LOSS OF O1/14/2021

Category: Propeller

Manufacturer: McCauley

Model: D3A34C401

S/N: 060578

P/N:

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2005-14-11 08/17/2005	To prevent blade failure that could result in separation of a propeller blade and loss of control of the airplane	04/14/2021 Hrs: 4272.4 C:	Not applicable, no record of Southern California prop service of Inglewood, ca, also by model number model number D3A34CA-01-C	No	D: - Hrs: - C: -	AP3612277IA/Authorized Inspector Cordaro Staub
82-27-02 R1 02/22/1983	TO PREVENT POSSIBLE PROPELLER BLADE SHANK FAILURE	04/14/2021 Hrs: 4272.4 C:	Not applicable by SN	No	D: Hrs: C:	AP3612277IA/Authorized Inspecto Cordaro Staub

Facility Cert No. / Type Description Method of TO PREVENT POSSIBLE BLADE PITCH 04/14/2021 Authorized By Signed Compliance Recur Next Due CONTROL FAILURES 607/1978 Not applicable by SN No Hrs: 4272.4 Hrs: -AP3612277IA/Authorized Inspector C: -C: --Cordaro Staub Category: Appliance

Manufacturer: Bendix Corporation

Model: S-20 SERIES

Position:

P/N: 10-79020-10

ATP Revision: 04/15/2021

S/N: 840157

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
07/19/2005	[Recurring] To prevent failure of the magneto impulse coupling assembly and possible engine failure	04/14/2021 Hrs: 4272.4 C:	NA by engine model		D: - Hrs: - C: -	AP3612277IA/Authorized Inspector
94-01-03 R2 06/28/1995	AND SUBSEQUENT ENGINE FAILURE	04/14/2021 Hrs: 4272.4 C:	Not applicable by SN		D: Hrs: C:	AP3612277IA/Authorized Inspecto Cordaro Staub
94-06-09 05/20/1994	TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON- GROUNDED MAGNETO	04/14/2021 Hrs: 4272.4 C:	Not applicable by date code	No	D: - Hrs: - C: -	AP3612277IA/Authorized Inspect
82-20-01 06/14/1983	COUPLING DUE TO IMPROPERLY	04/14/2021 Hrs: 4272.4 C:	Not applicable by SN	No	D: Hrs: C:	AP3612277IA/Authorized Inspec
74-26-09 2/24/1974	DEADY OF AUTOMOTIVE	04/14/2021 Hrs: 4272.4 C:	Not applicable by SN	No	D: Hrs; C:	AP3612277IA/Authorized Inspec
ategory: Appliance		Position:			ATP	Revision: 04/15/2021

Manufacturer: Bendix Corporation

Model: S-20 SERIES

P/N: 10-79020-10

S/N: 504803

CATP

04/19/2021 6:50 PM

7/8

12-06 19-2005	Description [Recurring] To prevent failure of the magneto impulse coupling assembly and possible engine failure.	Complied 04/14/2021	Method of Compliance Not applicable by	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
94-01-03 R2	TO PREVENT MAGNETO	Hrs: 4272.4 C:	engine model	No les	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Staub
06/28/1995	THE SOUSCEQUENT ENGINE FAILURE	04/14/2021 Hrs: 4272.4 C:	Not applicable by SN	No	D: Hrs: C:	AP3612277IA/Authorized Inspector Cordaro Staub
05/20/1994	TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON-GROUNDED MAGNETO	04/14/2021 Hrs: 4272.4 C:	Not applicable by date code	No	D: - Hrs: - C: -	AP3612277IA/Authorized Inspector
82-20-01 06/14/1983	HEAT TREATER (SOTTO AND ADDRESS OF THE ATTER ATTER (SOTTO ATTER AT	04/14/2021 Hrs: 4272.4 C:	Not applicable by SN	No	D: Hrs: C:	AP36122771A/Authorized Inspector
74-26-09 12/24/1974	S-20,-200,-1200 SERIES MAGNETOS	 Hrs: 4272.4 C:	Not applicable by SN	No	D: Hrs: - C:	AP3612277IA/Authorized Inspect

Category: Appliance

Position:

ATP Revision: 04/15/2021

Manufacturer: Airborne Manufacturing

P/N: 216CW

Model: 216CW

S/N: 51680

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
98-23-01 11/20/1998	PRIMARY DRY AIR PUMP CAUSED BY	04/14/2021 Hrs: 4272.4 C:	Not applicable by SN installed	No	D: Hrs: C:	AP3612277IA/Authorized Inspecto Cordaro Staub



Mangon Aircraft, Inc.

501 Sky Ranch Drive Petaluma, CA 94954-3908 Phone: (707) 765-1848 Fax: (707) 765-6610

Email: info@mangonaircraft.com

Date: 1/24/19, 9:11 AM

Doe.

CUSTOMER INFORMATION

Kopan Taber

232 Mocking Bird Circle Santa Rosa, CA 95409 (415) 577-4580 ktaber3@comcast.net Reg. No: 2376G 1959 Cessna 182B

Current A/C TT: 4233.9, Tach: 4233.9

A/C Serial: 51676 Terms: Due on Receipt

Customer Invoice Work Order: 15739

Work Order: 15739

Total Balance Due \$6,056.52 (See Below For Cost Detail)

Item: 1 - Customer Requested

Discrepancy

Annual Inspection due, FAR 91.409 (a) (1). Complete a 100 Hour inspection and service on aircraft using Cessna Service Manual D138-13, 03/24/03, Section II

Hours Subtota 8.500 977.50

Corrective Action

Aircraft was inspected for compliance of FAR 91.409 (a) (1). Inspected and Serviced using, a checklist with those items extracted from the manufactures inspection guidelines, Cessna Service Manual D138-13, 03/24/03, Section II, which meets the requirements, scope and detail of FAR 43.13, Appendix D.Customer stated that original log books lost and missing. Aircraft Total Time at this Inspection: 4233.9 hours. Engine Total Time: 2773.9 hours. Engine Time Since Major Overhaul by Eagle Engines of Redding California: 51 hours. Propeller Total Time Since New: 51 hours. Propeller Governor Time Since Major Overhaul: 51 hours. Engine, Propeller and Governor approximate overhaul and entered service date is June 3, 2007. Engine Model Serial Number installed on this aircraft is a Continental O-470-50 with serial number of 68680-8-L-4. Propeller Model Serial Number installed on this aircraft is a McCauley D3A4C4-01-C with a serial number of 060578.

Qty Unit	Part Number	Description	Serial Number	Unit Price	Subto
8 EA	10RX5/8THBSS	Screw Sheet Metal Truss Head # 10 7/16" head X 5/8L SS (P/O: 10970)		0.340	2.7
2 Ea	33-199-6	Gasket Fuel Strainer Glass C172, B-55 (P/O: 8800)		27.180	54.
3 Ea	4RX1/2THASS	Screw Sheet Metal Truss SS #4, 1/4" head X 1/2"L (P/O: 5767)		0.080	0.
3 Ea	8RX1/2SS-B	Screw Sheet Metal Truss hd #8 3/8" hd - 1/2" L (P/O: 10965)		0.160	0
2 Ea	8RX1/2THASS	Screw Sheet Metal Truss hd Type A #8 3/8 - 1/2" (P/O: 10965)		0.100	0
2 Ea	8RX1/2THBSS	Screw Sheet Metal Truss Head # 8 X 1/2 SS MS (P/O: 10904)		0.160	(
2 EA	A1789-8Z1D	Nut Tinnerman Folded # 8 Grip .325		0.270	(
7 EA	MS21207C8-8	Screw Sheet Metal Flat Head # 8 X 1/2 SS		0.300	
2 Ea	MS24665-151	Pin Cotter SS 1/16 X 1/2 L (Lot 200ea) (P/O: 10493)		0.200	
10 Ea	NAS391B4P	Washer Countersunk 82* #4 (P/O: 8651)		0.120	
3 Ea	NAS391B6P	Washer Countersunk 82* #6 (P/O: 8651)		0.100	
0 Ea	O4X10	Screw Trim Oval Head SS #4 5/8" (P/O: 10148)		0.080	
3 Ea	O6X12	Screw Trim Oval Head SS 6 X 3/4 (P/O: 8651)		0.080	

Labor: \$977.50 Parts: \$63.91 Shipping: \$0.00 Item Subtotal: \$1,041.41

	Item: 2 - Customer Requested	Hours	Subtotal
Discrepancy	fice on engine uning Cessna Service Manual D138-13.	8.000	920.000

Corrective Action

03/24/03, Section II. Engine Compartment.

100 Hour inspection and service completed on engine, using Cesana Service Manual D138-13, 03/24/03, 100 Hour inspection and service completed on engine, using Cesana Service Manual D138-13, 02/24/03, Section II Engine Compartment. Customer stated that original log books lost and missing. Aircraft Total Time at this inspection. 4233.9 hours. Engine Total Time: 2773.9 hours. Engine Time Since Major Overhaul by Eagle Engines of Redding California: 51 hours. Propeller Total Time Since New. 51 hours. Propeller Governor Time Since Major Overhaul: 51 hours. Engine, Propeller and Governor approximate overhaul and entered service date is June 3, 2007. This Engine Installed on Cessna 1828 N2376G, Serial Number 51675. Oil drained, oil filter removed, cut open & inspected, (no metal particles) and installed new CH48108-1 filter. Oil serviced to 12 Qts. with AeroShell 15W-50 oil. Complied with TCM SB03-3, Differential Pressure Test Reselling of A4380. Culinder 1,6080, 2,7080, 3,7580, A4280, 5,7080, 6,6880, TCM. Pressure Test, Baseline of, 44/80, Cylinder 1-60/80, 2-70/80, 3-75/80, 4-72/80, 5-70/80, 6-68/80, TCM MSB94-SA complied with for magnetos, timing and sync'd at 22 degrees BTC, cylinder #1. Primer and fuel system pressure checked, no leaks. Exhaust system pressure checked no leaks. See following corrected

					Unit Price	Subtotal
Qt	y Unit	Part Number	Description	Serial Number		12.100
11	Ea	07-00877	Gasket Spark Plug 18MM MS17840-1 (See notes) (P/O: 10703)		1.100	to the same of the
12	Ea	15W50QT	Oil Aeroshell 15W-50 Multi-Grade Qt.		11.650	139.800
			(Notes) (P/O: 10798)			7.050
1	Ea	16A48	Gasket Fuel Inlet Carb Con (P/O: 10922)		7.050	
1	Gal	66	Mineral Spirits Cleaning ASOLV55 (Gal) (P/O: 11798)		28.880	28.880
1	Ea	CH48108-1	Filter Oil Champion (P/O: 10905)		38.300	38.300

Labor: \$920.00	Parts: \$226,13	Shipping: \$0.00	Item Subtotal: \$1,146.13
			CONTRACTOR OF STREET

Item: 3 - Customer Requested

Hours Subtotal Propeller complete a 100 Hour inspection and service on propeller using Cessna Service Manual D138-13, 03/24/03, Section II 57.500 0.500

Corrective Action

Completed a 100 Hour inspection and service on propeller using Cessna Service Manual D138-13, 03/24/03, Section II, requirements. Blades cleaned, inspected, dressed out nicks and profiled leading edges as required. Track checked propeller. Blade backs painted. Customer stated that original log books lost and missing. Aircraft Total Time at this Inspection: 4233.9 hours. Engine Total Time: 2773.9 hours. Engine Time Since Major Overhaul by Eagle Engines of Redding California: 51 hours. Propeller Total Time Since New: 51 hours. Propeller Governor Time Since Major Overhaul: 51 hours. Engine, Propeller and Governor approximate overhaul and entered service date is June 3, 2007. This Propeller Installed on Cessna 182B N237G, Serial Number 51675.

Inspection Information:

Final Inspection: VICENTE SALINAS on 04/05/18

Qty Unit	Part Number	Description	Serial Number	Unit Price	Subtotal
3 Oz	A150	Paint - Flat Black (Tempo) 09-00982 (P/O: 8819)		1.030	3.100

Labor: \$57.50 Parts: \$3.10 Shipping: \$0.00 Item Subtotal: \$60.60

Item Subtotal: \$69.56

23.160

0.100

23.160

0.400

1 Ea

4 EA

Labor: \$46.00

BA8103-1

MS35333-38

			Item:	4 - Airworthy Air	rframe		200000000
ELT comply wi Sec. 12-22, a.	th FAR 91.207 (d), it thru f. Record next	nspection, test date a	battery expiration, a	and impact signal te	sting. See AC43-13-1E	Hours 0.500	Subtotal 57.500
Corrective Ac	ction complied with AC 43 original location with	3-1-1B, Sec	t. 12-22, a. thru f. E	LT meets FAR91.2	07 (d) requirements. I battery due		
-Function Check	mation: n: MICHAEL TILLM/ k: JAY KING on 02/2 s: JAY KING on 02/2	21/18	5/18				
Qty Un	it Part Number	r	Description		Serial Number	Unit Price	Subtotal
1 EA		7.		er (P/O: 9211) (Lot:		4.070	4.070
8 Ea	MN1300		Battery ELT Dry C Notes) (P/O: 111			2.560	20.480
Labor: \$57.5	0	Parts:	\$24,55	Shipping: \$	50.00	Item Subtotal: \$82.05	
			Item: 5	5 - Airworthy Air	rframe		
Discrepancy						Hours	Subtota
Battery requires f	Manufacture capacit	tance chec	k for annual certific	cation. Remove bat	ttery, clean, inspect,	0.500	57.50
	with Manufacture c	apacitance	check for annual	certification. Remo	ove battery, clean,		
rispect, charge for apacity tester rel	r two hours, perform fuses test due to lov acitance check due	n capacita w voltage o	nce test, recharge	to specifications, s	ervice and install.		
respect, charge for capacity tester ref co2972214. Capa respection informations: Virunction Check: J.	r two hours, perform fuses test due to lov acitance check due	n capacita v voltage of 12/2019. on 04/05/ 18	nce test, recharge of battery. Installed	to specifications, s	ervice and install.		
respect, charge for capacity tester ref co2972214. Capa respection informations: Virunction Check: J.	r two hours, perform fuses test due to lov acitance check due tion: //CENTE SALINAS AY KING on 03/14/	n capacita w voltage of 12/2019. on 04/05/ 18	nce test, recharge of battery. Installed	to specifications, s	ervice and install.	Unit Price	Subtot
inspect, charge for capacity tester rel 602972214. Capa ispection Informational Inspection: V function Check: J. ispector Code: J.	r two hours, perform fuses test due to lov acitance check due tion: //CENTE SALINAS AY KING on 03/14/ AY KING on 03/14/	n capacita w voltage of 12/2019. 6 on 04/05/ 18 18	nce test, recharge of battery. Installed	to specifications, s new fully serviced	ervice and install. Gill battery Sn.	Unit Price 5.980	Subtot 5.98
respect, charge for capacity tester rel co2972214. Capa respection Informational Inspection: Virunction Check: J. respector Code: J/ Qty Unit	r two hours, perform fuses test due to lov acitance check due tion: //CENTE SALINAS AY KING on 03/14/ Part Number	n capacita v voltage of 12/2019. 6 on 04/05/ 18 18	nce test, recharge of battery. Installed	to specifications, s new fully serviced (P/O: 11651) Hr (Req. 3 Qts	ervice and install. Gill battery Sn.		
inspect, charge for capacity tester rel co2972214. Capa ispection Informational Inspection: Virunction Check: J. aspector Code: J. Qty Unit 1 Ea	r two hours, perform fuses test due to lov acitance check due tion: //CENTE SALINAS /AY KING on 03/14/ Part Number DT3QT G35	n capacita v voltage of 12/2019. 6 on 04/05/ 18 18	nce test, recharge of battery. Installed 18 Description lattery Acid Quart attery Acid 35 Amp T1QT) (P/O: 1166 364581789)	to specifications, s new fully serviced (P/O: 11651) Hr (Req. 3 Qts	Serial Number G02972214	5.980	5.98
respect, charge for capacity tester rel co2972214. Capa isspection Informatinal Inspection: Virunction Check: J. Capacity Unit 1 Ea 1 Ea	r two hours, perform fuses test due to lov acitance check due tion: //CENTE SALINAS /AY KING on 03/14/ Part Number DT3QT G35	capacita v voltage of 12/2019. con 04/05/ 18 18	nce test, recharge of battery. Installed 18 Description leastery Acid Quart attery Acid Quart attery Gill 35 Amp T1QT) (P/O: 1166 364581789) 348.70	to specifications, s new fully serviced (P/O: 11651) Hr (Req. 3 Qts 60) (Lot:	Serial Number G02972214	5.980 342.720	5.9
respect, charge for capacity tester rel co2972214. Capa isspection Informatinal Inspection: Virunction Check: J. Capacity Unit 1 Ea 1 Ea	r two hours, perform fuses test due to lov acitance check due tion: //CENTE SALINAS /AY KING on 03/14/ Part Number DT3QT G35	capacita v voltage of 12/2019. con 04/05/ 18 18	nce test, recharge of battery. Installed 18 Description leastery Acid Quart attery Acid Quart attery Gill 35 Amp T1QT) (P/O: 1166-364581789) 348.70	to specifications, s new fully serviced (P/O: 11651) thr (Req. 3 Qts 60) (Lot: Shipping: \$	Serial Number G02972214	5.980 342.720	5.94 342.7
respect, charge for capacity tester rel for 2972214. Capa isspection Informational Inspection: Virunction Check: Junction Chec	r two hours, perform fuses test due to low acitance check due tion: //CENTE SALINAS/AY KING on 03/14/ Part Number DT3QT G35	n capacita w voltage of 12/2019. s on 04/05/ 18 18 E B D 7.	nce test, recharge of battery. Installed 18 Description lattery Acid Quart attery Acid Quart T1QT) (P/O: 1166-364581789) 348.70 Item: 6	to specifications, s new fully serviced (P/O: 11651) Hr (Req. 3 Qts 60) (Lot: Shipping: \$ - Airworthy Air	Serial Number G02972214	5.980 342.720 Item Subtotal: \$406.20	5.98
respect, charge for capacity tester rel for 2972214. Capa ispection Informatinal Inspection: Virunction Check: J. Capacity Unit 1 Ea 1 Ea abor: \$57.50	r two hours, perform fuses test due to low acitance check due tion: //CENTE SALINAS/AY KING on 03/14/ Part Number DT3QT G35	n capacita w voltage of 12/2019. s on 04/05/ 18 18 E B D 7.	nce test, recharge of battery. Installed 18 Description lattery Acid Quart attery Acid Quart T1QT) (P/O: 1166-364581789) 348.70 Item: 6	to specifications, s new fully serviced (P/O: 11651) Hr (Req. 3 Qts 60) (Lot: Shipping: \$ - Airworthy Air	Serial Number G02972214 0.00	5.980 342.720 Item Subtotal: \$406.20	5.98 342.77 Subto
respect, charge for capacity tester reliable for apacity tester reliable for apacity tester reliable for apacity tester reliable for apacity for apaci	r two hours, perform fuses test due to low acitance check due tion: //CENTE SALINAS/AY KING on 03/14/ Part Number DT3QT G35	r continue	nce test, recharge of battery. Installed 18 Description Lattery Acid Quart attery Acid Quart attery Gill 35 Amp T1QT) (P/O: 1166-364581789) 1848.70 Item: 6	to specifications, s new fully serviced (P/O: 11651) Hr (Req. 3 Qts 60) (Lot: Shipping: \$ - Airworthy Air quirements instruct	Serial Number G02972214 0.00 frame ctions as outlined in	5.980 342.720 Item Subtotal: \$406.20	5.98 342.77 Subto

Shipping: \$0.00

Filter Engine Air Brackett C182Q, C185F (P/O: 10905)

Washer Lock Internal Star # 8 (AN936A-8) (P/O: 10810)

Parts: \$23.56

Mangon Aircraft, Inc. - Phone: (707) 765-1848, Fax: (707) 765-6610 - 1/24/19, 9:11 AM 501 Sky Ranch Drive, Petaluma, CA 94954-3908

WIO: 15739

Item: 7 - Airworthy Airframe

Vacuum central and relief filter due for replacement I/A/W Airborne Service Letter 59A.

Hours 0.500 Subtotal 57,500

Corrective Action

Labor: \$57.50

Discrepancy

Vacuum central filter replaced VAW Airborne Service Letter 59A.

Qty Unit Part Number AAD9-18-1 1 Ea

Description Filter Vacuum Suction (10-01354)

Unit Price 33.000

Subtotal 33,000

100Hrs. OR Annually (P/O: 10762)

Parts: \$33.00

Shipping: \$0.00

Serial Number

Item Subtotal: \$90.50

Item: 8 - Airworthy Airframe

Discrepancy

A.D. 76-07-12 R1 Eff. date 08/30/1977, Bendix Ignition Switch grounding check inspection is due (every 100 hours). Insert next TT due time.

Hours N/C - 0.200 Subtotal 0.000

Corrective Action

A.D. 76-07-12 R1 Eff. date 08/30/1977. Bendix Ignition Switch, grounding complied with Para. 1., (a), (b), and (c), grounding function was satisfactory. Due again in 100 hrs. at 4333.9 hrs.

Inspection Information

Final Inspection: MICHAEL TILLMAN on 03/23/18

Labor: \$0.00

Parts: \$0.00

Shipping: \$0.00

Item Subtotal: \$0.00

Item: 9 - Airworthy Airframe

Discrepancy

A.D. 84-10-01 R1 eff. date 07/05/1988 Cessna bladder fuel cells inspection is due (every 12 months).

Hours 0.500 Subtotal 57.500

A.D. 84-10-01 R1 Eff. date 07/05/1988 Cessna bladder fuel cells inspected. Tanks drained no wrinkles found. Cap seals cleaned. Tanks serviced with customer removed fuel. No leaks noted. Due again 12/2019.

Inspection Information:

Labor: \$57.50

-Final Inspection: MICHAEL TILLMAN on 04/05/18

Parts: \$0.00

Shipping: \$0.00

Item Subtotal: \$57.50

Item: 10 - Airworthy Airframe

Discrepancy A.D. 96-12-22 Eff. date 07/31/96 Cessna/Continental Oil filter adapter torque inspection due (every oil filter change).

N/C - 0.200

Hours

Subtotal 0.000

Corrective Action

A.D. 96-12-22 eff. date 07/31/1996 Cessna/Continental Oil filter adapter torque inspection complied with, Torque putty in place, not loose and no defects found. Due again at next oil filter change.

Inspection Information:

Final Inspection: VICENTE SALINAS on 04/05/18

Labor: \$0.00

Parts: \$0.00

Shipping: \$0.00

Item: 11 - Airworthy Airframe

Item Subtotal: \$0.00

Discrepancy

A.D. 2011-10-09 Eff. dated 06/17/2011, Admt. 39-16690. Cessna Seats and Seat Tracks inspection is due. (Repetitive: 12 months or 100 hours). "(Estimated time is for Inspection Only make separate entry for repairs)"

Hours 0.750 Subtotal 86.250

Corrective Action

A.D. 2011-10-09 Eff. date 06/17/2011, Admt. 39-16690 Cessna Seats and Seat Tracks Complied with Paragraph's (e) through (g) Inspection due again 12/2019 or at 4333.9 hrs.

Inspection Information:

Final Inspection: VICENTE SALINAS on 04/05/18 Function Check: BRYAN MARSH on 02/22/18

Labor: \$86.25 Parts: \$0.00 Shipping: \$0.00 Item Subtotal: \$86.25 Item: 12 - Airworthy Engine Discrepancy Subtotal A.D. 2016-16-12, effective date 9/15/2016, Continental Motors, Inc. (CMI) model -520 and -550 Hours 0.000 N/C - 0.200

reciprocating engines when modified by supplemental type certificate (STC), with Continental Motors Inc., San Antonio (formerly Airmotive Engineering Corp.), replacement parts manufacturer approval (PMA) cylinder assemblies, marketed by Engine Components International Division (hereinafter referred to as ECi), part number (P/N) AEC 631397, with ECi Class 71 or Class 76, serial number (S/N) 1 through S/N

Corrective Action

A.D. 2016-16-12, effective date 9/15/2016, Continental Motors, Inc. (CMI) model -520 and -550 reciprocating engines when modified by supplemental type certificate (STC), with Continental Motors Inc., San Antonio (formerly Airmotive Engineering Corp.), replacement parts manufacturer approval (PMA) cylinder assemblies, marketed by Engine Components International Division (hereinafter referred to as ECi), part number (P/N) AEC 631397, with ECi Class 71 or Class 76, serial number (S/N) 1 through S/N 61176, installed. Not applicable to part number installed. No further action required.

Inspection Information: -Final Inspection: MICHAEL TILLMAN on 05/18/18

Labor: \$0.00 Item Subtotal: \$0.00 Parts: \$0.00 Shipping: \$0.00

Item: 13 - Airworthy Engine

Subtotal Discrepancy Hours 0.000 N/C - 0.340 Engine idle speed low at 500 RPM

Corrective Action

Adjusted idle speed screw in 1/2 turn to increase idle speed to 650 RPM. Ops check good.

Labor: \$0.00 Item Subtotal: \$0.00 Parts: \$0.00 Shipping: \$0.00

Item: 14 - Airworthy Airframe

Discrepancy Hours Subtotal Attitude indicator never did erect itself. 1.590 182.850

Corrective Action

Attitude indicator removed and replaced. Ops check good.

Inspection Information:

-Final Inspection: MICHAEL TILLMAN on 04/05/18

Part Number Qty Unit Description Serial Number **Unit Price** Subtotal 5000B-12 1 Fa Indicator Horizon Artifical, OH/Exch, A2729J 633.330 633.330 3850096 **Notes (P/O: 11774) (Ship: 50.000)

Labor: \$182.85 Parts: \$633.33 Shipping: \$50.00 Item Subtotal: \$866.18

Item: 15 - Airworthy Airframe

Discrepancy Hours Subtotal Wing left, fuel vent hose not attached. 1.450 166.750

Corrective Action

Wing left, fuel vent hose removed and replaced.

Inspection Information:

 Final Inspection: MICHAEL TILLMAN on 04/05/18 Function Check: BRYAN MARSH on 02/21/18

> Qty Unit **Part Number** Description Serial Number Unit Price Subtotal 1 Ft 193-8 Hose Low Press. Inst 1/2 ID 750 PSI 9.850 9.850 (306-8) (MIL-H-5593-8) Ft.

Labor: \$166.75 Parts: \$9.85 Shipping: \$0.00 Item Subtotal: \$176.60

Item: 16 - Airworthy Airframe Discrepancy Flap track, left wing, inboard left side, support bracket broken. Subtotal Hours 285.200 2.480 **Corrective Action** Flap track, left wing, inboard left side support bracket removed and replaced. Flap operation checked good, Inspection Information -Function Check: VICENTE SALINAS on 03/14/18 Qty Unit Part Number Subtotal Description Unit Price Serial Number 1 Ea 0523231-24 Bracket - Flap Track (P/O: 11736) (Ship: 14.990) 180,830 180.830 2 Ea MS20470AD3-3 0.100 Rivet Aluminum Universal 3/32 D X 0.050 2 Ea MS21044N06 0.500 1.000 Nut Self Locking Nylon 6-32 250 deg. (P/O: 10922) (Lot: 26596) Labor: \$285.20 Parts: \$181.93 **Shipping: \$14.99** Item Subtotal: \$482.12 Item: 17 - Airworthy Airframe Subtotal Hours Discrepancy 0.380 43,700 Fuel tank sender left ground not attached Corrective Action Secured ground on left fuel sender. Inspection Information:

Labor: \$43.70 Parts: \$0.00 Shipping: \$0.00 Item Subtotal: \$43.70

Item: 18 - Airworthy Airframe

Discrepancy
Air vent scat worm clamp in left wing root not secured.

Hours
Subtotal
N/C - 0.200
0.000

Corrective Action

Positioned and secured air vent clamp for scat tube in Lt. wing root.

Inspection Information:

Function Check: JAY KING on 02/21/18 Inspector Code: JAY KING on 02/21/18

Function Check: JAY KING on 02/21/18 Inspector Code: JAY KING on 02/21/18

Labor: \$0.00 Parts: \$0.00 Shipping: \$0.00 Item Subtotal: \$0.00

Item: 19 - Airworthy Airframe

Discrepancy
Lt. and Rt. flap top inboard edge has hole worn in from truss head screw used to secure aft wing root

66.700

Corrective Action

Lt. and Rt. flap top inboard edge installed flap chafe buttons.

section. Recommend dimpling wing root cover and using flat head sheet metal screw.

Inspection Information:

·Function Check: BRYAN MARSH on 02/21/18

Qty UnitPart NumberDescriptionSerial NumberUnit PriceSubtotal2 Ea\$1093-1Button Chafe Flap (Ces) (P/O: 10356)0.9901.980

Labor: \$66.70 Parts: \$1.98 Shipping: \$0.00 Item Subtotal: \$68.68

Item: 20 - Customer Rejected Subtotal Discrepancy 0.000 Doors pilot and co-pilot hinge and/or hinge pin worn. Rejected **Corrective Action** ...Customer declined any corrective action at this time Item Subtotal: \$0.00 Labor: \$0.00 Parts: \$0.00 Shipping: \$0.00 Item: 21 - Airworthy Airframe Subtotal Hours

Discrepancy Main brake disc Lt. below min. at .203 min is .205, right disc at .220

174.800 1.520

Removed Lt. main wheel assembly. Removed and installed new brake disc on wheel assembly and torqued wheel halves to 150 in. lbs. Installed new brake linings on Lt. brake caliper. Installed main wheel assembly and saftied. Tires serviced to 35 psi. Brake disc and linings conditioned IAW Cleveland maintenance procedures.

Inspection Information:

-Final Inspection: VICENTE SALINAS on 04/05/18 -Function Check: JUSTIN CASAUS on 02/22/18

iction Check: JU	JSTIN CASAUS on U	2/22/18		Unit Price	Subtotal
Qty Unit	Part Number	Description	Serial Number	0.780	3.120
4 Ea	105-00200	Rivet Brake Steel Cle (See Alt. RA105-00200) (P/O: 10818)		0.220	0.220
1 EA	MS24665-362	Pin Cotter 1/8 X 2 1/2" (Wheel) (P/O: 11016)		126.320	126.320
1 Ea 2 Ea	RA164-02601 RA66-10500	Disc Brake (06-00623) Lining Brake Rapco (06-16955) (P/O:		10.830	21.670
Z La	10 100 1001	11120)			

11130) Item Subtotal: \$326.13 Shipping: \$0.00 Parts: \$151.33 Labor: \$174.80

Item: 22 - Airworthy Airframe Subtotal Hours 1,076.400 9.360 Discrepancy Log Book re-construction

Corrective Action

Log Books and aircraft paperwork obtain from all sources possible and put into aircraft records.

og Books	and all c	rait paper none out		Serial Number	Unit Price	Subtotal
Qty	Unit	Part Number	Description	Gerjai italiibe.	40.370	40.370
1	Ea	13-12324	Manintenance Log Binder (Notes) (Ship: 15.720)		17.720	17.720
1	Ea	13-12325	Log Book Engine, ATP (P/O: 12335)		17.720	17.720
1	Ea	13-12327	Log Book Airframe, ATP (P/O: 12335)		17.720	17.720
1	Ea	13-12328	Log Book Propeller, ATP (P/O: 12335)	San Color of Paris	16.670	16.670
1	Ea	FAA CD	Aircraft Records FAA		53.940	53.940
1	Ea	P171-13	Owners Manual 1959 Skylane 182			THE RESERVE OF THE PERSON NAMED IN

Item Subtotal: \$1,256.26 Shipping: \$15.72 Parts: \$164.14 Labor: \$1,076.40

Item: 23 - Airworthy Airframe

Subtotal Hours Airworthiness Directive Records and listing for all components is not complete I/A/W FAR 91.417 (a)(2)(v). Discrepancy 647.450 5.630

Corrective Action

Airworthiness Directive records for all components brought up to date I/A/W FAR 91.417 (a)(2)(v) and made part of this aircraft's records.

Item Subtotal: \$647.45 Shipping: \$0.00 Parts: \$0.00 Labor: \$647.45

Discrepancy Seat track Pilot outboard pin stop holes wom below m	Item: 24 - Airworthy Airframe	Hours	Subtotal 476,100
Corrective Action	***	4,140	## F5 7535
Seat track Pilot outboard removed and replaced.			

Inspection Information:

·Final Inspection: VICENTE SALINAS on 04/05/18

-Function Check: JAY KING on 03/15/18 Inspector Code: JAY KING on 03/15/18

Qty	Unit	Part Number	Description	Serial Number	Unit Price	Subtotal
1	Ea	MC0511293-3	Seat Rail Pilot LH - Cessna 182 (P/O:		312.800	312.800
20	EA	MS20470AD4-5	11750) (Lot: 45370) (Ship: 25.000) Rivet Aluminum Universal 1/8 D X		0.050	1.000
10	EA	MS20470AD5-10	5/16 (Lot: Heat 2002) Rivet Aluminum Universal 5/32 D X		0.050	0.500
1	Ea	SRTEM1	5/8 (P/O: 8389) Template Seat Rail Narrow (P/O:		9.630	9.630
			10490) (Lot: 44986)			

Labor: \$476.10	Parts: \$323.93	Shipping: \$25.00	Item Subtotal: \$625.55
	Item	25 - Airworthy Airframe	

Item: 25 - Airworthy Airframe Subtotal Hours 606.050 Discrepancy 5.270 Fuel selector valve assembly under floor leaking severely. **See Notes**

Corrective Action

Fuel selector valve assembly under floor leaking severely. Removed fuel selector valve and sent out for overhaul to Aircraft Accessories Int. P/O # 11773. Reinstalled Fuel selector valve. Leak tested system, no defects noticed.

Aircraft Accessories Int. (Fuel, actuators) - Invoice #5096, Labor: 598.480, Parts: 0.000 (Fuel Selector Valve - Please Overhaul. Advise as needed.)

Inspection Information:

Final Inspection: VICENTE SALINAS on 04/05/18

Final Inspection: VICENTE SALINAS on 04/05/18

-Function Check, Dr. I All Min	NOT FOLL COLEGE			1000000
	Parts: \$0.00	Shipping: \$50.00	Item Subtotal: \$1,254.53	
Labor: \$1,204.53			SCHOOL SC	

Item: 26 - Airworthy Airframe Subtotal Hours Engine mount installed incorrectly, pilot and co-pilot bottom engine pads and co-pilot bushing torn off. 3.670 422.050

Engine mount removed, inspected, re-installed in original location with new pads, bushings and bolts. Bolts torqued to 190 inch lbs.

Inspection Information:
-Function Check: VICENTE SALINAS on 03/22/18

ICHOIL CHECK. A	CEITIE OF IEIT IS SI	Mariana Ma			
Qty Unit	Part Number	Description	Serial Number	Unit Price	Subtotal
4 Ea	0550154-2	Pad Engine Mount Firewall - Neoprene (P/O: 11736) (Lot: 11222021) (Ship: 15.000)		8.630	34.530
4 Ea	0550155	Bushing Cessna Engine Mount to Firewall (P/O: 11736) (Lot: 05152025)		15.480	61.930
2 Ea	AN6-25A	Bolt Hex Undrilled		4.610	9.220
2 Ea	AN6-34A	Bolt 3/8-24 X 3 37/64 L Grip 2 15/16 (P/O: 11839) (Lot: 22730)		2.880	5.760
4 Ea	MS21045-6	Nut Self Locking Steel 3/8-24 450 deg.		1.100	4.400

Item Subtotal: \$552.89 Shipping: \$15.00 Labor: \$422.05 Parts: \$115.84

Aircraft, Inc. - Phone: (707) 765-1848, Fax: (707) 765-6610 - 1/24/19, 9:11 AM of sky Ranch Drive, Petaluma, CA 94954-3908

W/O: 15739

Discrepancy

Battery box corroded.

Item: 27 - Airworthy Airframe

Hours 1.020 Subtotal 117.300

Corrective Action

Battery box corrosion neutralized with Baking soda and water. Cleaned box and painted with Acid Proof

Inspection Information:

Function Check: BRYAN MARSH on 02/22/18

Qty Unit **Part Number** Description **Unit Price** Subtotal Serial Number 3 Oz 345QT 3.240 Paint Acid Proof Black (Oz) (Haz Mat) 1.080

Labor: \$117.30 Item Subtotal: \$120.54 Parts: \$3.24 Shipping: \$0.00

Item: 28 - Airworthy Airframe

Discrepancy Weight & Balance Report is needed.

Corrective Action

Hours

7.450

Subtotal 856.750

Obtain replacement 1959 Cessna Skylane 182 Owner's Manual and Cessna Weight and Balance Report Form. Aircraft placed on scales and weighted. New Weight and Balance Report put into aircraft records

Item Subtotal: \$856.75 Labor: \$856.75 Parts: \$0.00 Shipping: \$0.00

Item: 29 - Airworthy Airframe

Subtotal Hours Discrepancy

FAR 91.411 (IFR)Altimeter, Static System & Altitude Reporting and FAR 91.411 & 91.413 Transponder Checks are due.

N/C - 0.000

0.000

Corrective Action FAR IFR Certification: An encoder, transponder, altimeter integration test was performed as required by FAR 91.411 (a)(3) and 91.413 (b) and found to comply with FAR Part 43 APP E and APP F. See attached certification results (CRS QJAR856X) Next FAR IFR certification is due Nov 2020.

Outside Repair:

Precision Static Testing (CRS QJAR856X), Labor: 300.000, Parts: 0.000

(FAR 91.411 IFR and 91.413 Due. Provide certification for Log Book and Work Order.)

Parts: \$0.00 Shipping: \$0.00 Item Subtotal: \$300.00 Labor: \$300.00

Item: 30 - Airworthy Airframe

Hours Subtotal Discrepancy

Complimentary aircraft belly de-greased, exterior wash, interior vacuumed and windows cleaned with Annual Inspection from Mangon Aircraft, Inc.

N/C - 10.730

0.000

Complimentary aircraft belly de-greased, exterior wash, interior vacuumed and windows cleaned with Annual Inspection from Mangon Aircraft, Inc.

Labor: \$0.00 Parts: \$0.00 Shipping: \$0.00 Item Subtotal: \$0.00

Please see the next page for the totals for this invoice

Aircraft, Inc. - Phone: (707) 765-1848, Fax: (707) 765-6610 - 1/24/19, 9:11 AM

W/O: 15739

Additional Cha Pilot Services 0.000	Tire Disposal	EPA Charge 25,000	Oil Analysis	Shop Supplies	Fuel	16-	0.14.44
Labor Summa		20.000	0.000	14.880	0.000	Misc. 0.000	Subtotal \$39.88
Shop Labor: \$		Outside Donni				Total Labor:	\$8,337.83
axed Items:		Outside Repair:	\$898.48			Total Parts:	\$2,308.52
Dil Analysis, Sho	Parts OSP D	arts, Shipping In, S				Total Shipping:	\$170.71
Four Deve 1	- 413, 001(P)	arts, Shipping In, S	hop Supplies			Additional Charges:	\$39.88
fax Breakdown: Oil Analysis: 0.00	(Before Calculat	ted)				Tax (8.250 %):	\$199.58
Shop Parts: 2,30	8.520					Amount Due:	\$11,056.52
OSR Parts: 0.00 Shipping In: 95.7	10					Deposit:	\$5,000.00
Shop Supplies:	14.880					Balance Due:	\$6,056.52

Important Information

INVOICES ARE DUE AND PAYABLE WHEN RECEIVED. A FINANCE CHARGE OF 1.5% PER MONTH WILL BE ASSESSED ON ALL INVOICES AFTER 30 DAYS PAST DUE.

Mangon Aircraft, Inc. 30 day warranty will cover only repairs or replacements made due to defects in material or workmanship. Mangon, Inc. will replace, at no cost to the customer, any parts that they furnish and install which are found to be defective. This warranty shall not apply to any goods or services which, in the judgment of Mangon Aircraft, Inc., have been altered so as to adversely affect performance or reliability, or which have been subjected to misuse, negligence or accident. This warranty does not extend to Customer-furnished parts or equipment. Mangon Aircraft, Inc. will not cover any time loss, inconvenience, loss of aircraft time, or other consequential damage. I have examined the work list above and agree that all work was done in a Professional Manner and to my satisfaction.

Kopan Taber or Authorized Representative Date



Piper

Service List - W/O: 15739 - Kopan Taber (2376G)

tem	Discrepancy	Corrective Action	Category	Sign-off	Created	Item Status
3	Battery requires Manufacture capacitance check for annual certification. Remove battery, clean, inspect, charge for two hours, perform capacity test, recharge to specifications and install.	Battery complied with Manufacture capacitance check for annual certification. Remove battery, clean, inspect, charge for two hours, perform capacitance test, recharge to specifications, service and install. Capacity tester refuses test due to low voltage of battery. Battery needs replaced. Due again ??/??/??.	Airworthy Airframe		МТ	Open
3	Engine air filter due for replacement per continued airworthiness requirements instructions as outlined in Brackett Air filter Document I-194.	Engine air filter BA8103-1, replaced I/A/W continued airworthiness requirements instructions per Brackett Air filter Document I-194.	Airworthy Airframe		МТ	Insp. Required
7	Vacuum central and relief filter due for replacement I/A/W Airborne Service Letter 59A.	Vacuum filter AAD9-18-1, replaced I/A/W Airborne Service Letter 59A.	Airworthy Airframe		МТ	Insp. Required
8	A.D. 76-07-12 R1 Eff. date 08/30/1977, Bendix Ignition Switch grounding check	A.D. 76-07-12 R1 Eff. date 08/30/1977. Bendix Ignition Switch, grounding complied	Airworthy Airframe		MT	Open
	inspection is due (every 100 hours). Insert next TT due time.	with Para. 1., (a), (b), and (c), grounding function was satisfactory. Due again in 100 hrs. at ????? hrs.				
9	A.D. 84-10-01 R1 eff. date 07/05/1988 Cessna bladder fuel cells inspection is due (every 12 months).	A.D. 84-10-01 R1 Eff. date 07/05/1988 Cessna bladder fuel cells inspected. Tanks drained no wrinkles found. Cap seals cleaned. Tanks serviced no leaks. Due again 04/08	Airworthy Airframe		МТ	Open
10	A.D. 96-12-22 Eff. date 07/31/96 Cessna/Continental Oil filter adapter torque inspection due (every oil filter change).	A.D. 96-12-22 eff. date 07/31/1996 Cessna/Continental Oil filter adapter torque inspection complied with, Torque putty in place, not loose and no defects found. Due again at next oil filter change.	Airworthy Airframe		МТ	Open
11	A.D. 2011-10-09 Eff. dated 06/17/2011, Admt. 39-16690. Cessna Seats and Seat Tracks inspection is due. (Repetitive: 12 months or 100 hours). "(Estimated time is for Inspection Only make separate entry for repairs)"	A.D. 2011-10-09 Eff. date 06/17/2011, Admt. 39-16690 Cessna Seats and Seat Tracks Complied with Paragraph's (e) through (g) Inspection due again, Date of or at	Airworthy Airframe		МТ	Open
12	Engine idle speed low at 500 RPM.		Airworthy Airframe		MT	Oper
3	Attitude indicator never did erect itself.		Airworthy Airframe		MT	Oper
4	Vacuum suction at 5" with full RPM.		Airworthy Airframe		MT	Oper
5	Stall warning inop. (click in speaker)		Airworthy Airframe	13 11 11	MT	Oper



Piper

Service List - W/O: 15739 - Kopan Taber (2376G)

		Corrective Action	Category	Sign-off	Created	Item Status
Item		Confective Action	Airworthy Airframe		vs	Open
16	Wing left, fuel vent hose not attached.		All Worting Airrianne			Open
17	Flap track, left wing, inboard left side, support bracket broken.		Airworthy Airframe		VS	
18	Fuel tank sender left ground not attached		Airworthy Airframe		VS	Open
19	Air vent scat worm clamp in left wing root not secured.		Airworthy Airframe		vs	Open
20	Lt. and Rt. flap top inboard edge has hole worn in from truss	Total Control	Airworthy Airframe		JGC	Open
	head screw used to secure aft wing root section. Recommend dimpling wing root cover and using flat head sheet metal screw.		min and of the			3.9
21	Doors pilot and co-pilot hinge and/or hinge pin worn.	164-02601	Airworthy Airframe		VS	Open
22	Main brake disc Lt. below min. at .203 min is .205, right disc at .220		Airworthy Airframe		VS	Open
23	AD search		Airworthy Airframe		DM	Open
24	Sea track Pilot outboard pin stop holes worn below min.	Service of the servic	Airworthy Airframe		VS	Oper
25	Fuel selector valve assembly under floor leaking severely.		Airworthy Airframe		VS	Oper
26	Engine mount installed incorrectly, pilot and co-pilot bottom engine pads and co-pilot bushing torn off.		Airworthy Airframe		VS	Ope



Piper

Service List - W/O: 15739 - Kopan Taber (2376G)

Servey Co		Compather Action	Category	Sign-off	Created	Item Status
tem	Discrepancy	Corrective Action	Customer Requested		MT	Open
	Annual Inspection due, FAR 91.409 (a) (1). Complete a 100 Hour inspection and service on aircraft using Cessna Service Manual D138-13, 03/24/03, Section II	Aircraft was inspected for compliance of FAR 91.409 (a) (1). Inspected and Serviced using, a checklist with those items extracted from the manufactures inspection guidelines, Cessna Service Manual D138-13, 03/24/03, Section II, which meets the requirements, scope and detail of FAR 43.13, Appendix D. See following corrected discrepancies				
2	Engine complete a 100 Hour inspection and service on engine using Cessna Service Manual D138-13, 03/24/03, Section II. Engine	100 Hour inspection and service completed on engine, using Cessna Service Manual D138-13, 03/24/03, Section II Engine Compartment. Oil	Customer Requested		МТ	Open
	Compartment.	drained, oil filter removed, cut open & inspected, (no metal particles) and installed new filter. Oil serviced to 12 Qts. with Shell 15W-50 oil. Complied with TCM SB03-3, Differential Pressure Test, Baseline of, /80. Cylinder 1- /80, 2-/80, 3-/80, 4-/80, 5-/80, 6-/80. TCM				
		MSB94-8A complied with for magnetos, timing and sync'd at 22 degrees BTC, cylinder #1. Primer and fuel system pressure checked, no leaks. Exhaust system pressure checked no leaks. See following corrected discrepancies.				Open
3	Propeller complete a 100 Hour inspection and service on propeller using Cessna Service Manual D138-13, 03/24/03, Section II	Completed a 100 Hour inspection and service on propeller using Cessna Service Manual D138-13, 03/24/03, Section II, requirements Blades cleaned, inspected, dressed out nicks and profiled leading edges as required. Track checked propeller. Blade backs painted,	Customer Requested		МТ	The state of the s
4	**Batters due** ELT comply with FAR 91.207 (d), inspection, battery expiration, and impact signal testing. See AC43-13-1B, Sec. 12-22, a. thru f. Record next test date and battery replacement due.	ELT removed, complied with AC 43-1-1B, Sec. 12-22, a. thru f. ELT meets FAR91.207 (d) requirements. ELT installed in original location with switch armed. Next inspection due ????? and battery due replacement ?????.	Airworthy Airframe		МТ	Oper



Mangon Aircraft, Inc. 501 Skyranch Rd Petatluma CA 94954-3908 USA

Phone: (707) 765-1848

FAA Airworthiness Directives Compliance Record

Company:

Category: Airframe

Astirame

Manufacturer: Cessna Aircraft Company

Model: 182B

Position:

P/N:

S/N: 51676

Aircraft Registration No: 2376G

ATP Revision: 05/11/2019

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2011-10-09 06/17/2011	[Recurring] To prevent seat slippage or the seat roller housing from departing the seat rail, which may consequently cause, contd.	05/14/2019 Hrs: 4233.9 C:	C/W paragraph's (e) through (g)	Yes	D: 05/27/2020 Hrs: 4333.9 C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
2008-26-10 C 01/05/2009	To prevent erroneous indications from the altimeter, airspeed, and vertical speed, contd.	05/18/2018 Hrs: 4233.9 C:	N/A to part number installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
008-10-02 5/12/2008	To prevent erroneous indications from the altimeter, airspeed, and vertical speed, contd.	05/18/2018 Hrs: 4233.9 C:	N/A this part not installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
004-19-01 C 1/01/2004	shoulder harness, which could result in	05/18/2018 Hrs: 4233.9 C:	N/A this part not installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
78-16-04 7/21/1998	TO PREVENT WING FAILURE DURING FLIGHT CAUSED BY THE ABSENCE OF AN ANGLE STIFFENER, WHICH COULD CAUSE LOSS OF, CONTD.		N/A STC is not installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
7-01-13 /03/1997	HYDRAULIC SYSTEMS FAILURE	05/18/2018 Hrs: 4233.9 C:	N/A to part number installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	D	Nort Dur	Facility Cert No. / Type Authorized By
96-12-22 07/31/1996	[Recurring] TO PREVENT LOSS OF ENGINE OIL CAUSED BY LOOSE OR SEPARATED OIL FILTER ADAPTERS, WHICH COULD RESULT IN ENGINE, CONTD.	05/14/2019 Hrs: 4233.9 C:	C/W by visual inspection. Torque seal in place. **Due Every Oil Filter Removal**	Yes	Next Due D: Hrs: 4283.9 C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
87-20-03 R2 09/24/1990	[Recurring] Superseded by 2011-10-09	05/18/2018 Hrs: 4233.9 C:	Superseded by 2011- 10-09	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
84-10-01 R1 07/05/1988	[Recurring] TO PREVENT POWER LOSS OR ENGINE STOPPAGE DUE TO WATE CONTAMINATION OF FUEL SYSTEM	11/16/2018 Hrs: 4233.9 C:	P/C/W Cessna Service Kit SK182- 85 is installed. No further action is required.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
79-10-14 R1 05/30/1988		05/18/2018 Hrs: 4233.9 C:	The state of the s	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
86-26-04 01/06/1987	Superseded by 2004-19-01	05/18/2018 Hrs: 4233.9 C:	Superseded by 2004- 19-01	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
83-17-06 09/01/1983	TO PREVENT POSSIBLE DESTRUCTUVE AILERON FLUTTER	05/18/2018 Hrs: 4233.9 C: —	N/A STC not installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
83-13-01 8/01/1983	[Recurring] TO ALERT THE PILOT TO THE POTENTIAL EFFECTS OF IMPROPER FUEL CAP SEALING	05/18/2018 Hrs: 4233.9 C:	PCW date unknown	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
79-25-07 2/13/1979	HOTHECEDED THE COOK	05/18/2018 Hrs: 4233.9 C:	N/A to aircraft serial number	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
78-01-14 /23/1978		05/18/2018 Hrs: 4233.9 C:	N/A to aircraft serial number	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

ssue Number Effective Date Amendment #	Description	Complied	Charles Agreement	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
78-26-09 01/01/1978	Superseded by 79-10-14	05/18/2018 Hrs: 4233.9 C:	Superseded by 79-10- 14	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
77-23-11 11/28/1977	HO THE CHOPE THE LONGING.	05/18/2018 Hrs: 4233.9 C:	N/A to aircraft serial number	No	D; Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
77-04-05 03/01/1977	TO PREVENT INGESTION OF THE INDUCTION AIR BOX SEAL INTO THE CARBURETOR	05/18/2018 Hrs: 4233.9 C:	PCW date unknown	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
75-16-01 07/30/1975	TO PRECLUDE INADVERTENT FUEL EXHAUSTION DUE TO INCORRECT FUEL PLACARDED CAPACITIES	05/18/2018 Hrs: 4233.9 C:	N/A to model aircraft	No	D: - Hrs: - C: -	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
75-05-02 02/21/1975	[Recurring] TO PREVENT LOSS OF ENGINE OIL, IMPROPER ENGINE LUBRICATION OR ENGINE OIL CONTAMINATION	05/18/2018 Hrs: 4233.9 C: —	N/A STC not installed	No	D: - Hrs: - C: -	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
73-17-01 08/16/1973	TO ADVISE THE PILOT CONCERNING PROPER FUEL TRANSFER PUMP OPERATION	05/18/2018 Hrs: 4233.9 C:	N/A to part number installed	No	D: Hrs: C: -	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
71-22-02 R(1) 11/09/1971	[Recurring] TO DECREASE THE POSSIBILITY OF FAILURE OF THE NOSE GEAR STRUCTURE	05/18/2018 Hrs: 4233.9 C:	PCW date unknown	No	D: - Hrs: - C: -	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Category: Engine

Position: STC convertions in 2006

ATP Revision: 05/11/2019

Manufacturer: Continental Motors

Description

P/N: O-470-50

Model: O-470-L

S/N: 2440

Issue Number Effective Date

Amendment #

Method of
Complied Compliance Recur Next Due

Facility Cert No. / Type Authorized By Signed

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	N-4 P-	Facility Cert No. / Type Authorized By
2016-16-12 09/15/2016	To prevent failure of the cylinder assemblies, which could lead to failure of the engine, in-flight, contd.	05/18/2018 Hrs: 4233.9 C:	N/A to part number installed	No	Next Due D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
2010-11-04 06/16/2010	To prevent excessive hydraulic lifter wear, which can result in loss of engine power & loss of control of the airplane	05/18/2018 Hrs: 4233.9 C:	N/A to date of Overhaul	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
2009-24-52 E 11/18/2009	Superseded by 2010-11-04	05/18/2018 Hrs: 4233.9 C:	Superseded by 2010- 11-04	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
2009-24-51 E 11/16/2009	Superseded by 2009-24-52	05/18/2018 Hrs: 4233.9 C:	Superseded by 2009- 24-52	No	D: - Hrs: - C: -	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
2009-19-07 C2 10/07/2009	[Recurring] To prevent loss of engine power due to cracks in the cylinder head, possible engine failure, and fire in,contd.	05/18/2018 Hrs: 4233.9 C:	N/A to part number installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
007-04-19 R1 5/07/2007	To prevent cylinder separation that can lead to engine failure, possible engine compartment fire, and,contd.	05/18/2018 Hrs: 4233.9 C:	N/A to part number installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
00-00-08 /26/2002		05/18/2018 Hrs: 4233.9 C:	Superseded by 96-12- 04	-No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
0-00-01 /22/2001	Assemblies listed in AD 96-12-22	05/18/2018 Hrs: 4233.9 C:	Superseded by 96-12	-No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
00-23-21 /12/2000	journal fracture, which could result in total	05/18/2018 Hrs: 4233.9 C:	N/A to date of Overhaul	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Issue Number Effective Date Amendment #	Description	Complied	Method of	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2000-08-51 E 04/28/2000	Superseded by 2000-23-21	05/18/2018 Hrs: 4233.9 C:	Superseded by 2000- 23-21	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
99-19-01 09/30/1999	To prevent crankshaft failure due to crankshaft cheek cracks, which could result in total engine power loss, contd.	05/18/2018 Hrs: 4233.9 C;	N/A to date of Overhaul	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
99-09-17 L 04/22/1999	Superseded by 99-19-01	05/18/2018 Hrs: 4233.9 C: -	Superseded by 99-19- 01	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P J/A Michael J. Tillman
98-17-11 C 10/19/1998	TO PREVENT CRANKSHAFT FAILURE DUE TO CRACKING, WHICH COULD RESULT IN AN INFLIGHT ENGINE FAILURE AND POSSIBLE, CONTD.	05/18/2018 Hrs: 4233.9 C:	N/A to date of Overhaul	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
98-01-08 E 12/23/1997	TO PREVENT EXHAUST ROLLER ROCKER ARM FAILURE, WHICH CAN RESULT IN BENT PUSH RODS, RUPTURED ROCKER ARM,CONTD.	11/16/2018 Hrs: 4233.9 C:	N/A Part not Installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
97-21-02 10/27/1997	TO PREVENT EXTREME SIDE LOADING OF THE PISTON, AND CONSEQUENT FAILURE OF THE PISTON AND ENGINE	11/16/2018 Hrs: 4233.9 C:	N/A Part not installed.	No	D: - Hrs: - C: -	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
97-15-01 L 17/30/1997	Superseded by 97-21-02	11/16/2018 Hrs: 4233.9 C:	Not applicable to dat as indicated in A.D.	teNo	D: - Hrs: - C: -	Mangon Aircraft, Inc. 2845176/A&P VA Michael J. Tillman
96-12-22 97/31/1996	[Recurring] TO PREVENT LOSS OF ENGINE OIL CAUSED BY LOOSE OR SEPARATED OIL FILTER ADAPTERS, WHICH COULD RESULT IN ENGINE, CONTD.	05/14/2019 Hrs: 4233.9 C:	C/W by visual inspection. Torque seal in place. **Due Every Oil Filter Removal**		D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
95-21-15 11/28/1995	TO PREVENT DETONATION DUE TO LOW OCTANE, WHICH CAN RESULT I SEVERE ENGINE DAMAGE AND SUBSEQUENT FAILURE	11/16/2018 NHrs: 4233.9 C:	Not applicable by "N" number and no record of contaminated fuel being used since engine overhaul.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
94-14-12 L 06/23/1994	Superseded by 95-21-15	11/16/2018 Hrs: 4233.9 C:	Not applicable Superseded by newer A.D.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
93-08-17 08/23/1993	TO PREVENT AN ENGINE FAILURE	11/16/2018 Hrs: 4233.9 C:	Not applicable to this scrial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
93-10-02 08/12/1993	TO PREVENT AN ENGINE FAILURE DUE TO A MISSING CYLINDER VALVE RETAINER KEY	11/16/2018 Hrs: 4233.9 C:	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
91-19-03 09/29/1991	TO PREVENT OPERATION WITH COLLAPSED OIL FILTER ELEMENTS, WHICH CAN RESULT IN LOSS OF OIL PRESSURE, CONTD.	11/16/2018 Hrs: 4233.9 C:	Not applicable to date code.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
88-03-06 14/15/1988	TO PREVENT POSSIBLE LOSS OF ENGINE OIL AND SUBSEQUENT ENGINE FAILURE	11/16/2018 Hrs: 4233.9 C:	Not applicable to the part number installed		D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
85-08-02 4/15/1985	TO PREVENT POSSIBLE VALVE STEM WEAR, OIL CONTAMINATION, VALVE SEIZURE, AND TOTAL LOSS OF ENGINE POWER		Not applicable to the part number installed		D; Hrs; C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
9-05-09 /12/1979		11/16/2018 Hrs: 4233.9 C:	Not applicable to thi serial number installed.	sNo	D; Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
3-13-04 20/1973		11/16/2018 Hrs: 4233.9 C:	Not applicable to thi serial number installed.	sNo	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
-15-01 11/1969		11/16/2018 Hrs: 4233.9 C:	Not applicable to the part numbers installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Street Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
61-03-02 01/01/1961	Superseded by 63-15-01	11/16/2018 Hrs: 4233.9 C:		No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
56-06-01 07/01/1956	TO PREVENT SERIOUS ENGINE DAMAGE IN FLIGHT	11/16/2018 Hrs: 4233.9 C:	Not applicable to the part numbers installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

P/N:

ATP Revision: 05/11/2019

Medel: D3A34C401

Manufacturer: McCauley

S/N: 066578

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2005-14-11 08/17/2005		11/16/2018 Hrs: 4233.9 C:	Not applicable no record of named company performing maintenance.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
82-27-02 R1 02/22/1983		11/16/2018 Hrs: 4233.9 C:	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
77-26-03 06/07/1978	TO PREVENT POSSIBLE BLADE PITCH CONTROL FAILURES	11/16/2018 Hrs: 4233.9 C:	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Category: Alternators Position: ATP Revision: 05/11/2019

Manufacturer: Motorola P/N: A152A Model: S/N: 2116

Issue Number Effective Date Amendment #			Method of			Facility Cert No. / Type Authorized By	
	Description	Complied	Compliance	Recur	Next Due	Signed	

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
					D:	
		Hrs:			Hrs:	
	The section of the se	C:	School Con		C: -	
			The state of the s			
Category: ELT Syste	ems	Positio	n:	-	ATP Re	evision: 05/11/2019

Manufacturer: ACK Technologies, Inc P/N:

Model: E-01 S/N: 006653

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
		-			D:	
		Hrs:			Hrs:	
		C:			C:	

Position: 1 ATP Revision: 05/11/2019 Category: Magnetos

P/N: 10-79020-10 Manufacturer: Bendix Corporation S/N: 840157 Model: S-20 SERIES

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2005-12-06 07/19/2005	[Recurring] To prevent failure of the magneto impulse coupling assembly and possible engine failure	05/14/2019 Hrs: 4233.9 C:	Not applicable to engine model installed.	No	D; Hrs; C;	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
96-12-07 07/18/1996	[Recurring] Superseded by 2005-12-06	11/16/2018 Hrs: 4233.9 C:	Not applicable Superseded by newer A.D.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
94-01-03 R2 06/28/1995		11/16/2018 Hrs: 4233.9 C:	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON-Hrs: 4233.9 GROUNDED MAGNETO C:	Complied	Method of Compliance Recur	Next Due	Facility Cert No. / Type Authorized By Signed
COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD. 78-09-07 R3 01/17/1983 [Recurring] Superseded by 96-12-07 74-26-09 12/24/1974 S-20,-200,-1200 SERIES MAGNETOS 11/16/2018 Hrs: 4233.9 C: 73-07-04 0/11/1973 Superseded by 94-01-03 11/16/2018 Hrs: 4233.9	A NON-Hrs: 4233.9	Not applicable to date No code.	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
74-26-09 S-20,-200,-1200 SERIES MAGNETOS 11/16/2018 Hrs: 4233.9 C: 73-07-04 Superseded by 94-01-03 11/16/2018 Hrs: 4233.9 0/11/1973	LY Hrs: 4233.9 EIGHTS C:	Not applicable to this No serial number installed.	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
73-07-04 Superseded by 94-01-03 11/16/2018 Hrs: 4233.9 C: 11/16/2018 Hrs: 4233.9	Hrs: 4233.9	Not applicable Superseded by newer A.D.	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
0/11/1973 Hrs: 4233.9	Hrs: 4233.9	Not applicable to this No serial number installed.	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
		Not applicable Superseded by newer A.D.	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Category: Magnetos Position: 2 ATP Revision: 05/11/2019

Manufacturer: Bendix Corporation P/N: 10-79020-10

Model: S-20 SERIES S/N: 504803

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2005-12-06 07/19/2005	[Recurring] To prevent failure of the magneto impulse coupling assembly and possible engine failure	05/14/2019 Hrs: 4223.9 C:	Not applicable to engine model installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
96-12-07 07/18/1996	[Recurring] Superseded by 2005-12-06	11/16/2018 Hrs: 422.39 C:	Not applicable Superseded by newer A.D.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Description	Complied	Method of			Facility Cert No. / Type Authorized By
IAND CUDCEOUTENED TO THE TOTAL	11/16/2018			D: Hrs: C:	Signed Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON-GROUNDED MAGNETO	11/16/2018 Hrs: 4233.9 C:	Not applicable to date code.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD.	11/16/2018 Hrs: 4233.9 C:	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
[Recurring] Superseded by 96-12-07	11/16/2018 Hrs: 4223,9 C:		10.50	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
S-20,-200,-1200 SERIES MAGNETOS	11/16/2018 Hrs: 4233.9 C:	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
		Not applicable Superseded by newer A.D.		D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
	TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON- GROUNDED MAGNETO TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD. [Recurring] Superseded by 96-12-07 S-20,-200,-1200 SERIES MAGNETOS Superseded by 94-01-03	TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON- GROUNDED MAGNETO TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD. [Recurring] Superseded by 96-12-07 S-20,-200,-1200 SERIES MAGNETOS 11/16/2018 Hrs: 4233.9 C: Superseded by 94-01-03 11/16/2018 Hrs: 4233.9 C:	TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE AND SUBSEQUENT ENGINE FAILURE TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON- GROUNDED MAGNETO TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD. [Recurring] Superseded by 96-12-07 [Recurring] Superseded by 96-12-07 S-20,-200,-1200 SERIES MAGNETOS TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD. To Prevent Failure of IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD. To Prevent INJURY OR DEATH TO GROUNDED MAGNETO To Prevent INJURY OR DEATH TO HIT/16/2018 Hrs: 4233.9 Not applicable to this serial number installed. Not applicable to this serial number installed. To Prevent Injury OR DEATH TO GROUNDED MAGNETO To Prevent Injury OR D	TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON-GROUNDED MAGNETO TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD. [Recurring] Superseded by 96-12-07 S-20,-200,-1200 SERIES MAGNETOS Compliance Not applicable to this No serial number installed. Not applicable No Superseded by newer A.D. S-20,-200,-1200 SERIES MAGNETOS 11/16/2018 Hrs: 4223.9 C: A.D. Not applicable to this No serial number installed. Not applicable to this No serial number installed. Not applicable to this No serial number installed.	TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE AND SUBSEQUENT ENGINE FAILURE TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON- GROUNDED MAGNETO TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD. Recurring] Superseded by 96-12-07 S-20,-200,-1200 SERIES MAGNETOS TI/16/2018 Hrs: 4233.9 C: Superseded by 94-01-03 TI/16/2018 Hrs: 4233.9 C: Superseded by 94-01-03 TI/16/2018 Hrs: 4233.9 C: Superseded by nower C: Superseded by 94-01-03 TI/16/2018 Hrs: 4233.9 Superseded by nower C: Superseded by 94-01-03 TI/16/2018 Hrs: 4233.9 Superseded by nower C: Superseded by 94-01-03 TI/16/2018 Hrs: 4233.9 Superseded by nower C: Superseded by 94-01-03 TI/16/2018 Hrs: 4233.9 Superseded by nower C: Superseded by 94-01-03 TI/16/2018 Hrs: 4233.9 Superseded by nower C: Superseded by 94-01-03 TI/16/2018 Hrs: 4233.9 Superseded by nower C: Superseded by 94-01-03 TI/16/2018 Hrs: 4233.9 Superseded by nower C: Superseded by nower C: Superseded by 94-01-03 TI/16/2018 Hrs: 4233.9 Superseded by nower C: Superseded by nower C: Superseded by 94-01-03 TI/16/2018 Hrs: 4233.9 Superseded by nower C: Superseded by nower C: Superseded by nower C: Superseded by nower C: Superseded by 94-01-03 TI/16/2018 Hrs: 4233.9 Superseded by nower C: Superseded by nower C: Superseded by 94-01-03 TI/16/2018 Hrs: 4233.9 Superseded by nower C: Superseded by nower C: Hrs: C: TO PREVENT INJURY OR DEATH TO Bristalled. C: C: C: C: TO PREVENT INJURY OR DEATH TO Bristalled. C: C: TO PREVENT INJURY OR DEATH TO Bristalled. C: C: TO PREVENT INJURY OR DEATH TO Bristalled. C: Tropical number Installed. C: Hrs: C: TO PREVENT INJURY OR DEATH TO Bristalled. C: Tropical number Installed. C: C: Tropical number I

Category: Starter Position: ATP Revision: 05/11/2019

 Manufacturer:
 Kelly Aerospace
 P/N: ES-646238-2

 Model:
 ES-646238-2
 S/N: G031909

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
	The second second	HALL THE PLANT		-	D:	
		Hrs:			Hrs:	
		C:			C:	

Category: Vacuum System Position: ATP Revision: 05/11/2019

Manufacturer: Airborne Manufacturing P/N: 216CW

Model: 216CW S/N: 51680

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
98-23-01 11/20/1998	TO PREVENT FAILURE OF THE PRIMARY DRY AIR PUMP CAUSED BY DEFECTIVE FLEXIBLE COUPLING, WHICH COULD, CONTD.	11/16/2018 Hrs: 4233.9 C:	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Category: Batteries Position: ATP Revision: 05/11/2019

Manufacturer: Teledyne Battery P/N: G35

Model: Lead/Acid Battery S/N: G02972214

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
		 Hrs: C:			D: Hrs: C:	



Mangon Aircraft, Inc.

501 Sky Ranch Drive Petaluma, CA 94954-3908 Phone: (707) 765-1848

Fax: (707) 765-6610 Email: info@mangonaircraft.com

Date: 5/14/2019

Customer Record

Kopan Taber (Reg. Number: 2376G)

Contact Info:

232 Mocking Bird Circle Santa Rosa, CA 95409 536-9614

Aircraft Info:

Reg. No: 2376G Serial No.: 51676

A/C Info: 1959 Cessna 182B

 Current Engine Information:

 Cyl:
 1
 2
 3
 4
 5
 6

 B/L 44
 60
 70
 75
 72
 70
 68

Last Oil Change:

Date: 5/14/2019 Tach: 4233.9

Current W/O or W/O Quote Number: 15739

Dates Due:

Annual: 5/30/2020

ELT Battery: 2/29/2020

Corr. Treat: O/2 Bottle:

FAR 91.411: 11/30/2020

FAR 91.413: 11/30/2020

Aircraft Information	on		Eng. Model No.	Eng O/H Date	Eng TSMOH
A/C TT 4233.9	A/C Tach 4233.9			6/3/2007	51
Eng TT	Eng SN	Eng TBO	TT Vac. Pump	A/C Battery Date 5/18/2018	Eng TSN 2773.9
2773.9	68680-8-L-4				Last Prop Balance
Prop O/H Date 6/3/2007	Prop S/N 060578	Prop Model No. D3A4C4-01-C	Prop TBO 2000	Prop TT 51	Last Prop Balance
Heater Hobbs	100 Hour 4333.9	50 Hour 4283.9	Warranty Date	Prop TSPOH	Prop TSN 51
Eng Manufacturer Continental	Prop Manufacturer McCauley	Airswitch	Reg. Expires 07/31/2020	Gross Weight	

Work Order / Quote History:

15739, 14253, 13673, 13508, 13180, 12893, 12846, 12763, 12534, 12479, 11371

Notes:



Mangon Aircraft, Inc. 501 Skyranch Rd Petatluma CA 94954-3908 USA

Phone: (707) 765-1848

FAA Airworthiness Directives Compliance Record

Company:

Category: Airframe

Manufacturer: Cessna Aircraft Company

Model: 182B

Position:

P/N:

S/N: 51676

Aircraft Registration No: 2376G

ATP Revision: 11/15/2018

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2011-10-09 06/17/2011	[Recurring] To prevent seat slippage or the seat roller housing from departing the seat rail, which may consequently cause, contd.	11/16/2018 Hrs: 4233.9 C:	C/W paragraph's (e) through (g)	Yes	D: 11/30/2019 Hrs: 4333.9 C: –	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
2008-26-10 C 01/05/2009	To prevent erroneous indications from the altimeter, airspeed, and vertical speed, contd.	05/18/2018 Hrs: 4233.9 C:	N/A to part number installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
2008-10-02 05/12/2008	To prevent erroneous indications from the altimeter, airspeed, and vertical speed, contd.	05/18/2018 Hrs: 4233.9 C:	N/A this part not installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
2004-19-01 C 1/01/2004		05/18/2018 Hrs: 4233.9 C:	N/A this part not installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
98-16-04 19/21/1998	TO PREVENT WING FAILURE DURING FLIGHT CAUSED BY THE ABSENCE OF AN ANGLE STIFFENER, WHICH COULD CAUSE LOSS OF, CONTD.		N/A STC is not installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
97-01-13 2/03/1997	HYDRAULIC SYSTEMS FAILURE	05/18/2018 Hrs: 4233.9 C:	N/A to part number installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Ssue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
96-12-22 7/31/1996		11/16/2018 Hrs: 4233.9 C:	C/W by visual inspection. Torque seal in place. **Due Every Oil Filter Removal**	Yes	D: Hrs: 4283.9 C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
87-20-03 R2 19/24/1990	[Recurring] Superseded by 2011-10-09	05/18/2018 Hrs: 4233.9 C:	Superseded by 2011- 10-09	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
84-10-01 R1 07/05/1988	[Recurring] TO PREVENT POWER LOSS OR ENGINE STOPPAGE DUE TO WATER CONTAMINATION OF FUEL SYSTEM	11/16/2018 Hrs: 4233.9 C:	P/C/W Cessna Service Kit SK182- 85 is installed. No further action is required.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
79-10-14 R1 05/30/1988	TO PROVIDE AN ALTERNATE SOURCE OF FUEL TANK VENTING IN CASE OF FUEL TANK VENT OBSTRUCTION BY FOREIGN MATERIAL, CONTD.	05/18/2018 Hrs: 4233.9 C:	PCW date unknown	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
86-26-04 01/06/1987		05/18/2018 Hrs: 4233.9 C:	Superseded by 2004- 19-01	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
83-17-06 9/01/1983		05/18/2018 Hrs: 4233.9 C:	N/A STC not installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
83-13-01 8/01/1983	[Recurring] TO ALERT THE PILOT TO THE POTENTIAL EFFECTS OF IMPROPER FUEL CAP SEALING	05/18/2018 Hrs: 4233.9 C:	PCW date unknown	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
79-25-07 2/13/1979		05/18/2018 Hrs: 4233.9 C:	N/A to aircraft serial number	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
78-01-14 01/23/1978	TO PRECLUDE THE POSSIBILITY OF AN IN-FLIGHT FIRE DUE TO A LOOSE ELT ANTENNA COAXIAL CABLE CONNECTOR, CONTD.	05/18/2018 Hrs: 4233.9 C:	N/A to aircraft serial number	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

11/16/2018 12:26 PM 2/11

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Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
78-26-09 01/01/1978	Superseded by 79-10-14	05/18/2018 Hrs: 4233.9 C: —	Superseded by 79-10- 14	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
77-23-11 1/28/1977	TO PRECLUDE THE POSSIBILITY OF AN IN-FLIGHT FIRE DUE TO A LOOSE ELT ANTENNA COAXIAL CABLE CONNECTOR, CONTD.	05/18/2018 Hrs: 4233.9 C: —	N/A to aircraft serial number	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
77-04-05 03/01/1977	TO PREVENT INGESTION OF THE INDUCTION AIR BOX SEAL INTO THE CARBURETOR	05/18/2018 Hrs: 4233.9 C: —	PCW date unknown	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
75-16-01 07/30/1975	TO PRECLUDE INADVERTENT FUEL EXHAUSTION DUE TO INCORRECT FUEL PLACARDED CAPACITIES	05/18/2018 Hrs: 4233.9 C;	N/A to model aircraft	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
75-05-02 02/21/1975	[Recurring] TO PREVENT LOSS OF ENGINE OIL, IMPROPER ENGINE LUBRICATION OR ENGINE OIL CONTAMINATION	05/18/2018 Hrs: 4233.9 C:	N/A STC not installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
73-17-01 08/16/1973	TO ADVISE THE PILOT CONCERNING PROPER FUEL TRANSFER PUMP OPERATION	05/18/2018 Hrs: 4233.9 C:	N/A to part number installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
71-22-02 R(1) 1/09/1971	[Recurring] TO DECREASE THE POSSIBILITY OF FAILURE OF THE NOSE GEAR STRUCTURE	05/18/2018 Hrs: 4233.9 C: —	PCW date unknown	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
Category: Engine	Description of the latest terms	Position:	STC convertions in		ATP Revision:	11/15/2018
Manufacturer: Continenta Model: O-478-L	Motors	P/N: O-4 S/N: 244				
Ssue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed

Issue Number Effective Date Amendment #	Description	Complied		Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2016-16-12 09/15/2016	To prevent failure of the cylinder assemblies, which could lead to failure of the engine, in-flight, contd.	05/18/2018 Hrs: 4233.9 C:		No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
2010-11-04 06/16/2010		05/18/2018 Hrs: 4233.9 C: —	N/A to date of Overhaul	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
2009-24-52 E 11/18/2009		05/18/2018 Hrs: 4233.9 C:	Superseded by 2010- 11-04	No	D: Hrs: C: -	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
2009-24-51 E 11/16/2009		05/18/2018 Hrs: 4233.9 C:	Superseded by 2009- 24-52	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
009-19-07 C2 0/07/2009	[Recurring] To prevent loss of engine power due to cracks in the cylinder head, possible engine failure, and fire in,contd.	r 05/18/2018 Hrs: 4233.9 C: —	N/A to part number installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
907-04-19 R.1 5/07/2007	The Control of the Co	05/18/2018 Hrs: 4233.9 C:	N/A to part number installed	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
0-00-08 (26/2002	Motors reciprocating engines listed in AD	05/18/2018 Hrs: 4233.9 C: —	Superseded by 96-12- 04	2-No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P VA Michael J. Tillman
-96-01 22/2001	Assemblies listed in AD 96-12-22	05/18/2018 Hrs: 4233.9 C: -	Superseded by 96-12 22	2-No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
0-23-21 2/2000	journal fracture, which could result in total	05/18/2018 Hrs: 4233.9 C: –	N/A to date of Overhaul	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Issue Number Effective Date Amendment #	Description	Complied		Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2000-08-51 E 04/28/2000	Superseded by 2000-23-21	05/18/2018 Hrs: 4233.9 C:	Superseded by 2000- 23-21		D: Hrs: C: -	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
99-19-01 09/30/1999	To prevent crankshaft failure due to crankshaft cheek cracks, which could result in total engine power loss, contd.	05/18/2018 Hrs: 4233.9 C:	N/A to date of Overhaul	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
99-09-17 L 04/22/1999	Superseded by 99-19-01	05/18/2018 Hrs: 4233.9 C:	Superseded by 99-19- 01	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
98-17-11 C 10/19/1998	DUE TO CRACKING, WHICH COULD	05/18/2018 Hrs: 4233.9 C: —	N/A to date of Overhaul	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
98-01-08 E 12/23/1997	TO PREVENT EXHAUST ROLLER ROCKER ARM FAILURE, WHICH CAN	11/16/2018 Hrs: 4233.9 C:	N/A Part not Installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
97-21-02 0/27/1997	LOADING OF THE PISTON, AND	11/16/2018 Hrs: 4233.9 C:	N/A Part not installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
7-15-01 L /30/1997	- Marine and the same of the s	11/16/2018 Hrs: 4233.9 C:	Not applicable to date as indicated in A.D.	:No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
5-12-22 31/1996	ENGINE OIL CAUSED BY LOOSE OR H	11/16/2018 Hrs: 4233.9 C:	C/W by visual inspection. Torque seal in place. **Due Every Oil Filter Removal**	Yes	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
21-15 8/1995	TO PREVENT DETONATION DUE TO LOW OCTANE, WHICH CAN RESULT IN	11/16/2018 Hrs: 4233.9 C: -		No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
94-14-12 L 06/23/1994	Superseded by 95-21-15	11/16/2018 Hrs: 4233.9 C:		No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
93-08-17 08/23/1993		11/16/2018 Hrs: 4233.9 C:	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
93-10-02 TO PREVENT AN ENGINE FAILURE DUE TO A MISSING CYLINDER VAL' RETAINER KEY		11/16/2018 EHrs: 4233.9 C: —	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
91-19-03 19/29/1991	COLLAPSED OIL FILTER ELEMENTS,	11/16/2018 Hrs: 4233.9 C:	Not applicable to date code.	:No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
88-03-06 4/15/1988	ENGINE OIL AND SUBSEQUENT	11/16/2018 Hrs: 4233.9 C:	Not applicable to the part number installed.		D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
5-08-02 /15/1985	TO PREVENT POSSIBLE VALVE STEM WEAR, OIL CONTAMINATION, VALVE SEIZURE, AND TOTAL LOSS OF ENGINE POWER		Not applicable to the part number installed.		D; Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
9-05-09 12/1979	OIL PRESSURE INDICATION	11/16/2018 Hrs: 4233.9 C:	Not applicable to this serial number installed.	sNo	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
-13-04 0/1973	Į.	11/16/2018 Hrs: 4233.9 C: -	Not applicable to this serial number installed.	sNo	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
15-01 1/19 69	Į.	11/16/2018 Hrs: 4233.9 C:	Not applicable to the part numbers installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur		Facility Cert No. / Type Authorized By Signed
61-03-02 01/01/1961	Superseded by 63-15-01	11/16/2018 Hrs: 4233.9 C: —	Not applicable Superseded by newer A.D.	No	D:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
56-06-01 07/01/1956	TO PREVENT SERIOUS ENGINE DAMAGE IN FLIGHT	11/16/2018 Hrs: 4233.9 C: -	Not applicable to the part numbers installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
Category: Propeller		Position:			ATP Revision:	11/15/2018
Manufacturer: McCaul	ey	P/N:				
Model: D3A34C401	Printed and the second	S/N: 066	5578			
Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2005-14-11 08/17/2005	To prevent blade failure that could result in separation of a propeller blade and loss of control of the airplane	11/16/2018 Hrs: 4233.9 C:	Not applicable no record of named company performing maintenance.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
82-27-02 R1 2/22/1983	TO PREVENT POSSIBLE PROPELLER BLADE SHANK FAILURE	11/16/2018 Hrs: 4233.9 C:	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
77-26-03 5/07/1978		11/16/2018 Hrs: 4233.9 C: —	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
ategory: Alternators anufacturer: Motorola odel:	particular from the con-	Position: P/N: A15 S/N: 2116	The late		ATP Revision:	11/15/2018
sue Number fective Date nendment #	PORTS OF THE PARTY	the cone	Method of			Facility Cert No. / Type Authorized By

Issue Number Effective Date Amendment #	Description	Method of Complied Compliance Recur Next Due				Facility Cert No. / Type Authorized By Signed
	SERVICE SERVICE OF THE SERVICE	Hrs: C:		D: Hrs: C:		
Category: ELT Syst Manufacturer: ACK Model: E-01		Position: P/N: S/N: 00			ATP Revision:	11/15/2018
Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
	The second secon	 Hrs: C:			D: Hrs: C:	a Am J Tamas
Category: Magnetos Manufacturer: Bendi Model: S-20 SERIES		Position P/N: 10 S/N: 84	-79020-10		ATP Revision:	11/15/2018
Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2095-12-06 07/19/2005	[Recurring] To prevent failure of the magneto impulse coupling assembly and possible engine failure	11/16/2018 Hrs: 4233.9 C:	Not applicable to engine model installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
96-12-07 07/18/1 996	[Recurring] Superseded by 2005-12-06	11/16/2018 Hrs: 4233.9 C:	Not applicable Superseded by newer A.D.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
94-01-03 R2 06/28/1995	TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE	11/16/2018 Hrs: 4233.9 C:	Not applicable to this serial number installed.	No	D; Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
94-06-09 05/20/1994	TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON- GROUNDED MAGNETO	11/16/2018	Not applicable to date code.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
82-20-01 06/14/1983	TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD.	11/16/2018 Hrs: 4233.9 C: —	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
78-09-07 R3 01/17/1983		11/16/2018 Hrs: 4233.9 C:	Not applicable Superseded by newer A.D.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
74-26-09 2/24/1974		11/16/2018 Hrs: 4233.9 C:	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
73-07-04 0/11/1973		11/16/2018 Hrs: 4233.9 C: —	Not applicable Superseded by newer A.D.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
stegery: Magnetos	Date of the State	Position:	2	-	ATP Revisi	on: 11/15/2018

Manufacturer: Bendix Corporation

Model: S-20 SERIES

P/N: 10-79020-10

S/N: 504803

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2005-12-06 07/19/2005	[Recurring] To prevent failure of the magneto impulse coupling assembly and possible engine failure	11/16/2018 Hrs: 4223.9 C:	Not applicable to engine model installed.	Yes	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
96-12-07 17/18/1996	[Recurring] Superseded by 2005-12-06	11/16/2018 Hrs: 422.39 C:	Not applicable Superseded by newer A.D.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
94-01-03 R2 06/28/1995	TO PREVENT MAGNETO FAILURE	11/16/2018 Hrs: 4233.9 C:	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P VA Michael J. Tillman
94-05-09 05/20/1994	TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON- GROUNDED MAGNETO	11/16/2018 Hrs: 4233.9 C:	Not applicable to date code.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
82-20-01 06/14/1983	TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD.	11/16/2018 Hrs: 4233.9 C:	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
78-09-07 R3 11/17/1983	[Recurring] Superseded by 96-12-07	11/16/2018 Hrs: 4223.9 C:	Not applicable Superseded by newer A.D.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
74-26-09 2/24/1974	S-20,-200,-1200 SERIES MAGNETOS	11/16/2018 Hrs: 4233.9 C:	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
73-07-04 0/11/1973		11/16/2018 Hrs: 4233.9 C:	Not applicable Superseded by newer A.D.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman
stegory: Starter		Position	:		ATP	Revision: 11/15/2018

Manufacturer: Kelly Aerospace

Model: ES-646238-2

P/N: ES-646238-2

S/N: G031909

Category: Vacuum System

Position:

ATP Revision: 11/15/2018

Manufacturer: Airborne Manufacturing

P/N: 216CW

Model: 216CW

S/N: 51680

Issue Number Effective Date Amendment #	Description	Complied	Method of	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
98-23-01 11/20/1998		11/16/2018	Not applicable to this serial number installed.	No	D: Hrs: C:	Mangon Aircraft, Inc. 2845176/A&P I/A Michael J. Tillman

Category: Batteries

Position:

ATP Revision: 11/15/2018

Manufacturer: Teledyne Battery

P/N: G35

Model: Lead/Acid Battery

S/N: G02972214

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
		Hrs: C:		-	D: Hrs: C:	

530-459-3392 2017 Mail: 11423 AXINITE CT. Shipping: 900A WEST OLD MONTAGUE RD. Montague, CA 96064

Bill To							
Kopan Constructi Taber L Kopan 24 lower drive Mill Valley, CA 9							
Mill Valley, CAS	74541		,			Date	
						6/3/2007	
P.O. No.	Terms	Reg#	Aircraft Model & Sn#	Tac	ch / Meter	Work	Order#
1671	Due on receipt	N2376G	182B/51676		4182.8	1	671
Item		Description	on	Qty	Rate S	Serviced	Amount
ANNUAL INS	SMOH,ENGINE: .1 AD'S CHECKED TELT BATTERY IS TRANSPONDER C C/W FAR 91.207D NO PROBLEMS NO C/W AD 76-07-12 E PROBLEMS NOTE C/W AD 96-12-22 E TORQUE SEAL. PI C/W AD 84-10-01 E FOUND, PERFORN C/W AD 83-13-01 E FUEL LEAKS, NO C/W AD 87-20-03 E FUNCTIONAL OF MECHANISM, NO C/W AD 77-04-05 E	0 TT,PROP SINCE I HRU REVISION 200 DUE REPLACEME HECK IS DUE BY VISUAL AND F OTED, REPLACE B SY FUNCTION OF ' ID, PERFORM EA 1 BY NISPECTION, S ERFORM EA FILTE PART 1 BY VISUAL M EA 100HR/ANNU BY VISUAL OF PLA NE FOUND, PERFO Z BY VISUAL, DIP THE SEAT TRACK PROBLEMS NOTE BY INSTALLING N R BOX INTAKE FI	TUNCTIONAL OF THE ELT, ATTERIES MARCH 2008 THE IGN SWITCH, NO 00HR/ANNUAL AFTIE AND APPLYING OF ER REMOVAL FOR FUEL LEAKS, NONE JAL ACARD, OK, VISUAL FOR DRM EA 100HR/ANNUAL	22	60.00	21/2007	1,320.00
LABOR	\$60.00 PER HOUR	REPLACEMENT	OF ENGINE AND	50	60.00	6/21/2007	3,000.00
ABOR				40	60.00	6/21/2007	2,400.00
	A toucher of a care	tay to con	PATER BEAUTIER	Tota			
			And the State of t	Paymo	ents/Cred	dits	1-1-1
				Balan			

530-459-3392 2017 Mail: 11423 AXINITE CT. Shipping: 900A WEST OLD MONTAGUE RD. Montague, CA 96064

Bill To								
Kopan Constructi Taber L Kopan 24 lower drive Mill Valley, CAS							Date	
					-		6/3/2007	
	Terms	Reg#	Reg # Aircraft Model & Sn#				Work	Order#
P.O. No.	Due on receipt N2376G 182B/51676				4	4182.8	1	671
1671	Due on receipt		Description			Rate	Serviced	Amount
Item	\$60.00 PER HOUR	Descriptio		4	1	60.00	6/21/2007	240.00
ABOR ARTS HIPPING/FR ARTS HIPPING/FR ARTS HIPPING/FR ARTS HIPPING/FR ARTS ARTS ARTS ARTS ARTS ARTS ARTS ARTS	1. RESEARCH AD'S UP OF LOGS 1EA. NEW FIREWASHIPPING / FREIG 1EA. NEW BATTEL SHIPPING / FREIG 2EA. WING VENT SHIPPING / FREIG 1EA. BRAKE DISK SHIPPING / FREIG FIREWALL TO EN WASHERS BOLTS		1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	75.00 212.03 10.00 50.80 10.00 132.00 150.88 15.00 9.37 1.20 0.73 3.7 12.4 0.5 7.1 17.8 35.0	6/21/2007 6/21/2007 6/21/2007 6/21/2007 6/21/2007 6/21/2007 6/21/2007 6/21/2007 6/21/2007 6/21/2007 6/21/2007 6/21/2007 6/21/2007 6/21/2007 6/21/2007 7 6/21/2007 7 6/21/2007 7 6/21/2007 8 6/21/2007 1 6/21/2007 1 6/21/2007 8 6/21/2007 8 6/21/2007 8 6/21/2007 8 6/21/2007 8 6/21/2007 8 6/21/2007			
				То	ta	d		19.0
				Pa	ym	ents/C	redits	
				Ва	lar	nce Du	е	

530-459-3392 2017 Mail: 11423 AXINITE CT. Shipping: 900A WEST OLD MONTAGUE RD. Montague, CA 96064

Bill To Kopan Constru- Taber L Kopan								
24 lower drive Mill Valley, CA				Г		Date		
						6/3/2007		
P.O. No.	Terms	Reg#	Aircraft Model & Sn#	Ta	ach / Meter	Work	k Order#	
1671	Due on receipt	N2376G	182B/51676		4182.8		1671	
Item		Description	1	Qty	Rate	Serviced	Amount	
LABOR	\$60.00 PER HOUR	ADDITIONAL SER	RVICE OOT FAIRINGS	4	60.00	6/21/2007	240.00	
ABOR	\$60.00 PER HOUR 1. DRAIN FUEL, RE CLEAN, INSTALL	FABRICATE AND INSTALL NEW ROOT FAIRINGS 50.00 PER HOUR ADDITIONAL SERVICE DRAIN FUEL, REMOVE MAIN FUEL SHUT OFF VALVE, LEAN, INSTALL NEW ACKINGS, REINSTALL, ADD FUEL			60.00	6/21/2007	270.00	
ABOR	\$60.00 PER HOUR 1.GET PLANE REA SECURE FERRY PE MONTAGUE BEFO Sales Tax	DY FOR FERRY FR ERMIT AND HAD P	OM SISKIYOU COUNTY, LANE FLOWN TO	10	60.00 7.25%	5/14/2007	600.00 340.48	
				Tota			\$13,226.	
accept Checks	s or Cash only				ents/Cr	edits	-\$13,226.	
					ice Due		\$0.	

530-459-3392 2017 Mail: 11423 AXINITE CT. Shipping: 900A WEST OLD MONTAGUE RD. Montague, CA 96064

INVOICE

Construction
L Kopan
ower drive
Valley, CA 94941

						Date	
						7/4/2007	
P.O. No.	Terms	Reg #	Aircraft Model & Sn#	1	Tach / Meter	r Wor	k Order#
1686	Due on receipt	N2376G	182B/51676		4183.5		1686
Item		Description		Qty	Rate	Serviced	Amount
HIPPING / FR HIPPING / FR HIPPING / FR PARTS OTHER CHAR	PANEL 3. SHIP OUT FOR I 4. REINSTALLED A 5. PUMP FUEL BA SHIPPING / FREIG 1. SHIP CELL PHO SHIPPING / FREIG 1. SHIP COMPONE OTHER CHARGE 1. REPAIR OF FUE SHIPPING / FREIG	ER AND REMOVE GAC REPAIR AFTER RETURN CK INTO TANK HT NE HT INTS OUT FOR REPAIR L GAGE AND SENDER HT SENDER AND GAGE ETS		9.5	29.28 20.03 405.55		29.28 20.03 405.55 35.00 28.00 25.00 2.03
We accept Checks	or Cash only			Total			\$1,162.39
			11 - 12 - 14	Payme	ents/Cre	dits	-\$1,162.39
			ESTREE IN CO.	Balanc			

530-459-3392 2017 Mail: 11423 AXINITE CT. Shipping: 900A WEST OLD MONTAGUE RD. Montague, CA 96064

Bill To								
opan Construction aber L Kopan 4 lower drive Aill Valley, CA 94				Г		Date		
		1011		-		5/29/2009		
			110 Catt	T	ach / Meter	Work	Order#	
P.O. No.	Terms	Reg #	Aircraft Model & Sn#		4202.4	-	1870	
1870	Due on receipt	N2376G	182B/51676			Serviced	Amount	
Item		Description	n	Qty	Rate 0.00		0.00	
LABOR	TTAF: 4183.0 TT.F ELT BATTERY IS AD'S CHECKED T \$70.00 PER HOUR PERFORMED AN FAR 43 APPENDE OF ALL CONTRO WHEEL BEARING C/W FAR 91.207D NO PROBLEMS N C/W AD 76-07-12 PERFORM EA 100 C/W AD 96-12-22 OF TORQUE SEA REMOVAL OR 50 C/W AD 83-13-01 LEAKS, NONE FO C/W AD 83-13-01 100HR/ANNUAL DIMENSIONAL ROLLERS AND I PERFORM EACH C/W AD 77-04-05 INTAKE BOX, PI COMPRESSIONS	ENG: 2723.0 SOH, END DUE REPLACEME THRU REVISION 20 ENUAL INSPECTION X D. CHECKED TR. LS, LUBED ALL COMES THIS INSP. SERVED TO THE SERVED TO T	IN ACCORDANCE WITH AVELS AND DEFLECTIONS DITROLS. LUBED ALL VICED BRAKES, AND FUNCTION OF THE ELT, BATTERY MARCH 2012. THE IGN SWITCH, OK, AFTIE, AND APPLICATION DAPTER. PERFORM EA L INSPECTION FOR FUEL A 100HR/ANNUAL. FUEL LEAKS, PERFORM EA 2 BY VISUAL, OF THE SEAT TRACKS, IISM, NO PROBLEMS NOTED, OF GASKET ON CARB			5/29/2009	1,400.00	
				Tota	al			
				Payn	nents/C	redits		
				Balance Due				

530-459-3392 2017 Mail: 11423 AXINITE CT. Shipping: 900A WEST OLD MONTAGUE RD. Montague, CA 96064

Bill To							
Kopan Construc Taber L Kopan 24 lower drive							100
Mill Valley, CA	\ 94941					Date	
						5/29/2009	h Albert
P.O. No.	Terms	Reg#	Aircraft Model & Sn#		Tach / Met	er Wor	k Order #
1870	Due on receipt	N2376G	182B/51676		4202.4		1870
Item		Description	n	Qty	Rate	Serviced 5/29/2009	Amount
	PLUGS, CHECKED EXHAUST. WASH PERFORMED MIN	MAGS TO ENGINED ENGINED ENGINED OWN OR ALTERATION	D, TESTED AND ROTATED E TIMING, OK. INSPECTED I AND INSPECTED OK. BY THE DYNAMIC FATIC RPM. TO .16 IPS @	То	tal		
	THE PARTY OF			Pav	ments/C	redits	
				-			
				Pal	ance Due		

530-459-3392 2017 Mail: 11423 AXINITE CT. Shipping: 900A WEST OLD MONTAGUE RD. Montague, CA 96064

		_	NVOICE					
Bill To								
Kopan Construct Taber L Kopan 24 lower drive Mill Valley, CA								
						Date		
						5/29/20	09	
P.O. No.	Terms	Reg#	Aircraft Model & Sn#		Tach / Mete	er V	Vork Order #	
1870	Due on receipt	N2376G	182B/51676		4202.4		1870	
Item		Descriptio	n	Qty	Rate	Serviced	Amount	
LABOR	\$70.00 PER HOUR BALANCE OF PRO	PELLER		4	70.00	5/29/2009 6/5/2009	280.00	
PARTS	12QTS AEROSHELI	L 100 MINERAL		12	4.25	6/5/2009	51.001	
PARTS	1 EA. OIL FILTER			1	24.38	6/5/2009	24.38T	
PARTS	IEA. AIR FILTER			1	17.88	6/5/2009	17.88T	
Dil	1EA. SOLVENT			1	9.38		9.38T	
Dil	IEA. GREASE			1	4.58	6/5/2009	4.58T	
ARTS	ELT BATTERIES			1	18.60	6/5/2009	18.60T	
ARTS	FUEL VALVE LEFT			1		6/5/2009	18.55T	
HIPPING / FR	SHIPPING / FREIGH	Г		1	9.94	6/5/2009	9.94	
	Sales Tax				8.25%		11.91	
. 4								
				Total			\$1.945.22	
accept Checks o	r Cash only		William Indiana		1 10		\$1,846.22	
			Ballaki -		nts/Cred	its	-\$1,846.22	
				Balance	Due		\$0.00	

Bepartment of Transportation -- Nederal Abiation Administration

Supplemental Type Certificate

Number SA5664NM

This certificate issued to

DV Aviation, Inc 16384 Airport Road, Suite #8 Brainerd, MN 56401

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon moets the aircorrhiness requirements of Part 3 of the Civil Air Regulations. (See Type Certificate Data Sheet No. 3A13 for complete certification basis).

Original Product - Type Certificate Number:

3A13

Make:

Model:

182H, J, K, L, M, N, and P

Description of Type Design Change. Installation of Continental/P.Ponk 0-470-50 engine and a McCauley D2A34C58/90AT-8, or 2A34C66/90AT-8, or D3A32C90/82NC-2, or D3A34C401/90DFA-10, or D2A37C230/90REB-8 propeller in accordance with P.Ponk Report Number PPA-10003, Revision A, dated September 22, 2005 or later FAA approved revision.

Similations and Conditions: Approval of this change in type design applied to the above model aircraft only. This approval should not be extended to other aircraft of this model on which other previously approved modifications have been incorporated unless it is determined that the relationship between this change and any of those other previously approved modifications, including changes in type design will introduce no adverse effect upon the airworthiness of that aircraft. A copy of this certificate, its continuation sheet, and FAA approved Aircraft Flight Manual Supplement (AFMS) dated December 1, 2005 shall be maintained as part of the permanent records for the modified aircraft.

If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration

Date of application: May 12, 1991

12, 1991

Tale reissued: March 21, 2019

Date of issuance: January 15, 1992

Date amended: February 9, 1994, December 2, 2005



By direction of the Administrator Original Signed By: Manzoor H. Javed for

(Signature)
Monica L. Nemecek, Manager
Airframe & Administrative Services Section
Chicago ACO Branch

(Title)

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification
Section Section 901 Federal Aviation Act 1958)

Office Identification
Section Office Identification
Sect Federal Aviation Administration

	Make Cessna				Model 182B			**		
Aircraft	Serial No. 51676			Nationality and Registration Mark USA N2376G						
. Owner		on registration certific ER L.	ate)		Address (As shown on registration certificate) 24 LOWER DR. MILL VALLEY, CA 94941					
	The daily absention is near requirements a making to confirmal	destified herein compl and in appeared only for a fragmental for a pass	or with the s or the ghorn or against	3. For FAA Use Capplicable airword dentaled circuit	only					
N.	8/15/67	Thomas	(Leek		_		E Time			
-		Principal Alread	thisees less	4. Unit identificat	25		5. Type			
Unit	M	ake		Model		Serial No.	Repair	Alteration		
AIRFRAME			(As descri	ibed in item 1 a	bove)			x		
POWERPLA	ANT				*					
PROPELLE	R		Ř							
	Туре									
APPLIANC	E Manufacturer									
			6	6. Conformity Sta	tement					
A. Agency	s Name and Address		B. F	Kind of Agency		C. Certificate N	0.			
Steelman	Aviation, LLC		Ø	U.S. Certificated	Mechanic	7568 I.A.				
	Steelman		_	Foreign Certifica	The state of the s	77500 1.74.				
	est Old Montague	Rd.	-	Certificated Repair Station						
D. I cert have	been made in accorda		unit(s) identifi ts of Part 43	ied in item 4 above		everse or attachments and that the information				
Date JUNE 21		and some of my knowledge	jo.	Signature of Auti	horized Individual	Heet &	Steel	In		
			7. Ap	pproval for Return						
		ersons specified below, ation Administration and	- Provide	- Innered	inspected in the manne	or prescribed by the				
. [FAA Fit Standards Inspector	Manufacturer	X	Inspection Author	rization	Other (Specify)				
BY	FAA Designee	Repair Station		Person Approved Canada Airworth						
Date of Ap	proval or Rejection	Certificate or Designation No. A&P 52070756	314	Signature of Auti Robert E. Sto	horized Individual	best E &	41			

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous afterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets, Identify with aircraft nationality and registration mark and date work completed.)

DATE: JUNE 21, 2007 N2378G SN# 51676 TTAF; 4183.0

THIS IS A FOLLOW ON FIELD APPROVAL TO N2401G 182B SN# 51701

REMOVED CONTINENTAL ENGINE AND MCCAULEY PROPELLER

INSTALLED P. PONK 0470-50 ENGINE IN ACCORDANCE WITH REPORT NUMBER PPA-10003 PER STC# SA 5664NM. UTILIZED THE ORIGINAL ENGINE MOUNT, BAFFLING, INTAKE AND EXHAUST SYSTEMS AND COWLING.

its and in expensed only for the above described elected,

INSTALLED MCCAULEY PROPELLER MODEL D3A-34C401/90DF A-8 3 BLADED PROPELLERSN# 060878 The data; when about idealified functin complies with the applicable airworthi-

UPDATED WEIGHT AND BALANCE

LIMITATIONS:

ENGINE: CONTINENTAL / P. PONK 0470-50

FUEL: 100/130 MINIMUM GRADE AVIATION GASOLINE

ENGINE LIMITS: TAKEOFF (5 MIN) 2700 RPM MAX CONTINOUS POWER: 2600 RPM

PROPELLER: MCCAULEY D3A34C401/90DF A-8 3 BLADE PROPERLEN INDEGEN CESTIFICATION AND ASSESSED FOR THE PROPERTY OF THE PROPERTY O

PROPELLER LIMITS: 82 MAXIMUM 80" MINIMUM

INSTRUMENT MARKINGS:

TACHOMETER: RED LINE: 2700 RPM - MAXIMUM, YELLOW ARC: 2600 TO 2700 RPM (1850 TO 2150 RPM D3A34C401/90DF A-8 PROPELLER ONLY). CONTINOUS OPERATION BETWEEN 1850 AND 2150 RPM IS PROHIBITED. GREEN ARC: 2200 TO 2600 RPM -NORMAL OPERATING RANGE

CYLINDER HEAD TEMP GAGE: RED LINE: 460 DEG F. MAXIMUM WITH BAYONET THERMOCOUPLE

OIL TEMP GAGE: RED LINE 240 DEG F- MAX

OIL PRESSURE GAGE: RED LINE 10 PSI - MMINIMUM, RED LINE 100 PSI - MAXIMUM GREEN ARC 20 PSI TO 60 PSI - NORMAL **OPERATING RANGE**

OPERATORS HAND BOOK SUPPLEMENT: DATED: JUNE 21 2007 IS TO BE USED WHEN PPA 0-470-50 ENGINE IS INSTALLED IN THIS AIRCRAFT

PLACARD: THE FOLLOWING PLACARD SHALL BE PLACED IN VIEW OF THE PILOT WHEN THE 0-470-50 ENGINE AND MCCAULEY PROPELLER ARE INSTALLED. (AVOID CONTINOUS OPERATION 1850 AND 2150 RPM 24" MANIFOLD AND ABOVE

INSTALLATION OF THE P.PONK 0-470-50 ENGINE AND MCCAULEY D3A34C401/90DFA, 2A 34C58/90AT OR THE 2A 34C66/90AT PROPELLER COMBINATION IS APPROVED ON ALL CESSNA 182 H THRU P MODELS UNDER STC# SA 566 4NM.

6	
US Department of Transportets	

MAJOR REPAIR AND ALTERATION

Form Approved OMB No. 2120-0020 11/30/2007

For FAA Use Only

Administr	etion		antine See Title 14	CFR S	13.9, Part 43	Appendix B, and /	e to report o	an	result in a civil perially to	
INSTRU	ICTIONS: Prin	at or type all a	form. This report is rec	quired to	y law (49 U.S.	C. Swalth Lynn	Mittellate	_	ubsequent revision thereof) for result in a civil penalty for each	
such vic	dellar /60115	C \$46301(A	3)		ALCOHOLD DO NOT THE OWNER.	Sarial No.			6 2 2	
	National	ty and Regist	ration Mark			51676		-	Series	
1. Aircreft	N23760	5				Model 182			1828	
11.5.41.41.50	CESSN				VV.	There is about	wn on regist	ratio	on certificate)	
	Name (A	a shown on n	egistration certificate)			Arterna 24 LOVE	EL DIT	_	State CA	
	KOPAN	TABER L				City MILL VA	LLEY	_	country USA	
2. Owner						Zip 94941			outry	
				3. F	or FAA Use (Only		_		
		44	(e) T						C 41.896 W	
					74	90			*	
									4	
								_		
4.1	Туре			5. (Unit Identifica			_	Serial No.	
Repair	Alteration	Unit		Make			Model	120-		
		NOCOANE				(As described	in Item 1 ab	OVE)	
Ø		AIRFRAME								
: [POWERPLA	ANT							
	-									
		PROPELLE	R							
			Туре							
		APPLIANCE								
П			Manufacturer							
				6.0	onformity Sta	tomant		_		
	bloom and b	44		256535220	Kind of Agen			-		
	Name and A		C	1		ated Mechanic			Manufacturer	
	B WEST OL		CONTRACTOR OF THE PARTY OF THE	- 4	The second participation of the second	tificated Mechanic		C.	Certificate No.	
	NTAGUE	D INIDIO	State CA			Repair Station				
9608	34 cx	usy USA			Certificated				P 520707568 I.A	
			eration made to the	unit(s) i				on t	he reverse or attachments hereto	
have	been made in	accordance v	with the requirements	s of Par	t 43 of the U.S				and that the information	
fumisi	hed herein is t		ect to the best of my l							
ixtended ra	COMPANY OF THE PARK OF THE PAR		ignature/Date of Auth				11		10 110	
er 14 CFR	Part 43	R	OBERT E. STEEL	MAN/	JUNE 21 20	07	1 5	1	1	
pp. B						100	nes	(El rosa	
			7.	Appro	val for Retur	n to Service				
Pursuant	to the author	rity given p	ersons specified be	elow th	e unit ident	ified in item 5 v	was inspec	ted	in the manner prescribed by	
Administra	itor of the Fed	leral Aviation	Administration and is	8		✓ Approved	Reject	ed	ar the meaner precented by	
	AA Fit. Stand	ards	Manufacturer	N	laintananan C	Name -	Pers	ons	Approved by Canadian	
	nspector		viaridiacturer	14	laintenance C	rganization			ent of Transport	
Y "	ALL DESCRIPTION OF THE PARTY OF	11	Repair Station	X	nspection Aut	hodzatloc	Other (Sp	eci.	(y)	
	AA Designee				······································	THUTE CHILDE				
F										
		s	Signature/Date of Aut ROBERT E. STEE	horized	Individual	w)	,	7	- 11 1	

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

N2376G

Nationality and Registration Mark

8. Description of Work Accomplished ENGINE, PROPELLER AND COMPLETE FIREWALL ASSEMBLY WAS REMOVED DUE TO POOR

REPAIRS ON ORIGINAL FIREWALL AFTER HARD LANDING. INSTALLED NEW COMPLETE OEM FIREWALL ASSEMBLY PN # 0791301-2 REINSTALLED ENGINE MOUNT, ENGINE AND PROPELLER AFTER FIREWALL INSTALLATION. FIREWALL INSTALLED PER ORIGINAL RIVETING AND RIVET SPACING USING STANDARD RIVETING PRACTICES REFERENCE: AC43.13-1B CHAPTER 4 SECTION 4 PARAGRAPH 4-57 AND 4-58. PAGES 4-14 THRU 4-39 INSTALLED ENGINE MOUNT TO FIREWALL USING NEW RUBBER BUSHINGS, STRAPS, AND WASHERS. USED NEW HARDWARE IN ALL LOCATIONS.

INSTRUCTIONS FOR CONTINUED AIRWORTHINESS REQUIRE INSPECTION AT EACH ANNUAL PER APPENDIX D OF FAR PART 43.

-END-

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only
Office Identification

FAA + FIDD

	Make	(Section 901 Federal A			Model				
1. Aircraft	2	Cessna		1828					
	Serial No	51676			Nation	ality and Registration N2376G	on Mark		
	Name (As	shown on registration	certificat	0)	Addres	ss (As shown on reg	istration c	ertificate)	100
2. Owner		Kopan, Taber I				24 Lower I Mill Valle		94941	
				3. For FAA Use	Only				
Unit			1	4. Unit Identifica	tion			5. Type	
Oint	Make			Model		Serial No).	Repair	Alteratio
MRFRAME	~		(As desc	ribed in Item 1 abo	ove)				
OWERPLANT									
	Sauce #		.,						1000
OPELLER			1.1		87	RECEIVE	D		
	уре					JUL 1 n 200	16		
	arbureto			A-4-5	SF	ATTLE 15400	,0		
M	arvel Sc	hebler	P	N:10-4893-1		113年40			Х
Agency's Name	and Address		6.	Conformity States	nent				
Lchard M.		is .	- 1	3. Kind of Agency	,		C. Certif		
Ponk Avi	ation		13	- Continuedio	Mechanic		C. Certii	icate No.	
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mano lela	nd LIA	00000	. -	Certificated Rep	air Station				
			thousie	Manufacturer					
furnished here	de in accord ein is true an	/or alteration made to lance with the require d correct to the best of	ments of	Part 43 of the U.S	4 above a	and described on the	ne reverse	or attachmer	nts hereto
	1						ns and the	at the inform	ation
-052006			10	ignature of Autho	rized Indiv	vidual			and the same of
03-2006				Richard M.	Haines	Dichel	n.		
sugat to the	uthority sive		7. Appro	val for Return To	Service	Vecker	voan	_	
nount to the at	e Federal A	viation Administration	below, ti	ne unit identified	in item 4				
inistrator of th	44.00	Manufacturer	and is	APPROVED	□ RE	Was inspected in	the mann	er prescribe	ed by the
FAA Fit. St. Inspector	andards	manufacturer	v lins	Dection Author	Tarita	Umer / Spanifus			
FAA Design	nee		X Ins	pection Authorizat		Other (Specify)	1		
	nee	Repair Station Certificate or	X Ins	pection Authorization son Approved by nada Airworthines:		Other (Specify)			

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

escription	of Work Accomplished ace is required, attach a	dditional sheets.	Identify with air				
I More sp						Engine	
	Modified MA-4-5 of Report No. 2, Revi use with PPA O-47	n-50 Engine S	SN:2440.			O-470-50 Engine C #SE4985NM for	
	Installation of the	invicator on t	he engine and	aircraft not inclu	ided.	3	
	Installation of the	1		End			
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MAJOR REPAIR AND ALTERATION

Form Approved OMB No. 2120-0020

US Departer of Transport Federal Avi Administra	Rottlern	Airframe, Powerp	olant,	Proj	seller, or Ap	pliance)	Nº 8	ffice Iden	diffication OALL	
and district	Oplition of this to	or type all entries. See Fi orm. This report is requir (Section 901 Federal Av	red twit	aw (ah	F1 52 CV 1 48 11 L.	and AG 43.9 allure to rep	1-1 (or subsequent port can result in a	revision t civil pena	hereof) for in Ity not to exce	ed \$1,000
	Make	CESSNA	30			Model	182B			
1. Alrereft	Serial No.						y and Registration	n Mark	,	
	Name (As	51676 shown on registration of				Address (As shown on regis	stration o	ertificate)	
2. Owner		KOPAN TABER	R L	X.			24 LOWER MILL VAL	DR ·		141
				3.	For FAA Use Or	nly				
			200 000			*				
	_			4.	Unit Identificati	on			5. Type	
Unit		Make			Model		Serial No.		Repair	Alteration
AIRFRAME	-		(As des	scribe	d in Item 1 abov	e)			х	
POWERPLANT		*								
PROPELLER							*			
	Туре									
APPLIANCE	Manufacturer									
			6	6. Cor	nformity Statem	ent				
. Agency's Na					Cind of Agency			C. Cert	ificate No.	
NICHOLAS DBA WEST				x	U.S. Certificated					
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JKIAH, C	A 95482	2			Certificated Repo					
TOTTIISHED 13	t the repair and made in accom erein is true a	d/or alteration made to dance with the require nd correct to the best	the uni	it(s) id	dentified in item	4 above ar	nd described on the Aviation Regulation	he reversions and t	e or attachm hat the infor	ents hereto
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			7. Apr	prova	for Return To	Sanda				
dministrator o	authority give the Federal A	ven persons specified Aviation Administration	below in and i	, the	unit identified	in item 4	was inspected in	the ma	nner prescri	bed by the
FAA Fit.	. Standards	Manufacturer	10		ction Authorizat		Other (Specify)			Jy tile
FAA Des	5.TAX (4.00)	Repair Station		Perso	n Approved by	Fransport S Group				
e of Approval or 20-99	Hejection	Certificate or Designation No. 569539512	8	Signa	ture of Author	ized Indivi	dual Philosophia	0		

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all provides a requirements. compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

(If more space is required, attach additional sheets, identify with aircraft nationality and registration mark and date work completed.) 8. Description of Work Accomplished

> REPLACE UPPER AND LOWER CORRUGATED SKINS ON LH FLAP P/N 0523901 USING PROCEDURES IN CESSNA 100 SERIES SERVICE MANUAL 1962 & PRIOR SEC.19(STRUCTURAL REPAIR) AND AC 43.13-1B CHAP.4 SEC.4(METAL REPAIR PROCEDURES).

SUPPORT FLAP IN FIXTURE AND REMOVE UPPER AND LOWER SKINS I.A.W. AC 43.13-1B 4-52b.

NEW SKINS FABRICATED FROM 2024T3 .020" ALCLAD ALUMINUM. SAME AS REMOVED I.A.W. AC 43.13-1B 4-53 4-55a.b.c. 4-56a.b.

INSTALL TRAILING EDGE BULB FROM AIR METAL FABRICATORS I.A.W. S.T.C. #SA00406WI.

RIVET ASSEMBLY I.A.W. AC 43.13-1B 4-57a.b.c.d. ----END-----

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

WP-Z7 and ix II, and AC 43.9-1 (or subsequent revision thereof) for instructions

Alrema		report is required by 901 Federal Aviation		3 Appendix B, and AC 9, U.S.C. 1421). Failure to 1958). Model		182B				
Records	Make CES	SNA		Nation	ality and	d Registration M	lark			
	Serial No.			A STATISTICS	100000000			ettlente)		
	516	76		Addre	s (As s	hown on registra	ation cer	grundata)		
	Name (As shown o	on registration certific	cate)		2	A LOWER I)R			
Owner	FOI	AN TABER L			M.	ILL VALLE	3Y, C	94941		
	1			For FAA Use Only			-			
				Unit Identification				5. Type		
			4	Model		Serial No.		Repair	Alteration	
Unit	Ma		-							
AIRFRAME		(As	descri	bed in Item 1 above) 🕶	******			X	-	
POWERPLA	NT .									
PROPELLES				4						
	Туре			· 100						
APPLIANCE	Manufacturer			* .						
			6.	Conformity Statement						
A. Agenci	s Name and Address			B. Kind of Agency		•	C. Cer	tificate No.		
NICHO	LAS EDWARD I	BISHOP	77.1	X U.S. Certificated Med			560	539512		
DBA W	EST COAST W	INGS		Foreign Certificated		c ·.	1 303	333312		
2880	EASTSIDE RD		1	Certificated Repair S	ation		1			
DKIYH	, CA 95482			Manufacturer						
have	tify that the repair and/ been made in accord ished herein is true an	ance with the require	ements	t(s) identified in item 4 a of Part 43 of the U.S. Fo nowledge.	bove an deral A	id described on t viation Regulati	he rever ons and	se or attachi that the info	nents heret ormation	
Date	8-20-99	4		Signature of Authorize	d Indivi	dual				
			7 .	2011	1 16	300				
Pursua	nt to the authority give	ven persons specifie	d belov	proval for Return To Se	Item 4	was inspected	in the m	nanner pres	cribed by t	
Admini	FAA Fit. Standards	Aviation Administrati	ion and	IS XXAPPROVED	O RE	Other (Specify				
ву	Inspector	Manufacturer	X	Inspection Authorizatio		-				
	FAA Designee	Repair Station Certificate or		Person Approved by Tr Canada Airworthiness Signature of Authoriz	iroup					

8-20-99

569539512

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to accurate a standard provides alterations to accura vergination balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

(If more space is required, attach additional sheets, identify with aircraft nationality and registration mark and date work completed.) 8. Description of Work Accomplished

REPLACE DAMAGED LEADING EDGE SKIN P/N 0720601-16 AND UPPER AND LOWER SKINS FROM STA. 190.0-208.0 ON RH WING USING PROCEDURES IN CESSNA 100 SERIES 1962 & PRIOR SERVICE MANUAL SEC.19(STRUCTURAL REPAIR) AND AC 43.13-1B CHAP.4 SEC.4(METAL REPAIR PROCEDURES).

REMOVE DAMAGED SKINS BY DRILLING I.A.W. AC 43.13-1B 4-52b.

FABRICATE NEW UPPER AND LOWER SKINS AND LEADING EDGE SKIN FROM 2024T3 ALCLAD ALUMINUM, SAME AS REMOVED I.A.W. AC 43.13-1B 4-53 4-55.

REINFORCE DAMAGED AUXILLIARY SPAR SECTION BETWEEN STA. 190.0-208.0 USING CESSNA SERVICE MANUAL SEC.19 FIG.19-7.

REPLACE DAMAGED END RIB P/N 0720608-7 WITH NEW.

RIVET I.A.W. AC 43.13-1B 4-57a.b.c.d.f -----END-----

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification WS

Federal A Administ	Awletion					D AC 43.9-	1 (or subsequent	revision t	hereof) for ins	tructions
INSTR and dis	Aviation tration RUCTIONS: Print or type sposition of this form. Ti ch such violation (Section	all entries. See FAR on the report is required to 901 Federal Aviation	ta.s. F/ by law on Act	AR 43 (49 U of 19	Appendi 9.G. 142 68).	1). Failure to rapo	ort can result in a	civil pena	ty not to exce	60 \$ 1,000
THE SHAPE	Make						182B			
1. Airora	C	ESSNA				Nationality	and Registration	n Mark		
. Airera	Serial No.	1676	X.				N2376G		- stiffents)	
	3	1676 n on registration cert	ificate)		Address (/	As shown on regi	stration o	ermicate)	
	Name (As snow	OPAN TABER	L				24 LOWER	T.EY.	CA	
2. Owne	st K	OPAN MARLYS	; A				MILL AND		94941	
					or FAA L	Ino Only				
									5. Type	
		. 1		4. (Model	ification	Serial No	o.	Repair	Alteration
Uni	it N	take		_	Mode				-	
AIRFRAM	Æ	(A	ls desc	ribed	in Item	1 above)			x	
POWERP	LANT	10								
PROPELL	LER									
APPLIAN	Type CE Manufacturer									
						1				
				1		Statement		Tc c	rtificate No.	
	cy's Name and Address			X	Kind of	rtificated Mechanic		0. 00	minicate No.	-
	HOLAS EDWARD WEST COAST W			1	- Inches	Certificated Mecha	Tokulus and the	569	9539512	
	EASTSIDE RE			-		ated Repair Station		7		
	H, CA 95482				Manufa	cturer				
hav	ertify that the repair and, we been made in accord nished herein is true an	ance with the requir	ement:	s of F	art 43 of	d in item 4 above the U.S. Federal	and described or Aviation Regula	n the reve	rse or attachr i that the info	nents hereto rmation
Date				Sig		of Authorized Ind	2 2 2 30 30 30 300			
8-20)-99	,		L	2	an Bus	Sen()		
0	and to the early of				al for Re	turn To Service	0			
Admin	ant to the authority gives istrator of the Federal A	en persons specifie Aviation Administrati	on and	w, th	e unit ic	fentified in item PROVED	4 was inspected REJECTED	in the n	nanner presc	ribed by the
3Y -	FAA Fit. Standards Inspector	Manufacturer	х	Ins	pection A	authorization	Other (Special	(y)		
	FAA Designee	Repair Station		Ca	nada Airv	roved by Transpor vorthiness Group				
8-20	pproval or Rejection -99	Certificate or Designation No. 56953951	2	Sig		Authorized Ind	Account to the second s	0		

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.) 8. Description of Work Accomplished

REPLACE DAMAGED UPPER AND LOWER OUTBOARD SKINS ON RH AILERON P/N 0523800-17 USING PROCEDURES IN CESSNA 100 SERIES SERVICE MANUAL 1962 & PRIOR SEC.19(STRUCTURAL REPAIR) AND AC 43.13-1B CHAP. 4 SEC.4(METAL REPAIR PROCEDURES).

SUPPORT AILERON IN FIXTURE AND REMOVE UPPER AND LOWER OUTBOARD SKINS I.A.W. AC 43.13-1B 4-52b.

FABRICATE NEW SKINS FROM 2024T3 .016" ALCLAD ALUMINUM, SAME AS REMOVED I.A.W. AC 43.13-1B 4-53 4-55a.b.c. 4-56a.b.

RIVET SKINS I.A.W. AC 43.13-1B 4-57a.b.c.d. -----END-----



Of Transportation
Federal Aviation

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 and disposition of this form. This report is required by law (49 U.S.C. 1421).

TOT BOOT	Make	essna				Model	1	L82B			
I. Aircraft	Serial No.	1676				Nationa		d Registration 12376G	n Mark		
		n on registration cert	ificate	,		Address		hown on regi		ertificate)	
2. Owner	Y	COPAN TABER	L					24 LOWER			
										94941	
				3. i	For FAA L	Jse Only			-		
		*	*			*	,	*			
				4. 1	Unit Ideni	ification				5. Type	
Unit		Make		*	Model			Serial No.		Repair	Alteration
AIRFRAME		(A	s desc	cribe	d in Item	1 above)	••••			х	
POWERPLA	NT ·					3	•				
PROPELLER	2										
APPLIANCE	Туре										
	Manufacturer					2					
6 Accom	fo blome and biddens	•	6	1		Statement					
	S Name and Address CLAS EDWARD	And the second s		В.	Kind of A				C. Certif	icate No.	
	VEST COAST V			X	-	ificated Mechanic Certificated Mech	-	-	5695	39512	
2880	EASTSIDE RI).		2310		ed Repair Station					
	I, CA 9548;				Manufac	urer					
D. I cert have furnis	ify that the repair and, been made in accord shed herein is true an	for alteration made to ance with the require d correct to the best of	the un ments of my l	it(s) of P	identified art 43 of t	in item 4 above he U.S. Federa	and d	escribed on th tion Regulatio	e reverse	or attachme	nts hereto
Date	8-20-99	*		-		Authorized Ind	ividua	0			
			7 40	Drov	al for Det	- 02	_	6			
Pursuant	t to the authority giverator of the Federal A	en persons specified viation Administratio	belov n and	v, the	e unit ide	ntified in item	4 was	inspected in	the man	ner prescrib	ed by the
	FAA Fit. Standards Inspector	Manufacturer	x		V CITE	thorization		her (Specify)			
200	FAA Designee	Repair Station		Pers	on Anne	ved by Transpor	t	4 1			
	proval or Rejection	Certificate or	_	SECTION S.		Authorized Indi					

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

(If more space is required, attach additional sheets, identify with aircraft nationality and registration mark and date work completed.) 8. Description of Work Accomplished

> REPAIR DAMAGED RH ELEVATOR P/N 0732000-154 AS FOLLOWS: USING PROCEDURES IN CESSNA 100 SERIES SERVICE MANUAL 1962 & PRIOR SEC.19(STRUCTURAL REPAIR) AND AC 43.13-1B CHAP. 4 SEC.4(METAL REPAIR PROCEDURES).

SUPPORT ELEVATOR IN FIXTURE AND REMOVE UPPER AND LOWER SKINS, LEADING EDGE SKIN, AND SPAR I.A.W. AC 43.13-1B 4-52b.

REPLACE THE FOLLOWING PARTS WITH NEW: LEADING EDGE SKIN P/N 0734600-5 SPAR P/N 0734601-3 STIFFENER P/N 0734605-1

NEW UPPER AND LOWER SKINS FABRICATED FROM 2024T3 .016" ALCLAD ALUMINUM, SAME AS REMOVED I.A.W. AC 43.13-1B 4-53 4-55a.b.c. 4-56a.b.

RIVET ASSEMBLY I.A.W. AC 43.13-1B 4-57a.b.c.d.f. -----END-----



US Department of Transportation

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 90). Federal Aviation Act of 1959). for each such violation (Section 901 Federal Aviation Act of 1958). Model 182B Make Nationality and Registration Mark CESSNA N2376G 1. Aircraft Serial No. Address (As shown on registration certificate) 51676 Name (As shown on registration certificate) 24 LOWER DR KOPAN TABER L MILL VALLEY, KOPAN MARLYS A 94941 2. Owner 3. For FAA Use Only 5. Type 4. Unit Identification Alteration Repair Serial No. Model Make Unit X (As described in Item 1 above) AIRFRAME POWERPLANT PROPELLER Type APPLIANCE Manufacturer 6. Conformity Statement C. Certificate No. B. Kind of Agency A. Agency's Name and Address U.S. Certificated Mechanic x 569539512 NICHOLAS EDWARD BISHOP Foreign Certificated Mechanic DBA WEST COAST WINGS Certificated Repair Station 2880 EASTSIDE RD. Manufacturer ПКТАН, CA 95482 D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge. Signature of Authorized Individual Date 8-20-99 7. Approval for Return To Service Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED

REJECTED **APPROVED** Other (Specify) FAA Fit. Standards Inspector Manufacturer X Inspection Authorization BY Person Approved by Transport Canada Airworthiness Group Repair Station **FAA Designee** Certificate or Designation No. Signature of Authorized Individual Date of Approval or Rejection 569539512

8-20-99

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the restriction of the continued conformity with the continued conti weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

Description of Work Accomplished
(If more space is required, attach additional sheets, identify with aircraft nationality and registration mark and date work-completed.) 8. Description of Work Accomplished

> REPLACE UPPER AND LOWER CORRUGATED SKINS ON RH FLAP P/N 0523901-1 USING PROCEDURES IN CESSNA 100 SERIES SERVICE MANUAL 1962 & PRIOR SEC.19(STRUCTURAL REPAIR) AND AC 43.13-1B CHAP.4 SEC.4(METAL REPAIR PROCEDURES).

SUPPORT FLAP IN FIXTURE AND REMOVE UPPER AND LOWER SKINS I.A.W. AC 43.13-1B 4-52b.

NEW SKINS FABRICATED FROM 2024T3 .020" ALCLAD ALUMINUM, SAME AS REMOVED I.A.W. AC 43.13-1B 4-53 4-55a.b.c. 4-56a.b.

INSTALL TRAILING EDGE BULB FROM AIR METAL FABRICATORS I.A.W. S.T.C. #SA00406WI.

RIVET ASSEMBLY I.A.W. AC 43.13-1B 4-57a.b.c.d. ----END-----



MAR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

WP-27 Mea

Federal Aviation	in .			10 A	epondix B	and AC 43.	9-1	(or subsequent r	evision th	nereof) for ins	structions ed \$1,000
INSTRUCT and disposi for each su	IONS: Print or typ tion of this form. T ch violation (Sect	pe all entries, See FAR 43.9, This report is required by lav tion 901 Federal Aviation Ac	w (49) at of	U.S. 1958	.C. 1421). I	Model		t can result in a c	ivil penai	ty not to o	
	Make					182					
1. Aircraft	Cess	sna			600	National	ity a	and Registration	Mark		
Mr.	Serial No. 5167	76				N2:	145	shown on regis	tration ce	ertificate)	
	Nama / As show	wn on registration certificat	(e)			DA T	OTO	er Drive			
2. Owner	Кора	an Construction	1			Mill	. V	alley, C	A 949	41	
			3.	For	FAA Use	Only					
•			4.	Uni	it Identifica	ation	-			5. Type	
Unit		Make			Model			Serial No.		Repair	Alteration
AIRFRAME		(As des	scribe	ed in	n Item 1 ab	ove)					х
POWERPLANT			i e								*
PROPELLER											
APPLIANCE	Type Manufacturer										
			6. C	onfo	ormity Sta	tement	_			L	
	Vame and Address	s	-		nd of Agen				C. Certi	ificate No.	
James P		The state of the s	Market Street	U	J.S. Certifica	ated Mechanic	_			mouto ito.	
2240 Ai	Aviation rport Blv	Maintenance	_	10000	The second secon	ificated Mech					
Santa P	losa, CA 9	5403	-	1000	Name and Address of the Owner, where	Repair Station	n				
D. I certify to have been furnished	hat the repair and in made in accord d herein is true ar	d/or alteration made to the u dance with the requirement and correct to the best of my	unit(s)) ide	Manufacture entified in it 43 of the		and al Av	d described on the		77444 e or attachme hat the inform	ents hereto
Date 04/20		- Control of the beat of the	A MINATES		7.00	thorized Ind	_				madon
52/2	3/36			7	1/0	_	1	X			
Pursuant to	the authority gir	ven persons specified but	ppro	VAL	lór Return	To Service		9			
Administrato	or of the Federal	ven persons specified belo Aviation Administration an	ow, pr	he u	Init identif	ied in item	4 v	was inspected in	the mai	nner prescrit	had by the
1 1000	A Fit. Standards Dector	Manufacturer	100		tion Autho		HEJ	Other (Specify)			bed by the
FAA	Designee	Repair Station	Per	rson	Approved	by Transpor	rt				
04/20	val or Rejection	Certificate or Designation No. 555277444	Sig	gnati	ure of Aut	horized Ird	livid	lual			
AA Form 337				-	Afra			2			

Weight and balance or operating limitation changes shall be entered in the appropriate arcraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

good shop practi	END		



MAJOR REPAIR AND ALTERATION. (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

101 61	Make		DON SOT FEDERAL AVIAN			Model	, 0on			
1. Alrei	TRIX Promountment	ess	na ·			Nations	182B ality and Registration	on Mark		
	Serial No.	167	6			radione	N2376G	on man		
2. Own	Name (As	shor	wn on registration certi er Air Service		te)	Address	Unit 3 Box Casper, Wy.	3, Air	port	
					3. For FAA Use C	only				
			٠							
					4. Unit Identificat	ion			5. Type	
Un	it	1	Make		Model		Serial No).	Repair	Alteration
AIRFRAN	ME .		(As	s des	cribed in Item 1 abo	ve)				xx
POWERP	LANT				n √ n#.					
PROPELL	ER									
APPLIAN	Type Manufactu	rer								
		_								
A. Agend	cy's Name and Ad	dress		- 0	B. Kind of Agency					
Ca	sper Air Se	rvi	ce		U.S. Certificate			C. Certi	ficate No.	
Un	it 3 Box 3,	A1	rport		Foreign Certific					
Ca	sper, Wy				XX Certificated Rep	pair Station	ATTIC .	CBCR7	'33C	
D. I cer	rity that the	826	S. Company of the Com		Manufacturer					
have furn	been made in ac ished herein is tru	corda e and	or alteration made to the ence with the requirement correct to the best of	ne un ients my k	it(s) identified in iter of Part 43 of the U.S knowledge.	n 4 above : S. Federal	and described on t Aviation Regulation	he reverse	or attachmen	nts hereto
0.000000111	muary 3, 19				Signature of Author Troy Eskew	orized Indi	vidual 7	Est	1	
D			7.	. App	proval for Return To	Service	1.	roy	e e	
Adminis	trator of the Fede	give	n persons specified b	elow	, the unit identified	in item				
,	FAA FIt. Standards Inspector		n persons specified b	and i	Inspection Authoriza	OR	Other (Specify)	the man	ner prescrib	ed by the
	FAA Designee	xx	Repair Station		Person Approved by	T				
ite of App	oroval or Rejection	1	Certificate or Designation No.	-	Signature of Author	ss Group rized Indiv	ridual ,	(.)	1	
Jellos	337 (12-88)	_	CBCR733C	\Box	William Howe	ell, In	spector	11.	A	00
							-ac	- will	100	ell



Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.) N2376G Cessna 182B, S/N 51676

1) Installed Cleveland wheels P/N 40-97A, and brakes P/N 30-63A in accordance with S.T.C. SA13GL and Cleveland

Drawings 20-134, Rev A, dated October 12, 1972, and Mounting Instructions 50-26, Rev C, dated February 2, 1972.

2) Brackett Induction Air Filter assembly P/N BA 8110 was previously installed by persons unknown. The installation was inspected and determined to a second and determined to a s spected and determined to conform to S.T.C. SA71GL.

----- END -----

3) Weight and Balance change is negligable. Updated Aircraft Equipment List.

D Additional Sheats Are Attached

FEVERAL AVIATION AGENCY

MAJOR REPAIR AND ALTERATION

Form Approved Budget Bureau No. 04-R060.1 FOR FAA USE ONLY OFFICE IDENTIFICATION

(Airframe, Powerplant, Propeller, or-Appliance)

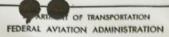
	(Allifallio	,	WAR 43 0 FAR 43 Ap	pendix B, and AC 43.9-1	(or subsequent re	vision th	ereof)
INSTRUCT	TIONS: Print or ions and disposi	type all entries. See	FAR 43.5, FAR 45 1-F	MODEL.			
	MAKE	Cessna			182	ov.	
1. AIRCRAFT	SERIAL NO.			NATIONALITY AND RE	7		
		51676	Cardal	ADDRESS (As shown on	registration certific	ate)	
2. OWNER	Warren H	on on registration certification certification and an arrangement of the control	nd Eleanor V.		to Avenue		
	Buckingh	lam	3. FOR FAA USE O				
		4. UI	NIT IDENTIFICATION			5. 1	YPE
UNIT		MAKE	MODEL	SE	RIAL NO.	REPAIR	ALTER- ATION
AIRFRAME	••••	•••••	(As described in item 1 c	bove)	*****	x	
POWERPLANT							
PROPELLER							
APPLIANCE -	MANUFACTURER						
Α. Α	GENCY'S NAME	E AND ADDRESS	6. CONFORMITY STATE				
Hank Hi	lsmann		B. KI	ND OF AGENCY	C. CERT	IFICATE	NO.
Nation	Flight Se	ervice, Inc.	X U.S. CERTIFIC				
2444 AL	rport Bla	7d -		IFICATED MECHANIC REPAIR STATION	1939	831	
Santa R	osa, Ca.	95401	MANUEACTUR	n .			
D. I certify to attachment and that t	that the repair a ts hereto have to the information	and/or alteration ma been made in accorda furnished bassis is	de to the unit(s) idea	41G . 1	and described on	the revi	erse or
ATE			the and correct to the	best of my knowledge.	redetat Aviati	on Regu	lations
	3, 1975		SIGNATURE OF	AUTHORIZED INDIVIDU	AL		
		7 ,	-N- D.	Nelsman			
Pursuant to th	ne authority give	en persons specifical	PPROVAL FOR RETURN	O SERVICE			
FAA FI	T. STANDARDS	eral Aviation Agency	and is APPROVI	O SERVICE ed in item 4 was inspected REJECTED	ted in the manne	r prescri	bed by
INSPEC	TOR	MANUFACTURER	X INSPECTION AUTHOR	OTHER (Specif			
	ESIGNEE	REPAIR STATION	CANADIAN DEPARTA OF TRANSPORT INS	AENT PECTOR			
TE OF APPRO	OVAL OR il 3, 75	CERTIFICATE OR DESIGNATION NO 1939831	SICHIATION	AUTHORIZED INDIVIDU	JAL		
A Form 33	7 (1-65) OBS	OLETE PREVIOUS ED	ITION	S. Klama			

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.) Strengthened firewall by use of two "HAT" section stiffeners running

vertically between nose gear support doubler and angle between top engine mounts. -----END-----



Form Approved Budget Bureau No. 04-R060.1 FOR FAA USE ONLY

OFFICE IDENTIFICATION

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

MAKE

OPCONT.

(Airframe, Powerplant, Propeller, or Appliance)

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

MODEL

182

1. AIRCRAFT	SERIAL NO.	76			N2376G	STRATION M	ARK	
2. OWNER	NAME (As sho Warren	wn on registration certific Buckingham & P V. Buckingh	am	ADDRESS 39	(As shown on re 80 Monted nta Rosa,	ito Ave	ficate) nue 5404	
				ON THE OSE ONE				
		4. UNI	T IDENT	IFICATION			5.	TYPE
UNIT		MAKE		MODEL	SERIAL	. NO.	REPAIR	ALTER-
AIRFRAME	****	(A	As descri	bed in item 1 above)	************	•••		х
POWERPLANT								
PROPELLER								
APPLIANCE	MANUFACTURER							
	ACTIONS III		6. CON	FORMITY STATEMENT				
	S. HILSM	ANN		B. KIND OF AG X U.S. CERTIFICATED MECHAN FOREIGN CERTIFICATED MEC CERTIFICATED REPAIR STATIC	IC HANIC	C. CER	19398	
D. I certify attachmen	that the repair nts hereto have the information	and/or alteration made been made in accordan furnished herein is to	e to the	manufacturer c unit(s) identified in ite the requirements of Part correct to the best of my	m 4 above and 43 of the U.S. I	described on ederal Aviari	the rever	rse or
ATE	21, 1973		SIC	SNATURE OF AUTHORIZE	D INDIVIDUAL	•		
		7. AP	PROVAL	FOR ARETURN TO SERVICE				
Pursuant to t	the authority giverator of the Federat	ven persons specified be eral Aviation Administra	low, th	e unit identified in item 4 id is	was inspected in	the manner	prescribe	ed by
Y FAA F	LI. SIAHDARDS	MANUFACTURER	X INS	PECTION AUTHORIZATION	OTHER (Specify)			
and the same	DESIGNEE	REPAIR STATION	OF OF	VADIAN DEPARTMENT TRANSPORT INSPECTOR AIRCRAFT				
	21, 1973	CERTIFICATE OR DESIGNATION NO. 1906206 I.		PATURE OF AUTHORIZE	D INDIVIDUAL Miels	4.		
A Form 33	37 17 47			any or le	- Mels	ac .		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

REMOVED STARTER P/N 109471 & REPLACED WITH HEAVY DUTY STARTER USED ON LATER MODELS P/N 1108249 (Note: The starter installed is an overhaused unit from Ward Aero Repair Station 3779) RELOCATED STARTER SOLENOID ON FIREWALL.

ITEM	WGT	ARM	MOMENT
A/C	1750	40.7	71347
Original Starter Original Solenoid	-14 =1	-4 -4	56 4
Replacement Starter Replacement Solenoid	18	-4 -1	-72 -1
	1754		71334

New Empty Wgt 1754 lbs.

New Useful Load 896 lbs.

Negligable change in C.G = 40.7 inches aft of datum

NATION FLIGHT SERVICE, INC. 2244 Airport Blvd. Santa Rosa, Calif. 95401

Work Order No. 1660

PEDERAL AVIATION AGENCY

MAJOR REPAIR AND ALTERATION Airframe Powerplant, Propeller, or Appliance)

Form Approved Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

for instruc	TIONS: Print or type all entries. Sections and disposition of this form.	The same of the sa			
	MAKE	MODEL	TO MAKE THE PARTY OF THE PARTY		
1. AIRCRAFT	Ceasns	11.70	182 VALITY AND REGISTRATION	MARY	-
. AIRCRAFT	SERIAL NO.	NATION	NALITY AND REGISTRATION	WAKK	
	51876	LEN2	3760 S (As shown on registration o	estificate)	
	NAME (As shown on registration certifi				
2. OWNER	Warren Buckingham &	398	O Montecito Ave	aue	
	Eleanor V. Buckinghe	3. FOR FAA USE ONLY SOM	ta Rosa, Calif.	95404	
		airworthiness requirement described aircraft subj person authorized in F	d herein complies with ap nts and is approved only f ect to conformity inspecti AR 43.7	or above	
	4. UN	IT IDENTIFICATION 200 40	SAME - 7	5.	TYPE
UNIT	MAKE .	MODEL 4-0-	SERIAL NO.	REPAIR	ALTE
					ATIO
AIRFRAME	***************************************	As described in item 1 above)	*************		
					v
OWERPLANT					X
JIERITANI					
ROPELLER					
	TYPE				
PPLIANCE	MANUFACTURER				
ľ	- Indiana				
		4 CONFORMER STATEMENT			
A. AC	SENCY'S NAME AND ADDRESS	6. CONFORMITY STATEMENT			
NATION	FLIGHT SERVICE, INC.	B. KIND OF AC	GENCY C. (ERTIFICATE	NO.
2	244 Airport Blvd.	FOREIGN CERTIFICATED MECHAI	NIC		
Sant	a Rosa, Calif. 95401	CERTIFICATED REPAIR STATE			
		MANUEACTURER			
. I certify the	it the repair and/or alteration mad-		I-3A	15567	90
and that the	at the repair and/or alteration made hereto have been made in accordan information furnished herein is tru	ce with the requirements of Part	em 4 above and described	on the reve	erse o
TE .	e information furnished herein is tru	and correct to the best of my	knowledge.	riation Regu	lation
		SIGNATURE OF AUTHORIZ	D INDIVIDUAL		
	70	1 11 . 11 1/1			
8 05	12	Nover (1. 84)	eaken .		
8-25-	7 ADI	POVAL COD DETUN			
	authority given persons specified to	PROVAL FOR RETURN TO SERVICE			
	7. API authority given persons specified be or of the Federal Aviation Agency ar	PROVAL FOR RETURN TO SERVICE	4 was inspected in the ma	nner pro-	
rsuant to the Administrato	or of the Federal Aviation Agency as	div. the unit identified in item	4 was inspected in the ma	nner prescri	bed by
	or of the Federal Aviation Agency are	PROVAL FOR RETURN TO SERVICE TOW, the unit dentified in item and is APPROVED INSPECTION AUTHORIZATION	4 was inspected in the ma	nner prescri	bed by
rsuant to the Administrato	or of the Federal Aviation Agency at STANDARDS MANUFACTURER	inspection authorization Canadian Department	4 was inspected in the ma	nner prescri	bed by
FAA PESI	or of the Federal Aviation Agency at MANUFACTURER GNEE REPAIR STATION	INSPECTION AUTHORIZATION CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	OTHER (Specify)	nner prescri	bed by
FAA FLT.	or of the Federal Aviation Agency at MANUFACTURER GNEE REPAIR STATION AL OR CERTIFICATE OR	INSPECTION AUTHORIZATION CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	OTHER (Specify)	nner prescri	bed by
FAA FIT.	or of the Federal Aviation Agency at STANDARDS MANUFACTURER GREE REPAIR STATION AL OR CERTIFICATE OR DESIGNATION NO	INSPECTION AUTHORIZATION CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR	OTHER (Specify)	nner prescri	bed by

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements. Identify with air-

B. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets-craft nationality and registration mark and date work completed.)

Installed cee Bailey 1/4" tinted winshield. with new felt and weatherstrip. Center strimanufactured under C.A.A., P.M.A., C.A.R., winshield. Winshield installed Center strip removed. Winshield ., C.A.R., 1.55

JUL 8 - RECO

Form approved. Budget Bureau No. 04-R000.

FEDERAL AVIATION AGENCY

MAJO	R REPAIR AND ALTERATION	MODE	182	21010	14 55		REGISTRATION MARK	
1. AIRCRAFT	Cessna NAME (First, widdle, iost) Peridenford, II	nc.		Napa County Airpo	ort			
2. OWNER	TE ONLY FOR UNIT REPAIR CE WITH CIVIL AERONAUTIO		· TERED. D	ESCRIBE WORK ACCOMP	LISHED C	N REV	ERSE IN AC-	
. COMPLE	TE ONLY FOR UNIT REPAIR CE WITH CIVIL AERONAUTION	ED AND/OR A	LIERED		; NA	TURE OF	WORK (Check)	
	MAKE ;	,	MODEL	SERIAL NO.	MAJOR REPAIR MAJOR ALTERAT			
UNIT		· · · · · · · · · · · · · · · · · · ·	n item I above)	***************		;		
a. AIRFRAME			10000 D T 4		x			
, POWERPLAN	ANT Continental 0470-I		4	98133-0-L-1	+-			
e. PROPELLER					-			
	TYPE AND MANUFACTURER							
. APPLIANCE				ompleted by repair or alter			lowever, in th	
	ER the repairs and/or alterations described below were made. (EMPTY WEIGHT (F	if applica	able.	ompleted by repair or alter onent, it will not be compl- At this time, it will be co CENTER OF GRAVITY (Inches from dat			FUL LOAD (Pounds)*	
Normal	CATACON.			35.1			959	
. CONFORM	MITY STATEMENT (Complete an	d check)				CERTIFICA	TE NO.	
. AGENCY'S NAM			b. KIND OF A	Certificated Mechanic.	e-	JULIFIC		
1901 Em	n Aviation, Inc. mbarcadero Road, to, California	U.S. Forei		#57				
				tion procedures.)				
I certify t attachmen the inform	hat the repair and/or alteration to have been made in a sation furnished herein is true 6/60 air and/or alteration completed) L FOR RETURN TO SERVICE	and correct to	the best of r	entified under item 3 abovements of Part 18 of the U. ny knowledge (Signature of author)	1	92	on the reverse	
I certify t attachmen the inform (Date repe APPROVAL Pursuant te	ts hereto have been made in a ation furnished herein is true	E (Check and comp) w the unit iden	the best of r	entified under item 3 abovements of Part 18 of the U. my knowledge (Signature of authoritems)	rized individ	E/C ual)		
I certify t attachmen the inform 7/ (Date reps. APPROVAL Pursuant t trator of the property of	ts hereto have been made in a ation furnished herein is true 6/60 air and/or alteration completed) L FOR RETURN TO SERVIC to the authority specified below	E (Check and complete with the unit identity of is	lete appropriate in tified in item	(Signature of authors) Canadian Departmen Cher (Specify)	manner pr	ual) rescribe	d by the Admi	
Date repersion of the inform the inform (Date repersion APPROVAL) APPROVAL Pursuant trator of the inform trator of the information of the informa	ts hereto have been made in a ation furnished herein is true 6/60 If and/or alteration completed) L FOR RETURN TO SERVIC to the authority specified below the Federal Aviation Agency at PROVED BY FAA Designation FAA Flighted Standards Inspector	E (Check and complete with and is gnee Man	lete appropriate in tified in item	entified under item 3 abovements of Part 18 of the University knowledge (Signature of authoritems) m 3 was inspected in the income and items of the i	manner pr	ual) rescribe	d by the Admi	
I certify to attachmenthe inform 7/ (Date repersion of the trator of the	ts hereto have been made in a ation furnished herein is true 6/60 Air and/or alteration completed) L FOR RETURN TO SERVICE to the authority specified below the Federal Aviation Agency at the Federal Aviation Agency	E (Check and complete with and is gnee Man	lete appropriate intified in iter	contified under item 3 above ments of Part 18 of the University knowledge (Signature of authorized in the 18 of the University) Canadian Department Other (Specify) (Signature of authorized individual apparentment) Other (Specify) (Signature of authorized individual apparentment) Other (Specify) (Signature of authorized individual apparentment) Other (Specify) Other (Sp	manner print of Transal; title or ide	rescribe sport Ir	d by the Admi	
Date reper Control of the Control of	ts hereto have been made in a ation furnished herein is true 6/60 Air and/or alteration completed) L FOR RETURN TO SERVIC to the authority specified below the Federal Aviation Agency at the Federal Aviation Agency at the FAA Flight Standards Inspector Approval or rejection) MPLETED ONLY BY FAA PER and for engineering comment	E (Check and complete with and is genee Man at Preparent Complete	lete appropriate intified in iter	contified under item 3 above ments of Part 18 of the University knowledge (Signature of authorized in the 18 of the University) Canadian Department Other (Specify) (Signature of authorized individual apparentment) Other (Specify) (Signature of authorized individual apparentment) Other (Specify) (Signature of authorized individual apparentment) Other (Specify) Other (Sp	manner pr	rescribe sport Ir	d by the Admi	

INSTRUCTIONS

This form must be completed in duplicate each time a major repair and/or alteration is made of an aircraft, airframe, power-plant, propeller or appliance. After the repair and/or alteration has been inspected and item 6 completed, the original copy of this form will be made available to the aircraft owner for retention as part of the aircraft records. The duplicate copy is retained by the FAA for administrative purposes.

See CAM 18 for detailed instructions concerning the information to be furnished with this form and instructions concerning the FAA for administrative purposes, its preparation.

B. DESCRIPTION OF WORK ACCOMPLISHED.

Major Overhanl: All steel parts were magnetically inspected; all aluminum and magnesium parts flourescent penetrant inspected. Installed new main and red bearings; new rings, exhaust valves, camshaft, gaskets and seals installed.

*If additional space is needed attach additional sheets bearing aircraft nationality and registration mark and date work completed. Check block if additional sheets are attached. $\hfill \square$

U.S. GOVERNMENT PRINTING OFFICE: 1960-O-538224

U. S. DEPARTMENT OF COMMERCE CIVIL AERONAUTICS ADMINISTRATION

Form Approved. Budget Bursan No. 41-R041.5.

INSTRUCTIONS

FOR AIRWORTHINESS CERTIFICATE

CIVIL AERONAUTICS ADMINISTRATION

Please print or type. Submit this form to the Civil Aeronautics Administration Aviation Safety

APPLICATION FOR AIR WOR	ION OF AN AIR	CRAFT	Field Re	presentative.	
1. TYPE OF APPLICATION (Check solice)					HEIONE OF CAR 8
■ S ORIGINAL ISSUANCE OF CERTIFICATE		d. RECE	TIFICATION	UNDER THE PROV	VISIONS OF CAR 8
b. ☐ ANNUAL INSPECTION FOR RENEWAL	OF CERTIFICATE	e. MULT	PLE CERTIFI	CATE UNDER THE	PROVISIONS OF CAR 8
e. AMENDMENT OR MODIFICATION OF C	URRENT CERTIFICATE	r. 🗆			
2. AIRWORTHINESS CLASSIFICATION It is requested that the Certificate of worthiness classification(s): . STANDARD (NORMAL, UTILITY, ACROST	Airworthiness be a	and to pos-	nit operatio	on of the aircra	aft in the following air-
L ☐ LIMITED (SEE CAR 9)					
a □ RESTRICTED (SEE CAR 8)					
(Check the restricted special purpose operati	ion(s) to be conducted)	_			- 1
☐ AGRICULTURAL AND PEST CON	TROL	☐ PATRO			ON
☐ AERIAL ADVERTISING				LIFE CONSERVATI	on
AERIAL SURVEYING			ER CONTRO	•	
GLIDER TOWING		OTHER			
d. EXPERIMENTAL (Check the type of experimental operation(*).	I to he conducted?				
RESEARCH AND DEVELOPMENT	, to de consumer,	☐ RACIN	G		
AMATEUR-BUILT		☐ EXHIB			
DEMONSTRATION		· OTHE			
			•		
3. AIRCRAFT IDENTIFICATION (Comp	ilete all items)				
a. AIRCRAFT MAKE	b. AIRCRAFT MODEL	100		c. AIRCRAFT S	
Cessna		182B			51676
d. ENGINE MAKE		e ENGINE N	ODEL		
Continental	*	0-470) - L		
4. AIRCRAFT REGISTRATION INFOR	MATION (Complete all	Rema)			
* REGISTERED OWNER'S FULL NAME		b. PERMANE	NT MAILING	ADDRESS	e. AIRCRAFT NATIONALITY AND REGISTRATION MARK
					AND REGISTRATION MARK
Cessna Aircraft Com	pany	Wich	ta, K	ansas	_{N-} 2376G
AIRCRAFT OWNER'S CERTIFICATI	ON (Check and complete	annes estate in			
registered* with the Civil Aeronautics or 502 and when operated displays the	gistered owner (or lands Administration as a following evidence	nis agent) of required by	the aircra	of tidentified in ations of the	n Item 3 above which is Administrator, Part 501
. CERTIFICATE OF REGISTRATION, FORM	ACA-500 (PART A). DAT	TE OF ISSUE _			
b. APPLICATION FOR REGISTRATION, FOR W-300 ON	M ACA-500 (PART B). FO	ORM ACA-500, F	ART A. FOR	WARDED TO CALL	
W-300 ON				INTED TO CAR	AIRCHAFT RECORDS BRANCH,
c. A DEALER'S REGISTRATION CERTIFICATE,	FORM ACA-1707, DATED	_7-28	-58		
*In order to be eligible for registration an airco owned by a citizen of the United States, as defined 1 (13) of the Civil Aeronauties Act of 1938, as ame					
TACHMENTS (Check which)			_		
ACA-319 WEIGHT AND BALANCE RE	DODT	10	013	2 1	\
ACA-337 DATA, DRAWINGS, ETC.	PORT	1	o Del	Roskam	200
ACA-317 UNAPPROVED DEVIATION		3 (SIGNAI	A REGIS	TERED OWNER OR A	UTHORIZED AGENT)
SIGHTROVED DEVIATION		2-19-	8		Owner's Agent

10 DEC 0 9 1966 Porm ACA-305 (11-51)

U. S. DEPARTMENT OF COMMERCE CIVIL AERONAUTICS ADMINISTRATION

AIRCRAFT INSPECTION REPORT

The aircraft described in Item 3 on the reverse of this for				
(Check and complete applicable stems)			/	
AIRCRAFT AND ENGINE CERTIFICATION BASIS AIRCRAFT SPECIFICATION NO. 3A13 THROU	UGH SHEET REVISION	1 NO		
b. AIRCRAFT LISTING PAGE NO.		7/		
	_ THROUGH CARD N	0		
4. CIVIL AIR REGULATION PART 8 (MODIFIED TYPE CERTIFICAL	(TE)			
2. AIRCRAFT AND ENGINE OPERATING RECORDS				
. IS AIRCRAFT NEW-NO PREVIOUS OPERATION OR MAINTENANCE	E HISTORY			
▶ ☐ COMPLIANCE WITH APPLICABLE AIRWORTHINESS DIRECTIVES	S RECORDED			
C. AIRCRAFT RECORDS INDICATE THE AIRFRAME HAS BEEN OP	ERATED A TOTAL OF		- HOURS	
d ENGINE RECORDS INDICATE THE FOLLOWING OPERATION:				
SERIAL NO TO'	TAL HOURS			
SERIAL NO TOT	TAL HOURS			
	TAL HOURS			
	TAL HOURS			
			- man	
PREVIOUS INSPECTION RECORD (INSPECTION REC	CORDED ON FO	RM ACA-31	9) .	
LAST AIRWORTHINESS INSPECTION CONDUCTED				
☐ BY AIRCRAFT MANUFACTURER	(DATE)			
BY APPROVED REPAIR STATION, CERTIFICATE NO				
BY MECHANIC, CERTIFICATE NO.				
PERIODIC AIRCRAFT INSPECTION REPORT, FORM ACA-319, WA				
E 1-10-10 Minister I mor zerion neroni, ronm aca-319, wa	IS KETUKNED TO OF	INER		
Ø OPERATION LIMITATIONS. F ORM ACA-389, WAS ISSUED (COPY) CURRENT OPERATION LIMITATIONS, FORM ACA-389, IS AVAILA CURRENT APPROVED AIRPLANE FLIGHT MANUAL IS AVAILABLE © CURRENT WEIGHT AND BALANCE INFORMATION IS AVAILABLE.	AFFAGHED CA	R 3.777	7 (b) di aircraft	splayed in
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Approved repair station certification (NAME OF ISSUING REPRESENTATIVE) (NAME OF ISSUING REPRESENTATIVE) Approved Repair Station Certification (NAME OF ISSUING REPRESENTATIVE) Approved Repair Station Certification aircraft described on the reverse has been inspect by CAR 52 and was found to be:	ARTAGHED CA ABLE IN AIRCRAFT E IN AIRCRAFT IN AIRCRAFT CORDS EXPIRE	ndefin: (DATE) (DATE)	ite	
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COPERATION LIMITATIONS. FORM ACA-359, WAS-16SUED (COPY- CURRENT OPERATION LIMITATIONS, FORM ACA-359, IS AVAILABLE CURRENT APPROVED AIRPLANE FLIGHT MANUAL IS AVAILABLE CURRENT WEIGHT AND BALANCE INFORMATION IS AVAILABLE THIS INSPECTION HAS BEEN RECORDED IN THE AIRCRAFT REC CERTIFICATE OF AIRWORTHINESS, FORM ACA-1362, ISSUED TO PREVIOUS FORM ACA-1362 WAS ISSUED TO EXPIRE BY (NAME OF ISSUING REPRESENTATIVE) APPROVED REPAIR STATION CERTIFICATION aircraft described on the reverse has been inspect by CAR 52 and was found to be: DMCR 3.1 (REPAIR S	ATTAGHED) CA ABLE IN AIRCRAFT E IN AIRCRAFT CORDS EXPIRE	ndefin: (DATE) (DATE) (DATE) (DATE) (DATE)	ite	
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APPROVED REPAIR STATION CERTIFICATION aircraft described on the reverse has been inspect by CAR 52 and was found to be: DMCR 3.1 REPRESENTATIVE CERTIFICATION THIS FEDERAL PROCESSION AIRCRAFT COMPANY DESIGN DISTRIBUTION DESIGN DESIG	MATTAGHED) CA MBLE IN AIRCRAFT E IN AIRCRAFT IN AIRCRAFT CORDS EXPIRE	INATION NO.) SIGNATURE) AIRWORTHY (Check app.	ite corded certification of the certification of t	icated repair statio

U. S. GOYERHHENT PRINTING OFFICE 16 GOR



Form ACA-305a (11-51)

UNITED STATES OF AMERICA DEPARTMENT OF TRANSPORTATION-FEDERAL AVIATION ADMINISTRATION STANDARD AIRWORTHINESS CERTIFICATE 4 CATEGORY 3 AIRCRAFT SERIAL NUMBER 2 MANUFACTURER AND MODEL 1 NATIONALITY AND REGISTRATION MARKS 51676 Normal N2376G Cessna 182B 5 AUTHORITY AND BASIS FOR ISSUANCE This airworthiness certificate is issued pursuant to 49 U.S.C. § 44704 and certifies that, as of the date of issuance, this aircraft has been inspected and found to conform to its type certificate and be in a condition for safe operation. This aircraft meets the requirements of the applicable airworthiness standards in Annex 8 to the Convention on International Civil Aviation, except as follows: None. 6 TERMS AND CONDITIONS Unless sooner surrendered, suspended, revoked, or a termination date is otherwise established by the FAA, this airworthiness certificate is effective as long as maintenance, preventative maintenance, and alterations are performed per the applicable Federal Aviation Regulations and the aircraft is registered in the United States. FAA REPRESENTATIVE BART HAUGER Digitally signed by BART HAUGER Date: 2019.04.22 16:11:00 -07:00' DESIGNATION NUMBER

Any alteration, misuse, or reproduction of this certificate for a fraudulent purpose may be punishable by certificate revocation, fine,

THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT PER THE APPLICABLE FEDERAL AVIATION REGULATIONS.

FAA Form 8100-2 (11-2016) Previous Edition Dated 04-11 May be Used Until Depleted

WP27FSDO

DATE OF ISSUANCE

R 07/28/1958

and / or imprisonment.



Cord Aero LLC

395 delmonte center box 261 Monterey, CA 93940 Cordaro Staub • cord@cordaero.com • 831-869-1605

Invoice WO_483 Date: February 22, 2022

Work completed: February 22, 2022 oil change and cyl # 2 compression

AIRCRAFT

N2376G

2-2

3. 3-1

3-2

Cessna 182B S/N: 51676 Tach: 4293.4

BILL TO

Chris Tinker 831-521-1270

1	Item	Discrepancy	Parts	Labor	Misc
	1.	Perform oil change	\$133.95	\$256.00	\$0.00
	1-1	Corrective Action			
		Drained used engine oil and removed oil filter PN AA4\$108-2 out open and inspected oil filter. No major decres at this time. Serviced engine with 10 qts Phillips 20WS0. No leaks noted.			
		- labor: 2.0 hrs @ \$128.00			
1	-2	Corrective Action			
		Drained used engine oil and removed oil filter PN A448108-2, cut open and inspected oil filter. No major debris at this time. Serviced engine with 10 qts Phillips 20W50. No leaks noted.			
		12 x Phillips 20W50 Oil @ \$8.46	\$101.52		
		1 x AA48108-2 Tempest oil filter @ \$32.43	\$32.43		
2	1	perform compression check on cylinder 2	\$0.00	200	
2-1		Corrective Action	\$0.00	\$42.24	\$0.00
	P 63	erformed compression check on cylinder #2. Cyl #2) 2/80.			
	-	labor: 0.33 hrs @ \$128.00			
2	Co	rrective Action			
	Per 62/	rformed compression check on cylinder #2. Cyl #2)			
	Cor	nply with AD 96-12-22 Oil filter adapter inspection			
	Con	rective Action	\$0.00	\$32.00	\$0.0
	Con	nplied with AD 96-12-22 by (a) 1-3 and not ence of oil leaking			
	la	bor: 0.25 hrs @ \$128.00			
		ective Action			
(Comp	olied with AD 96-12-22 by (a) 1-3 and not			

LABOR PARTS \$330.24 \$133.95

MISC

\$0.00

SALES TAX @ 9.000%

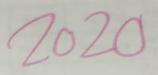
\$0.00

Payments are due upon completion of the services performed unless prior arrangements have been made.

REMAINING
FROM
LAST
ANNIAL

7 12.32

Radcliffe Enterprices 913 Camelia Drive Livermore California 94550 United States



Phone: 9252162514

Extended Compliance Report For Aircraft Registration

Company: Category:

Airframe

Manufacturer: Cessna Aircraft Company

Model: 182B

Position:

P/N:

S/N: 51676

Aircraft Registration No: N2376G

Current Hours:

Cycles:

As Of:

ATP Revision: 04/08/2020

Issue Number Effective Date	SB# Description	Complied	Complied Manual of Compliance	Recur	Next Due	Facility Cert No/Type Authorized By	
Amendment #	CAA#			Method of Company	Interval		Signed
2011-10-09 06/17/2011		Recurring] To prevent seat slippage or the seat roller housing from departing the seat rail, which may consequently cause, contd.	04/06/2020 Hrs: 4235.4 C:	Complied with by paragraphs (g) (1-3)	D: 365 Hrs: 100 C: -	04/06/2021 Hrs: 4335.4 C: – D: 363 Hrs: – C: –	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
2008-26-10 C 01/05/2009		To prevent erroneous indications from the altimeter, airspeed, and vertical speed, contd.	Hrs: C: -	Not applicable by part not installed	D: - Hrs: - C:	Hrs: - C: - D: - Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
08-10-02 /12/2008	1	To prevent erroneous indications from the altimeter, airspeed, and vertical speed, contd.	 Hrs: C:	Not applicable by part not installed	D: Hrs: C:	Hrs: - C: - D: - Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
4-19-01 C 01/2004	S	To prevent slippage of the pilot/co-pilot houlder harness, which could result in silure of the shoulder,contd.	 Hrs: C:	Not applicable by part number	D: - Hrs: - C: -	10	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe

Effective Date	SB# CAA#	Description	Complied	Method of Compliance	Recur	Next Due Remaining	Facility Cert No./Type Authorized By
98-16-04 09/21/1998		DURING FLIGHT CAUSED BY THE Hrs: ABSENCE OF AN ANGLE C:		Not applicable STC is not installed No		 Hrs: C:	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
		TIFFENER, WHICH COULD AUSE LOSS OF, CONTD.		1	D: Hrs: C:	D: — Hrs: — C: —	
97-01-13 02/03/1997	JH.	FO PREVENT FUEL, OIL, OR HYDRAULIC SYSTEMS FAILURE CAUSED BY A COLLAPSED HOSE	Hrs:	Not applicable by part number	No	 Hrs: C:	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
	CAUSED BY A COLLAPSED HOSE				D: Hrs: C:	D: - Hrs: - C: -	
96-12-22 07/31/1996	IEN OR	Recurring] TO PREVENT LOSS OF 04/06/2020 ENGINE OIL CAUSED BY LOOSE Hrs: 4235.4 OR SEPARATED OIL FILTER C: — Complied with by (a)(1-3) and no evidence of oil leaking.		Yes	Hrs: C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe	
	RE	APTERS, WHICH COULD SULT IN ENGINE, CONTD.			D: - Hrs: - C: -	D: - Hrs: - C: -	S&adrlote
87-20-03 R2 99/24/1990	Res		 Hrs: C:	[Recurring] Superseded by 2011-10-09		Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
					D: - Hrs: - C: -	D: Hrs: C:	
4-10-01 R1 /05/1988	[Recurring] TO PREVENT POWER LOSS OR ENGINE STOPPAGE DUE I TO WATER CONTAMINATION OF FUEL SYSTEM		04/06/2020 Hrs: 4235.4 C:	Complied with by AD instructions (1) -iii), (2)(i-ii), and (3)(1) by borescope inspection. Visually verified quick drain valve installed and operable.		Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
				drain valve installed and operable.	D: - Hrs: - C: -	D: - Hrs: - C: -	
-10-14 R1 0/1988	SOURCE IN CAS OBSTR	OVIDE AN ALTERNATE 04 CE OF FUEL TANK VENTINGHI SE OF FUEL TANK VENT C: RUCTION BY FOREIGN RIAL, CONTD.		Not applicable by seriel numbers liste in Cessna Service Letter SE77-6. Fue caps installed p/n C156003-0101	D: -		Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe

e Number	SB# CAA#	Description	Complied	Method of Compliance	Recur	Next Due Remaining	Facility Cert No./Type Authorized By Signed
Amendment # 86-26-04	CAA#	Superseded by 2004-19-01	Hrs:	Superseded by 2004-19-01	No	 Hrs:	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
01/06/1987		C:	C: -		D: Hrs: C:	C: - D: - Hrs: - C: -	Sieve Radellite
83-17-06 09/01/1983		DEGIROCIO LE LIMEDA	 Hrs: C:	Not applicable STC not installed	No	 Hrs: C:	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
		FLUTTER	C. –		D: - Hrs: - C: -	D: - Hrs: - C: -	D. 11105 Enterprises
83-13-01 08/01/1983			04/06/2020 Hrs: 4235.4 C: -	Complied with paragraph (a), (b)(1), (2) is not applicable by fuel cap part number, and (3) of this AD.	Yes D: 365 Hrs: - C: -	04/06/2021 Hrs: - C: - D: 363 Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
79-25-07 2/13/1979		TO PRECLUDE THE POSSIBILITY OF ELECTRICAL OR ELECTRONIC COMPONENT DAMAGE OR AN INFLIGHT FIRE, CONTD.	 CHrs: -C:	Not applicable by serial number	D: Hrs: C:		Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
78-01-14 01/23/1978		TO PRECLUDE THE POSSIBILITY OF AN IN-FLIGHT FIRE DUE TO A LOOSE ELT ANTENNA COAXIAL CABLE CONNECTOR, CONTD.	Hrs: C:	Not applicable by serial number	D: - Hrs: - C: -	Hrs: - C: - D: - Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
78-26-09 01/01/1978		Superseded by 79-10-14	 Hrs: C: -	Superseded by 79-10-14	D: Hrs: C:	Hrs: - C: - D: - Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe

					Recur	Next Due	Facility Cert No./Type	
Effective Date	SB# CAA#	Description	Complied	Method of Compliance	Interval	Remaining	Authorized By Signed Radcliffe Enterprices	
77-23-11 11/28/1977		LOOSE ELT ANTENNA COAXIAL	Hrs: -	Not applicable by aircraft serial number	No	Hrs: - C: -	2735886/A&P IA Steve Radcliffe	
	CABLE CONNECTOR, CONTD.	C: -	19-12 12 13-13-13	Aut. 1	D: - Hrs: - C: -			
77-04-05 *	*	INDUCTION AIR DOLL SELLE	 Hrs:		No	Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe	
		THE CARBURETOR		C:		D: - Hrs: - C: -	D: - Hrs: - C: -	Stalleti
FUEL EXHAUS		 Hrs:	Not applicable by model aircraft	No	Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe		
		INCORRECT FUEL PLACARDED CAPACITIES	C: -		D: - Hrs: - C: -	D: - Hrs: - C: -		
75-05-02 02/21/1975		[Recurring] TO PREVENT LOSS OF ENGINE OIL, IMPROPER ENGINE	Hrs: -	Not applicable, STC not installed	No	- Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe	
		LUBRICATION OR ENGINE OIL CONTAMINATION	C:		D; - Hrs: - C: -	D: - Hrs: - C: -		
73-17-01)8/16/1973		TO ADVISE THE PILOT CONCERNING PROPER FUEL	 Hrs:	Not applicable by part number	No	Hrs: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe	
	TRANSFER PUMP OPERATION	TRANSFER PUMP OPERATION	C: -		D: - Hrs: - C:	D: - Hrs: - C: -		
71-22-02 R(1) 1/09/1971	[Recurring] TO DECREASE THE POSSIBILITY OF FAILURE OF THE		E Hrs: C:	Not applicable, Nose Gear P/n 0543018-497 installed on 182 taken from a 206	No No	Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe	
	NOSE GEAR STRUCTURE	non a zoo		D: - Hrs: - C:	D: - Hrs: - C: -			

Engine

Manufacturer: Continental Motors

Model: O-470-L

Position: STC Convertions in 2006

P/N: O-470-50

S/N: 2440

Current Hours:

Cycles:

As Of:

ATP Revision: 04/08/2020

Issue Number Effective Date Amendment #	SB#	Description	Complied		Recur	Next Due	Facility Cert No./Type
	CAA#		Computed	Method of Compliance	Interval	Remaining	Authorized By Signed
2016-16-12 09/15/2016		To prevent failure of the cylinder assemblies, which could lead to failure of the engine, in-flight,contd.	 Hrs: C:	Not applicable by part number installed	No	 Hrs: C:	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
			D: – Hrs: – C: –	D: Hrs: C: -			
2010-11-04 06/16/2010		To prevent excessive hydraulic lifter wear, which can result in loss of engine power & loss of control of the airplane		Not applicable by overhaul date	No	 Hrs: C:	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
					D: Hrs: C:	D: - Hrs: - C: -	
2009-24-52 E 1/18/2009		Superseded by 2010-11-04	 Hrs: C:	Superseded by 2010-11-04	No	Hrs: C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
					D: - Hrs: - C:	D: - Hrs: - C: -	
009-24-51 E 1/16/2009		Superseded by 2009-24-52	 Hrs: C:	Superseded by 2009-24-52	No	 Hrs: C:	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
					D: - Hrs: - C: -	D: - Hrs: - C: -	
009-19-07 C2 0/07/2009		 Hrs: C:	Not applicable by part number installed	d No	Hrs: C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe	
				D: - Hrs: - C:	D: - Hrs: - C: -		

SB#	Description	Complied	Method of Compliance	Recur	Next Due Remaining	Facility Cert No./Type Authorized By Signed	
CAA#	Description		Lable by part number installed		-	Radcliffe Enterprices	
To prevent cylinder separation that car lead to engine failure, possible engine compartment fire, and,contd.	head to engine failure, possible engine	HIS:	Not applicable by part number installed	110	Hrs: - C: -	2735886/A&P IA Steve Radcliffe	
	C: -		D: - Hrs: - C:	D: - Hrs: - C: -			
			i If the businestallation of	No	-	Radcliffe Enterprices 2735886/A&P IA	
	Continental Motors reciprocating	Hrs:	Continental / P. Ponk engine in 2007		C: -	Steve Radcliffe	
	engines listed in AD 96-12-04	C: -		D: - Hrs: -	Hrs: -		
			ti 1 with the (a)(1 3) and no	No	-	Radcliffe Enterprices	
	Important for Cessna Oil Filter Adapter Assemblies listed in AD 96-12-22	HIS: 4233.4	evidence of oil leaking.		Hrs: - C: -	2735886/A&P IA Steve Radcliffe	
	C: -		D: - Hrs: -	D: - Hrs: -			
			No. 11 - bla by date of overhaul	1 2000	-	Radcliffe Enterprices	
	fournal fracture, which could result in	Hrs:	Not applicable by date of overland		Hrs: - C: -	2735886/A&P IA Steve Radcliffe	
	total engine power,contd.	C: -		D: - Hrs: -	D: - Hrs: -		
			Commented by 2000_23_21	No No	-	Radcliffe Enterprices	
	Superseded by 2000-23-21	Hrs:	Superseded by 2000-23-21		Hrs: C: -	2735886/A&P IA Steve Radcliffe	
		C. 2		Hrs: -	Hrs: -		
			Li 15 lie by date of averband		-	Radeliffe Enterprices	
	To prevent crankshaft failure due to crankshaft cheek cracks, which could		Not applicable by date of overnaut		Hrs: - C: -	2735886/A&P IA Steve Radcliffe	
	result in total engine power loss,contd.	C: -		Hrs: -	- Hrs: -		
	76000000	To prevent cylinder separation that can lead to engine failure, possible engine compartment fire, and, contd. Important notice for Teledyne Continental Motors reciprocating engines listed in AD 96-12-04 Important for Cessna Oil Filter Adapter Assemblies listed in AD 96-12-22 To prevent crankshaft connecting rod journal fracture, which could result in total engine power, contd. Superseded by 2000-23-21 To prevent crankshaft failure due to crankshaft cheek cracks, which could	To prevent cylinder separation that can lead to engine failure, possible engine compartment fire, and, contd. Important notice for Teledyne Continental Motors reciprocating engines listed in AD 96-12-04 Important for Cessna Oil Filter Adapter O4/06/2020 Hrs: 4235.4 C: — To prevent crankshaft connecting rod journal fracture, which could result in total engine power, contd. Superseded by 2000-23-21 Hrs: — C: — To prevent crankshaft failure due to —	To prevent cylinder separation that can lead to engine failure, possible engine compartment fire, and, contd. Important notice for Teledyne Continental Motors reciprocating engines listed in AD 96-12-04 Important for Cessna Oil Filter Adapter O4/06/2020 Important for Cessna Oil Filter Adapter O4/06/2020 Assemblies listed in AD 96-12-22 To prevent crankshaft connecting rod ournal fracture, which could result in total engine power, contd. Superseded by 2000-23-21 To prevent crankshaft failure due to crankshaft cheek cracks, which could trest our crankshaft cheek cracks.	To prevent crankshaft connecting rod fournal fracture, which could result in total engine power, contd. Complied Method of Compliance Interval	To prevent cylinder separation that can lead to engine failure, possible engine compartment fire, and, contd.	

amber	SB# CAA#	Description	Complied Method of C	Method of Compliance	Recur	Next Due Remaining	Facility Cert No./Type Authorized By
				Method of Compliance	Interval	Remaining	Signed
99-09-17 L 04/22/1999	Superseded by 99-19-01		Hrs: C:	No	 Hrs: C:	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe	
					D: - Hrs: - C:	D: - Hrs: - C: -	
98-17-11 C 0/19/1998		TO PREVENT CRANKSHAFT FAILURE DUE TO CRACKING, WHICH COULD RESULT IN AN	Hrs: C:	Not applicable by date of overhaul	No	Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
		INFLIGHT ENGINE FAILURE AND POSSIBLE, CONTD.			D: Hrs: C:	D: - Hrs: - C: -	
98-01-08 E 2/23/1997		TO PREVENT EXHAUST ROLLER ROCKER ARM FAILURE, WHICH CAN RESULT IN BENT PUSH	 Hrs: C:	Not applicable by part not installed	No	Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
		RODS, RUPTURED ROCKER ARM, CONTD.	C		D: - Hrs: - C: -	D: - Hrs: - C: -	
97-21-02 0/27/1997		TO PREVENT EXTREME SIDE LOADING OF THE PISTON, AND CONSEQUENT FAILURE OF THE PISTON AND ENGINE	Hrs: C: -	Not applicable by part not installed	No	 Hrs: C:	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
					D: - Hrs: - C: -	D: - Hrs: - C: -	
97-15-01 L 7/30/1997	Superseded by 97-21-02		 Hrs: C:	Superseded by 97-21-02	No	Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
		D: - Hrs: - C: -			D: - Hrs: - C: -		

amber	SB#		G		Recur	Next Due	Facility Cert No./Type Authorized By
Amendment #	CAA#	Description	Complied	Method of Compliance	Interval	Remaining	Signed
96-12-22 07/31/1996		Recurring] TO PREVENT LOSS OF ENGINE OIL CAUSED BY LOOSE OR SEPARATED OIL FILTER ADAPTERS, WHICH COULD RESULT IN ENGINE, CONTD.	04/06/2020 Hrs: 4235.4 C: —	Complied with by (a)(1-3) and no evidence of oil leaking.	Yes	Hrs: Due at every oil filter removal. C: —	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
					D: - Hrs: Due at every oil filter removal. C:	D: - Hrs: - C: -	400
95-21-15 11/28/1995		RESULT IN SEVERE ENGINE	 Hrs: C:	Not applicable by N number	No	 Hrs: C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
		DAMAGE AND SUBSEQUENT FAILURE			D: - Hrs: - C: -	D: - Hrs: - C: -	
94-14-12 L 6/23/1994		Superseded by 95-21-15	 Hrs: C:	Superseded by 95-21-15	No	Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
					D: - Hrs: - C: -	D: - Hrs: - C: -	
3-08-17 /23/1993		TO PREVENT AN ENGINE FAILURE	 Hrs: C:	Not applicable by serial number	No	Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
					D: - Hrs: - C:	D: - Hrs: - C: -	
-10-02 12/1993	- 1	TO PREVENT AN ENGINE FAILURE DUE TO A MISSING CYLINDER VALVE RETAINER KEY	 Hrs: C:	Not applicable by serial number	No	 Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
				D: - Hrs: - C: -	D: - Hrs: - C: -		

number	SB#	Description	Complied	Method of Compliance	Recur	Next Due Remaining	Facility Cert No./Type Authorized By Signed
Effective Date Amendment #	CAA#	Considerate Constant		Not applicable by date code	No	-	Radcliffe Enterprices 2735886/A&P IA
91-19-03 09/29/1991		TO PREVENT OPERATION WITH COLLAPSED OIL FILTER ELEMENTS, WHICH CAN RESULT IN LOSS OF OIL PRESSURE, CONTD.	Hrs: C: -	Not applicable by date so		Hrs: - C: -	Steve Radcliffe
					D: - Hrs: - C: -	D: - Hrs: - C: -	
88-03-06 04/15/1988		ENGINE FAILURE	 Hrs: C: –	Not applicable by partl number	No	 Hrs: C:	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
					D: Hrs: C:	D: - Hrs: - C: -	
				Not applicable by part numbers	No.	-	Radcliffe Enterprices 2735886/A&P IA
85-08-02 04/15/1985 79-05-09		TO PREVENT POSSIBLE VALVE STEM WEAR, OIL CONTAMINATION, VALVE SEIZURE, AND TOTAL LOSS OF ENGINE POWER	 Hrs: C:	Not approved by F	D: - Hrs: - C: -	Hrs: - C: - D: - Hrs: - C: -	Steve Radcliffe
		TO PREVENT THE POSSIBLE LOSS	S -	Not applicable by serial numbers	No	- Hrs: -	Radcliffe Enterprices 2735886/A&P IA
)3/12/1979		OF OIL PRESSURE INDICATION	Hrs: C:		D: - Hrs: - C: -	C: - D: - Hrs: - C: -	Steve Radcliffe
73-13-04 16/20/1973	CARBURETOR BOY	CARBURETOR BOWL DRAIN PLUG	DRAIN Hrs: C:	Not applicable by serial numbers	No	Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
					D: Hrs: C:	D: - Hrs: - C: -	
63-15-01 2/31/1969	EXHAUST VALVES	EXHAUST VALVES	 Hrs: C:	Not applicable by part numbers installed	No	 Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
					D: - Hrs: - C: -	223	

.umber					Recur	Next Due	Cert No./Type
	SB# CAA#	Description	Complied	Method of Compliance	Interval	Remaining	Authorized By Signed
Amendment # 61-03-02 01/01/1961		Superseded by 63-15-01	 Hrs:	Superseded by 63-15-01	No	 Hrs:	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
		The second	C: -	Managine	D: - D: - Hrs: - Hrs: - C: C: -		
56-06-01 TO PREVENT SERIOUS EN 7/01/1956 DAMAGE IN FLIGHT	TO PREVENT SERIOUS ENGINE DAMAGE IN FLIGHT		Not applicable by part number	No	Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe	
					D: Hrs: C:	D: Hrs: C:	

Propeller

... McCauley

Model: D3A34C401

Position:

S/N: 066578

P/N:

Current Hours:

Cycles:

As Of:

Issue Number en		124	Complied Method of Compliance		Recur	Next Due	Facility Cert No./Type Authorized By
Effective Date	SB# CAA#	Description			Interval	Remaining	Signed Radcliffe Enterprices
2005-14-11		To prevent blade failure that could	Hrs: -	Not applicable, no record of Southern		Hrs: - C: -	2735886/A&P IA Steve Radcliffe
08/17/2005		result in separation of a propeller blade H and loss of control of the airplane	C: – Inglewood, CA, also by model nu Model number D3A4CA-01-C	Inglewood, CA, also by model number Model number D3A4CA-01-C	D: - Hrs: - C: -	D: - Hrs: - C: -	
82-27-02 R1 02/22/1983	IDD/ODELLEV REALIE SDAIN	 Hrs: C: -	Not applicable by serial number installed	No D: - Hrs: -	Hrs: - C: - D: - Hrs: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe	
77-26-03		TO PREVENT POSSIBLE BLADE PITCH CONTROL FAILURES	 Hrs:	Not applicable by serial number installed	C: -	C: - - Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
06/07/1978		PITCH CONTROL PAILURES	C: -		D: - Hrs: - C: -	D: - Hrs: - C: -	

Magnetos

acturer: Bendix Corporation

Model: S-20 SERIES

Position: Left

P/N: 10-79020-10

S/N: 840157

Current Hours:

Cycles:

As Of:

Issue Number Effective Date	SB#	Description	Complied	Method of Compliance	Recur	Next Due Remaining	Facility Cert No./Type Authorized By Signed
Amendment #	CAA#		A	APPRIATE CONTRACTOR OF THE PROPERTY OF THE PRO	No		Radcliffe Enterprices 2735886/A&P IA
2005-12-06		[Recurring] To prevent failure of the magneto impulse coupling assembly		Not applicable b engine model		Hrs: - C: -	Steve Radcliffe
11/19/2003		and possible engine failure			D: - Hrs: -	D: - Hrs: - C: -	
96-12-07		[Recurring] Superseded by 2005-12-06	 Hrs:	[Recurring] Superseded by 2005-12-06	No	Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
07/18/1996			C: -		D: - Hrs: - C: -	D: - Hrs: - C: -	
94-01-03 R2 06/28/1995		TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE	Hrs: C:	Not applicable by serial number	No D: - Hrs: -	Hrs: - C: - D: - Hrs: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
94-06-09 05/20/1994		TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON-GROUNDED MAGNETO	Hrs: C: -	Not applicable by date code	D: - Hrs: -	C: - Hrs: - C: - D: - Hrs: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
82-20-01 06/14/1983		TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD.	- Hrs: C:	Not applicable by serial number	D: - Hrs: - C:	C: - Hrs: - C: - D: - Hrs: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe

aper T	SB#		Complied	Method of Compliance	Recur	Next Due	Facility Cert No./Type
Amendment #	CAA#	Description			Interval	Remaining	Authorized By Signed
78-09-07 R3 01/17/1983		[Recurring] Superseded by 96-12-07	- Hrs:	[Recurring] Superseded by 96-12-07	No	 Hrs: C:	Radeliffe Enterprices 2735886/A&P IA Steve Radeliffe
		C:	- promittasphose	D: - Hrs: C:	D: Hrs: C:		
Carlo		S-20,-200,-1200 SERIES MAGNETOS	Hrs: C: -	Not applicable by serial number	No	 Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
		C: -		D: - Hrs: - C: -	D: - Hrs: - C: -		
73-07-04 0/11/1973		Superseded by 94-01-03	Hrs: C:	Superseded by 94-01-03	No	 Hrs: C:	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
					D: Hrs: C:	D: - Hrs: - C: -	

..... Air Precision

Model: ELT91A

Position:

P/N: E-01

S/N: 006653

Current Hours:

Cycles:

As Of:

	SB#	Description		Method of Compliance	Recur	Next Due	Facility Cert No./Type
	CAA#		Complied		Interval	Remaining	Authorized By Signed
-			 Hrs: C:		-	 Hrs: C:	
					D: - Hrs: C:	D: - Hrs: C:	

Position: P/N: A152A

Cycles:

Current Hours:

ATP Revision: 04/08/2020

As Of:

Any Manufacturer

Model: Any Model

S/N: 2116

Facility Next Due Recur Cert No./Type Issue Number Authorized By SB# Effective Date Description Complied Method of Compliance Remaining CAA# Interval Signed Amendment # Radcliffe Enterprices Not applicable by manufacturer No [Recurring] TO DETECT 2735886/A&P IA 76-02-07 Hrs: -Steve Radcliffe 02/02/1976 DEFECTIVE ALTERNATOR SLIP RING END BEARINGS AND MINIMIZE THE PROBABILITY OF Hrs: -Hrs: -IN-SERVICE FAILURES Radeliffe Enterprices Not applicable by manufacturer No 2735886/A&P IA [Recurring] TO PRECLUDE IN-72-15-02 Hrs: -Hrs: -Steve Radcliffe SERVICE FAILURES OF 09/01/1972 ALTERNATOR COOLING FANS C: -Hrs: -Hrs: -Radcliffe Enterprices No Superseded by 72-15-02 2735886/A&P IA Superseded by 72-15-02 72-01-05 Hrs: -Hrs: --Steve Radcliffe 01/01/1972 Hrs: -Hrs:

Magnetos

anulacturer: Bendix Corporation

Model: S-20 SERIES

Position: Right

P/N: 10-79020-10

S/N: 504803

Current Hours:

Cycles:

As Of:

Issue Number	SB#				Recur	Next Due	Facility Cert No./Type
Effective Date Amendment #	CAA#	Description	Complied	Method of Compliance	Interval	Remaining	Authorized By Signed
2005-12-06 07/19/2005		[Recurring] To prevent failure of the magneto impulse coupling assembly and possible engine failure	- Hrs:	Not applicable by engine model	No	Hrs: C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
	and possion engine failure	C: -		D: Hrs: C:	D: - Hrs: - C: -		
96-12-07 07/18/1996	[Recurring] Superseded by 2005-12-06	6 Hrs: C:	[Recurring] Superseded by 2005-12-06	No	 Hrs: C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe	
				D: - Hrs: - C:	D: - Hrs: - C: -		
94-01-03 R2 06/28/1995	TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE	 Hrs: C:	Not applicable by serial number	No	- Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe	
				D: - Hrs: - C: -	D: - Hrs: - C: -		
94-06-09 05/20/1994		TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON-GROUNDED MAGNETO) Hrs: C:	Not applicable by date code	No	 Hrs: C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
					D: Hrs: C:	D: - Hrs: - C: -	
82-20-01 06/14/1983	HWIF OLDE COOL PILLOR	 Hrs: C:	Not applicable by serial number	No	Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe	
				D: Hrs: C:	D: Hrs: C:		

mber pate Amendment # 78-09-07 R3	SB# CAA#	Description		Method of Compliance	Recur	Next Due	Facility Cert No./Type Authorized By Signed
	COCKER		Complied		Interval	Remaining	
01/17/1983	[Recurring] Superseded by 96-12-07	 Hrs: C:	[Recurring] Superseded by 96-12-07	No	 Hrs: C:	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe	
					D: Hrs: C:	D: Hrs: C:	
74-26-09 12/24/1974	S-20,-200,-1200 SERIES MAGNETOS	 Hrs: C:	Not applicable by serial number	No	 Hrs: C:	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe	
					D: - Hrs: - C: -	D: - Hrs: - C: -	
73-07-04 0/11/1973		Superseded by 94-01-03	Hrs: C:	Superseded by 94-01-03	No	Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
					D: – Hrs: – C: –	D: - Hrs: - C: -	

Macturer: Kelly Aerospace

Model: Any Model

Position:

Current Hours:

P/N: ES-646238-2

Cycles:

S/N: G031909

As Of:

Issue Number Effective Date Amendment #	SB# CAA#	Description	6	Method of Compliance	Recur	Next Due	Facility Cert No./Type
			Complied		Interval	Remaining	Authorized By Signed
			 Hrs:		-	 Hrs:	
			C:		D:	C:	
					Hrs: C:	Hrs: C:	

Vacuum System

Airborne Manufacturing

Model: 216CW

Position:

P/N: 216CW

S/N: 51680

Current Hours:

Cycles:

As Of:

Issue Number Effective Date Amendment #	SB#	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No./Type
	CAA#				Interval	Remaining	Authorized By Signed
98-23-01 11/20/1998		TO PREVENT FAILURE OF THE PRIMARY DRY AIR PUMP CAUSED BY DEFECTIVE FLEXIBLE COUPLING, WHICH COULD, CONTD.	 Hrs: C:	Not applicable by serial number installed	D: Hrs: C:	Hrs: - C: - D: - Hrs: - C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe

reledyne Battery

Position: P/N:

S/N: 51680

Current Hours:

Cycles:

As Of:

SB# CAA#	Description	Complied		Recur Next Due		Facility Cert No./Type
-			Method of Compliance	Interval	Remaining	Authorized By Signed
		Hrs: C:			-	
				100000	C	
			CAA# Description Complied	CAA# Description Complied Method of Compliance Hrs:	CAA# Description Complied Method of Compliance Interval Hrs: C: D:	CAA# Description Complied Method of Compliance Recur Next Due Interval Remaining

Senttion Systems

Teledyne Continental

set: Any Model

Position:

P/N:

S/N:

Current Hours:

Cycles:

As Of:

Issue Number Effective Date Amendment #	SB# CAA#	Description Complied		A	ATP Revision: 04/08/2020 Recur Next Due Facility		The second secon
76-07-12 R1 08/30/1977		[Recurring] IGNITION SWITCHES		Method of Compliance	Interval	Remaining	Cert No./Type Authorized By Signed
			04/06/2020 Hrs: 4235.4 C: —	Complied with by paragraph (1)(a-c) Ruplaced switch	Yes	Hrs: 4335.4 C: -	Radcliffe Enterprices 2735886/A&P IA Steve Radcliffe
					D: Hrs: 100 C:	D: - Hrs: - C:	S Radulfli

WD: WO 483



Airframe - page 1 of 1

Cord Aero LLC · 395 delmonte center box 261 · Monterey, CA 93940 · 831-869-1605

N2376G

Cessna, 182B, S/N: 51676

February 22, 2022

Tach: 4293.4

- Drained used engine oil and removed oil filter PN AA48108-2, cut open and inspected oil filter. No major debris at this time. Serviced engine with 10 qts Phillips 20W50. No leaks noted.
- 2. Performed compression check on cylinder #2. Cyl #2) 62/80.
- 3. Complied with AD 96-12-22 by (a) 1-3 and not evidence of oil leaking.

With respect to the work performed, this airframe is approved for return to service. Details of work performed are on file with Cord Aero LLC under work order: WO_483.

Cordaro Staub A&P 3612277IA

